

# Improving Adhesive Bonding of Composites Through Surface Characterization

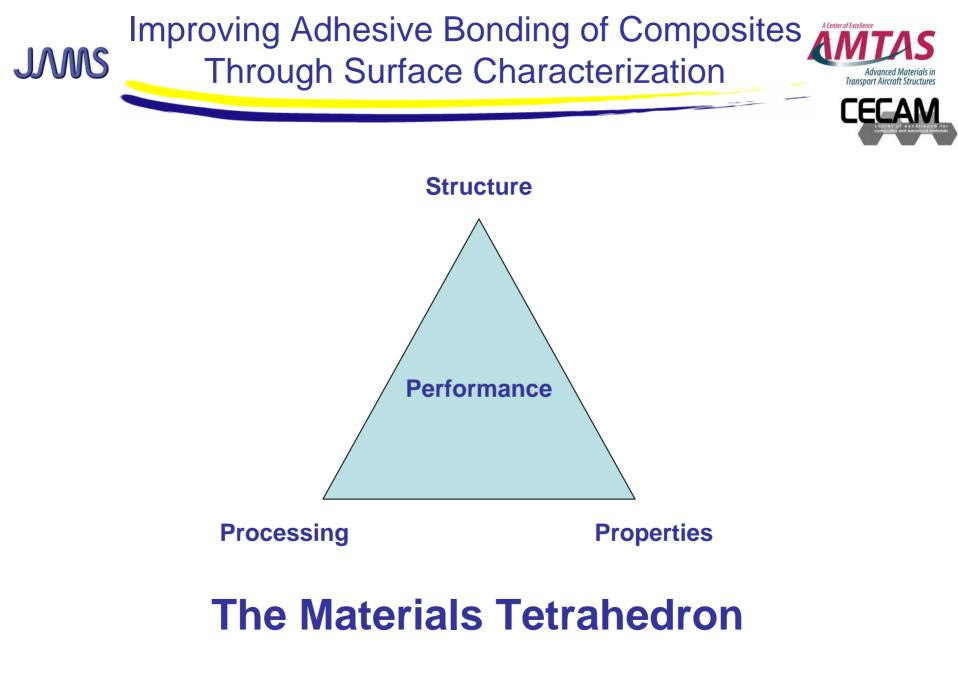
(of Peel Ply Prepared Surfaces)

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- Motivation and Key Issues
  - Peel ply surface preparation is being used for bonding primary structure in commercial transport aircraft
  - Good bonds are produced but questions remain:
    - What are appropriate techniques to inspect surfaces?
    - What are key factors for making a good/poor bond?
    - How to predict material and surface preparation compatibility?
- Objective
  - Further understand the requirements for peel ply surface preparation to produce strong primary structural composite bonds with different substrates and adhesives

# Improving Adhesive Bonding of Composites Through Surface Characterization

### • Approach

- Investigate the effect of various peel-ply and prepreg material systems on the adherend surface chemistry/structure and subsequent bond performance
- Prepreg Materials:
  - Glass Fiber Epoxy 127° C (260 ° F)
  - Carbon Fiber Epoxy 176° C (350 ° F)
  - Peel Plies: Dry and Preimpregnated Nylon and Polyester
  - Adhesives 127° C (260° F) and 176° C (350° F)
- Characterization
  - Surface chemistry, SEM, mechanical testing and fractography

# FAA Sponsored Project Information

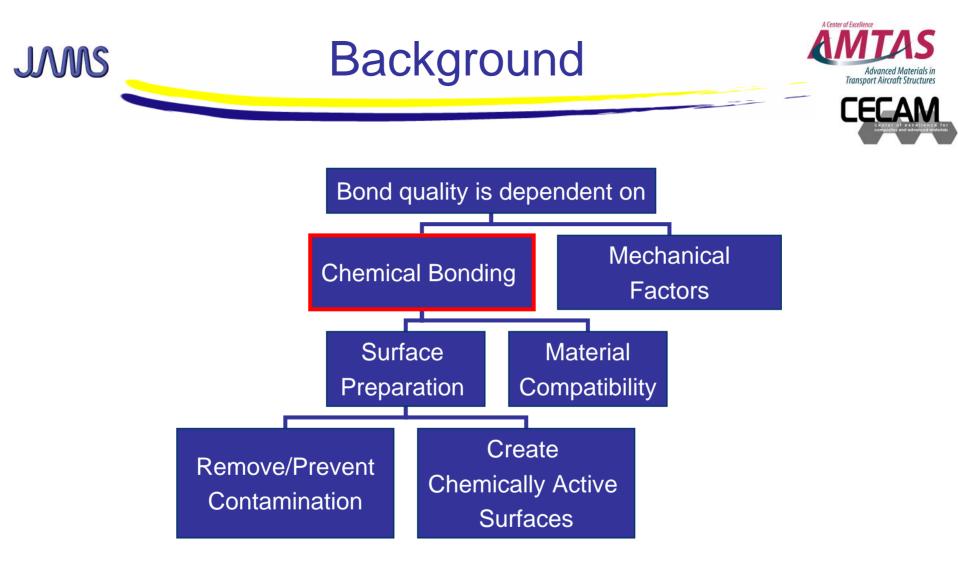




- Principal Investigators & Researchers
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JMS

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- Brian Clark (Masters student, U of Wa.)
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- FAA Technical Monitor
  - Curt Davies (Peter Shyprykevich, Retired)
- Other FAA Personnel Involved
  - Larry Ilcewicz
- Industry Participation
  - Boeing: Peter Van Voast, William Grace, Paul Shelly
  - Precision Fabrics Group, Cytec, Toray, 3M, Henkel, Yokahama
- JAMS Participation
  - Mark Tuttle: Technical Discussions, Wettability Envelopes
  - Lloyd Smith (WaSU): Parallel study on durability



Surface preparation is a key ingredient to most successful adhesive bonding applications



- John Hart-Smith- Curse of the Nylon Peel Ply
- Bardis and Kedward showed peel ply was not an effective method for some resin systems-adhesion failure, low fracture energy.
- Previous research on carbon fiber reinforced epoxy prepreg, BMS8-276 (177° C; 350° F) cure showed
  - Polyester peel-ply prepared surfaces produced good bonds
  - Nylon peel-ply prepared surfaces did not bond well
  - Remnants of nylon peel-ply found on surface (SEM, XPS)
- This research:
  - Glass fiber epoxy prepregs: BMS8-79 (127° C; 260° F) cure
  - Carbon Fiber epoxy prepregs: BMS8-256 and Toray 3631 (177° C)
  - Nylon and polyester peel plies (dry and preimpregnated)
  - Various film adhesives

# JMS Peel Ply Surface Preparation



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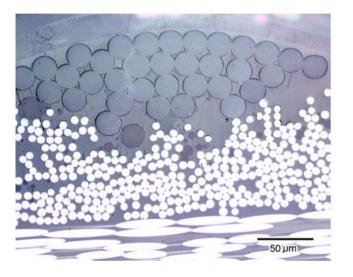
Peel ply

Composite

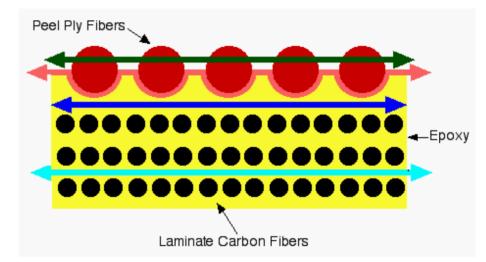


- Peel Ply-Woven fabric
  - Typically thermoplastic polymer
  - Placed on surface during layup
- Cured with the part matrix resin infiltrates peel ply weave
- Removed just before bonding
- Ideally Leaves rough, clean, chemically active surface
- Benefits:
  - straightforward
  - consistent
- If only they always worked!

# **Peel Ply Surface Preparation** JMS



### Fracture Possibilities Upon Peel Ply Removal





Fracture of the epoxy between peel ply and carbon fibers

• Fresh, chemically active, epoxy surface is created



Interfacial fracture between the peel ply fabric fibers and the epoxy matrix



Peel ply fiber fracture Interlaminar failure



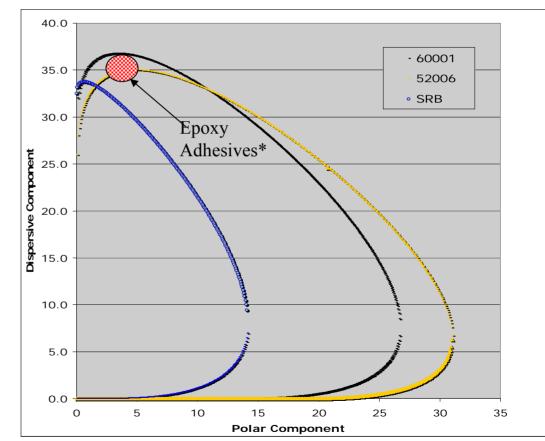
# Wettability envelopes showed the difference in the prepared surfaces.



 Fluids inside the envelope will wet spontaneously

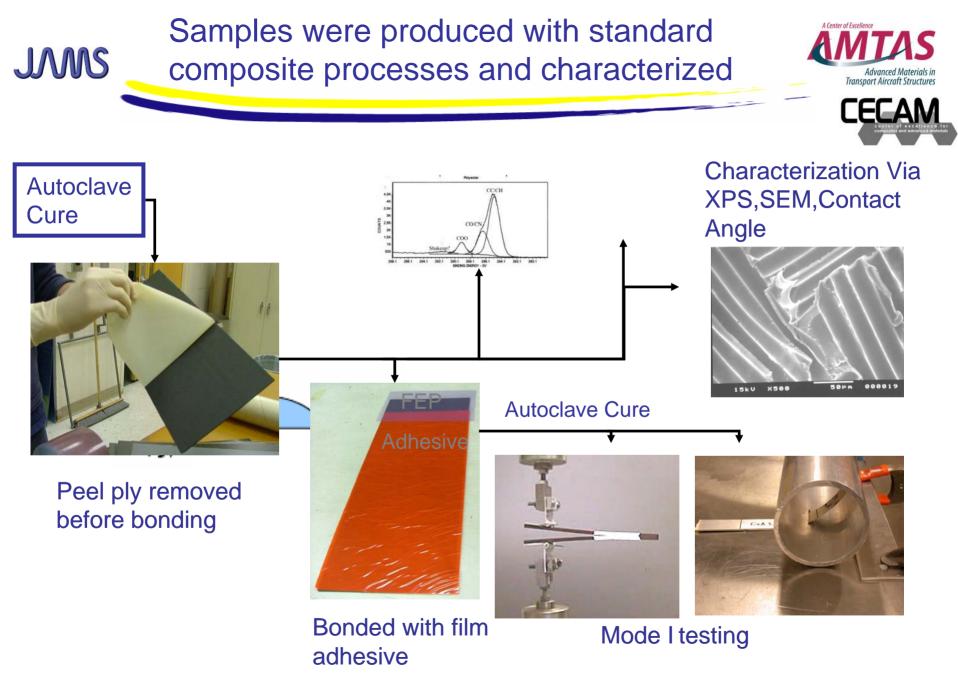
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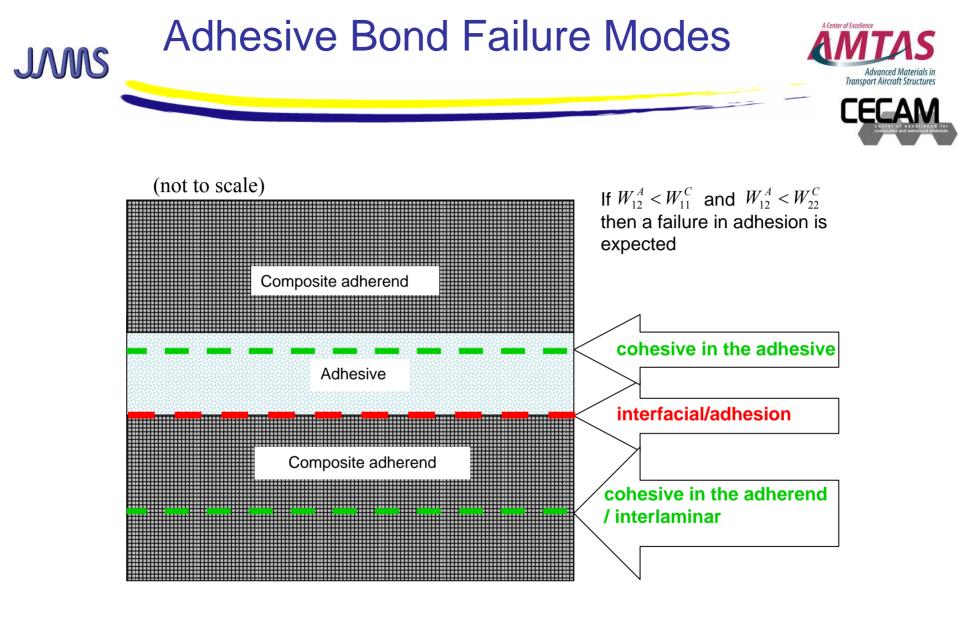
- Critical condition for bonding?
- Wettability envelopes a potential method to determine suitability of a surface for bonding
- Epoxy adhesives\* on boundary for nylon prepared surfaces



\* Literature values for aerospace epoxies

- Curves generated using WET program (M. Tuttle)





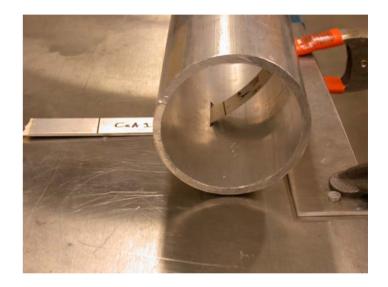
Failure modes for adhesive bonds: cohesive/interlaminar or adhesion

### The Rapid Adhesion Test (RAT) Method JMS

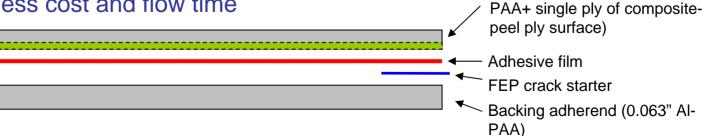




- A quick, low cost test which assesses the adhesion between metal-composite bonds.
- Failure typically in composite-adhesive not metal-adhesive, therefore evaluates composite bond quality
- A modification of metal-to-metal peel test developed by Boeing.
- The backing adherend clamped to while the peeling adherend is removed
- Failure mode representative of bond
  - Adhesion Failure-Poor Bond
  - Cohesive Failure-Strong Bond
- Failure modes correlate with DCB test with
  - ~90% less cost and flow time



Peeling adherend (0.020" Al





- RAT was created at Boeing as an easy, fast qualitative measure of bonding
- Mode I test
- Intended for screening out poor adherendadhesive-surface prep combinations
- Found to have a qualitative agreement with DCB testing in terms of mode of failure
  - Cohesive / interlaminar failure: acceptable
  - Adhesion failure (failure at the adherend-adhesive interface): BAD!
- A tenth of the cost & time for DCB testing

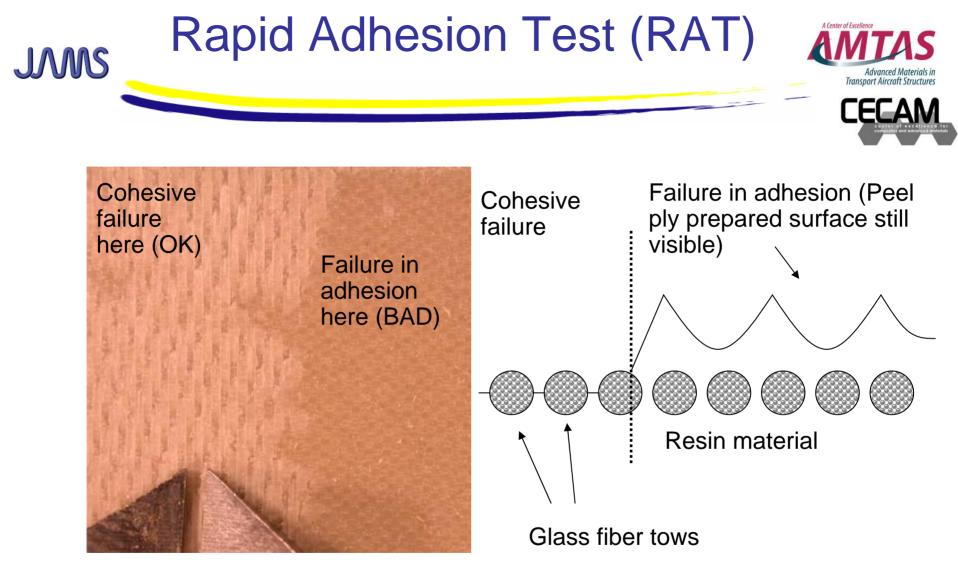


### Cohesive failure (left) vs. Adhesion failure (right)

**FEP starter crack** 

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**FEP starter crack** 



Mixed failure mode of RAT specimen (looking at substrate surface after peeling off adhesive); Cytec Cycom MXB 7701/7781 – P 60001 – Cytec FMx209



### Glass Fiber-Epoxy 126° C (260° F)

- The purpose of this task was to evaluate the effectiveness of peel ply surface preparation for adhesive bonding of GFRP
- Compatibility of different commercial prepregs, peel plies and adhesives
- Further fundamental understanding of peel ply surface preparation for adhesive bonding



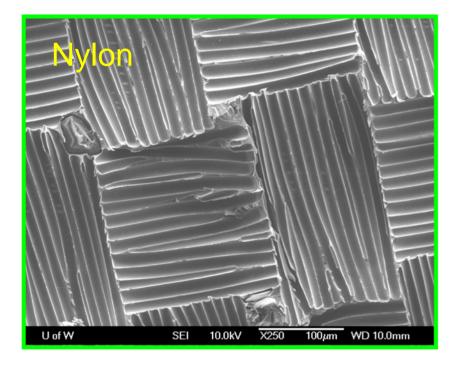
- 2 Peel Plies: Polyester 60001 and Nylon 52006
- 3 prepregs-260° F cure
  - HexPly® F155
  - Yokohama G7781
  - Cytec MXB7701
- 6 adhesives-260° F cure
  - 3M AF500; 3M AF163-2;
  - Henkel EA 9696; Henkel EA 9628
  - Cytec FM94; Cytec FMx 209
- Bond quality assessed by failure mode
  - Adhesion (poor) vs. Cohesive (good)

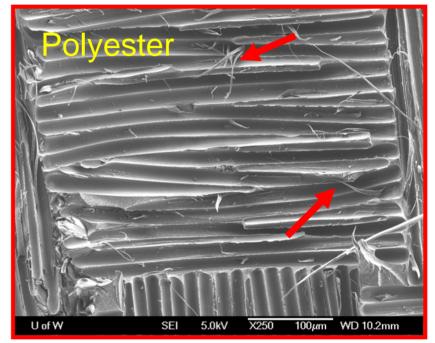
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# JMS Peel Ply Surface Prep. - SEM Results

### Advanced Materials in Transport Aircraft Structures

### Composite surfaces after removal of peel ply:





**Clean surface** 

### Remnants of polyester peel ply fibers left on surface

# JMS Surface Energy Measurement Avanced Materials in

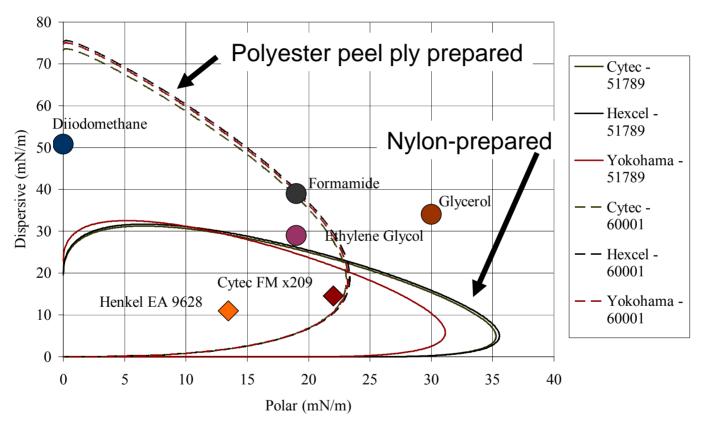


### Table of surface energies from contact angle data

Substrate with Peel Ply	Surface Energy (mN/m)			
	polar	dispersive	total surface	
	component, $\gamma^d$	component, $\gamma^p$	energy, γ	
Hexcel 1581-F155 with Nylon 51789	26.7	20.7	47.4	
Hexcel 1581-F155 with Polyester 60001	0.7	66.4	67.0	
Yokohama F6986 with Nylon 51789	20.2	23.9	44.1	
Yokohama F6986 with Polyester 60001	0.5	66.8	67.4	
Cytec Cycom MXB 7701/7781 with Nylon 51789	25.1	21.6	46.7	
Cytec Cycom MXB 7701/7781 with Polyester 60001	1.3	60.3	61.6	
Epoxy (ave. literature value)	34.1	2.6	36.7	
Polyester (PET)-(ave. literature value)	4.5	37.9	42.4	
Nylon-6,6-(ave. literature value)	33.6	7.8	41.4	

### Peel ply type influenced surface energy of composite

# JMS Wettability envelopes for peel-ply Marines prepared surfaces



- Fluids inside the envelope will wet spontaneously
  - Critical condition for bonding?
- Epoxy adhesives\* on boundary for polyester prepared surfaces





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#### Nylon peel ply (Precision code 51789-52006)

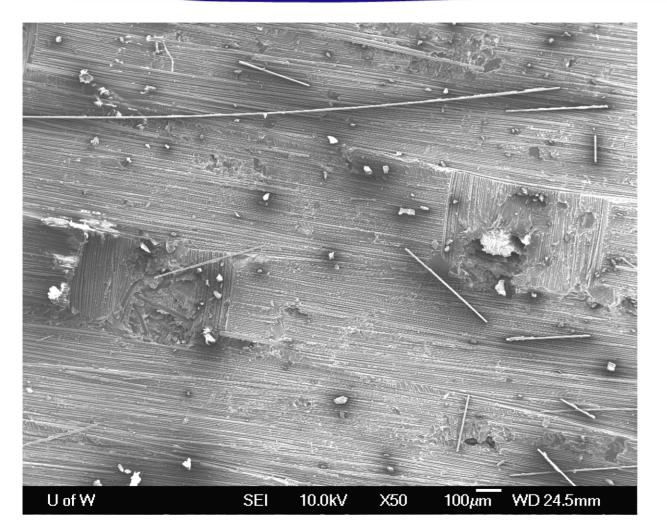
Adhesive	Hexcel 1581-F155	Yokohama F6986	Cytec Cycom MXB 7701/7781
3M AF500	COHESIVE	COHESIVE	COHESIVE
3M AF 163-2M	COHESIVE	COHESIVE	COHESIVE
Cytec FM 94	COHESIVE	COHESIVE	COHESIVE
Henkel Hysol EA 9696	COHESIVE	COHESIVE	COHESIVE
Cytec FM x209	COHESIVE	COHESIVE	COHESIVE
Henkel Hysol EA 9628	COHESIVE	COHESIVE	COHESIVE

#### Polyester peel ply (Precision 60001)

3M AF500	ADHESION	ADHESION	ADHESION
3M AF 163-2M	ADHESION	ADHESION	ADHESION
Cytec FM 94	ADHESION	ADHESION	ADHESION
Henkel Hysol EA 9696	ADHESION	ADHESION	ADHESION
Cytec FM x209	MIXED	MIXED	MIXED
Henkel Hysol EA 9628	ADHESION	ADHESION	ADHESION

# SEM – RAT bond fracture





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Nylon-prepared surface *after* performing RAT (Cytec 7701/7781, Henkel EA 9628)

=> Good bond

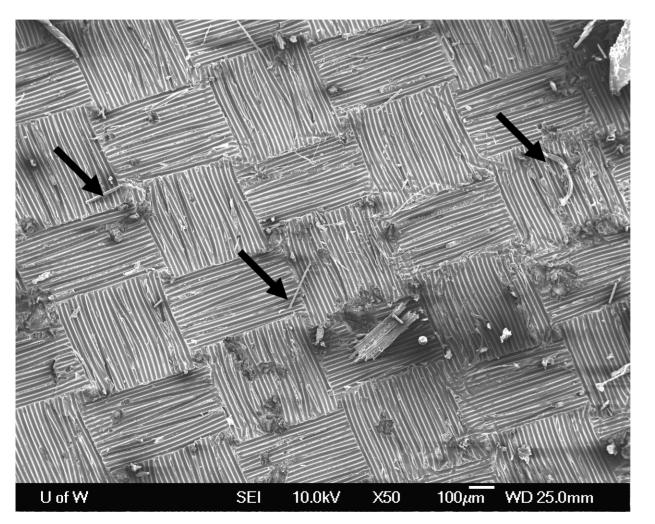
Glass fibers are clearly visible, post-fracture

### **Cohesive Fracture in GFRP substrate**

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# SEM – RAT bond fracture





Polyesterprepared surface *after* performing RAT (Cytec 7701/7781, Henkel EA 9628)

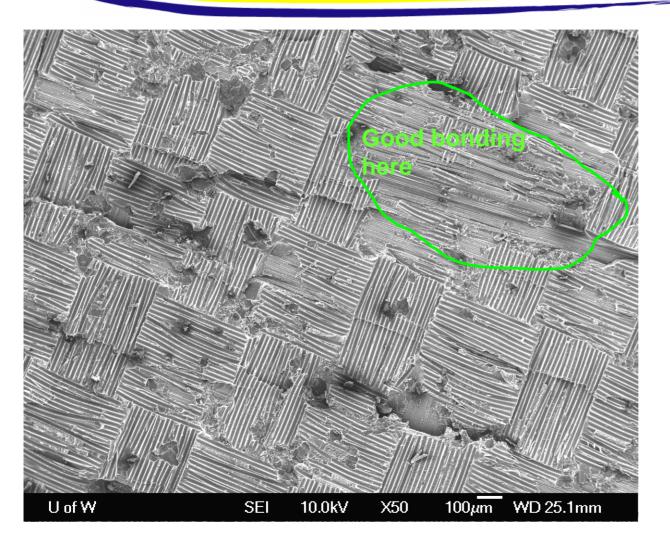
=> BAD BOND

Tendrils of peel ply originally left behind seem to be still present

### Adhesion Failure at Bond Line

## SEM – RAT bond fracture





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Polyesterprepared surface *after* performing RAT (Cytec 7701/7781, Cytec FMx209)

=> Mixed result

### Mixed Adhesion(~80%) and Cohesive (~20%)



Bond Quality Depends on:

### Peel Ply Material and Adhesive

- Nylon : high toughness bonds, cohesive failure all adhesives
- Polyester peel ply: low toughness, adhesion failure
- One adhesive bonded to all surfaces
- Opposite Trend than BMS8-276 (350 F) system
   Nylon bad, Polyester good
- The wetting envelopes generated for the various prepared surfaces gave no real insight into why polyester was inadequate.
  - Surface energy of polyester surfaces>nylon surfaces
- The SEM surface examination revealed a potential cause of the problem the polyester peel ply is interacting with the matrix to leave tendrils of material, indicates contamination



- Investigate fiberglass prepregs with higher temperature cure cycles with nylon and polyester peel plies
- Conduct similar tests using different weaves of peel plies of the same materials
- Determine role (if any) of curing and toughening agents in epoxies
- Examine other surface characterization techniques that may predict poor bonding behavior of the polyester peel ply surfaces



- Expand study to 2 additional 176° C carbon fiber prepreg systems
- Expand to other peel-plies, including epoxy preimpregnated polyester and nylon ("wet")
- Further understand the effect of peel ply surface preparation on the durability of primary structural composite bonds through surface analysis coupled with mechanical testing and fractography



### Aerospace carbon fiber-epoxy prepregs

- UD Toray 3631 toughened hot melt epoxy with T-800 fiber
- Cytec-Cycom 970 toughened epoxy and plain weave 3K-70

### Aerospace grade film adhesives

- Cytec Metal Bond 1515-3
- 3M AF 555

### **Peel plies**

- Dry polyester Precision Fabrics 60001 Nat
- Dry nylon Precision Fabrics 52006/51789 Nat
- Epoxy-preimpregnated polyester Henkel EA-9895
- Epoxy-preimpregnated nylon Cytec MXM 7934/52006

# Contact angle results

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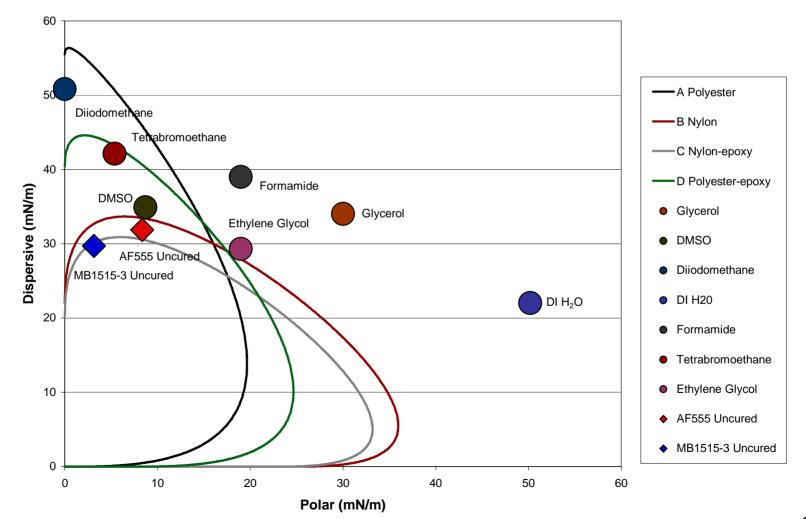


- Using the contact angles for four fluids—ethylene glycol, deionized water, glycerol and formamide—Kaelble plots were generated and the polar and dispersive surface energies evaluated for each prepreg-peel ply combination
- Notably, nylon-prepared surfaces exhibit a greater polar surface energy component and polyester-prepared surfaces exhibit a greater dispersive surface energy component

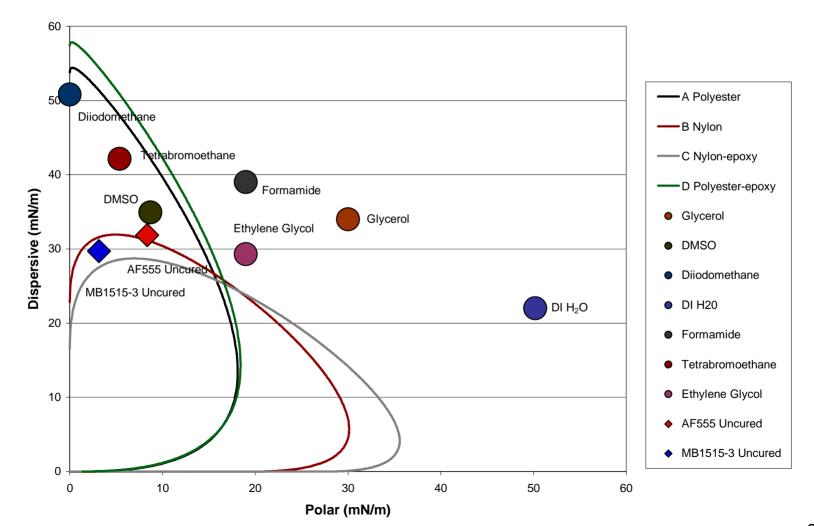
Substrate - Peel ply *	$\gamma_{d}$	γ <sub>p</sub>	$\gamma_{tot}$
Cytec970 – 60001 polyester	55.5	1.7	57.2
Cytec970 – 51789 nylon	22.0	25.8	47.8
Cytec970 – EA9895 polyester/epoxy	40.4	8.6	49.0
Cytec970 – nylon/epoxy	20.1	23.9	44.0
Toray 3631 – 60001 polyester	53.8	1.2	55.0
Toray 3631 – 51789 nylon	22.8	19.8	42.6
Toray 3631 – EA9895 polyester/epoxy	57.4	0.9	58.3
Toray 3631 – nylon/epoxy	16.5	27.8	44.3
(Adhesive) 3M AF555 uncured	31.6	8.9	40.5
(Adhesive) Cytec MB1515-3 uncured	29.7	3.1	32.8

\* Units in mN/m









# JMS X-Ray Photospectroscopy results



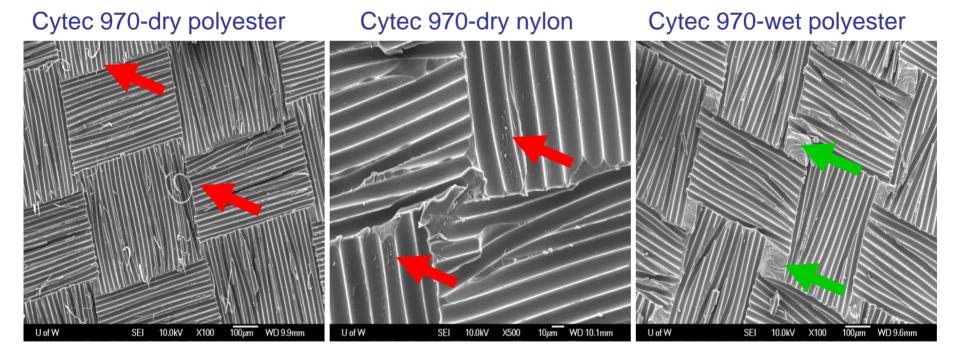


- XPS was carried out on all of the peel ply-prepared surfaces to determine composition
- The polyester-prepared surfaces demonstrated high oxygen content due to the C=O bonds within polyester fiber; the nylon-prepared surfaces demonstrated high nitrogen content due to the presence of amide C=N bonds in the nylon

Substrate - Peel ply	C (At.%)	O (At.%)	N (At.%)	Si (At.%)	Br (At.%)	S (At.%)
Cytec 970 - PF60001	73.8	25.2	1.0	**	**	**
Cytec 970 - PF51789	76.1	12.4	11.5	**	**	**
Cytec 970 - Epoxy/nylon	77.5	12.9	9.6	**	**	**
Cytec 970 - EA9895	76.8	19.6	3.1	**	0.5	**
Toray 3631 - PF60001	70.5	25.9	1.6	1.3	**	0.6
Toray 3631 - PF51789	77.1	13.3	9.0	**	**	0.7
Toray 3631 - Epoxy/nylon	76.2	12.1	10.7	**	**	1.0
Toray 3631 - EA9895	79.0	18.3	1.2	**	1.5	**

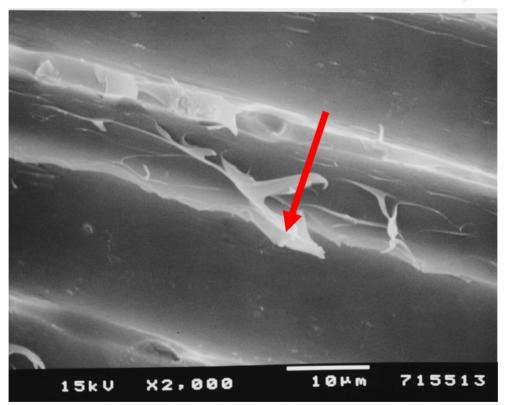
# SEM Results: peel ply removed

- SEM imaging was carried out on the prepreg surfaces both immediately subsequent to peel ply removal as well as after Mode I fracture by the rapid adhesion test
- The images below are those taken after peel ply removal; only the Cytec 970 resin system is imaged because the peel ply texture is the same for both the Cytec 970 and Toray 3631





### Laminate surface after removal of nylon peel ply



#### Nylon from peel ply on surface before bonding?



- Samples which exhibited cohesive failure (interlaminar in the composite, or cohesively within the epoxy/adhesive zone) were classified as good bonds; those which exhibited adhesion failure (along the adhesive-matrix bondline) were classified as poor bonds
- Bonds with a significant fraction of both are labeled mixed

	Peel ply				
Substrate - Adhesive	PF60001 (dry polyester)	PF51789 (dry nylon)	EA9895 (polyester- epoxy)	Nylon- Epoxy	
Cytec 970 -	Mixed	Adhesion	Cohesive	Adhesion	
MB1515-3 Cytec 970 - AF555	Mixed	Mixed	Cohesive	Cohesive	
Toray 3631 -	Adhesion	Adhesion	Cohesive	Adhesion	
MB1515-3					
Toray 3631 - AF555	Adhesion	Adhesion	Cohesive	Adhesion	

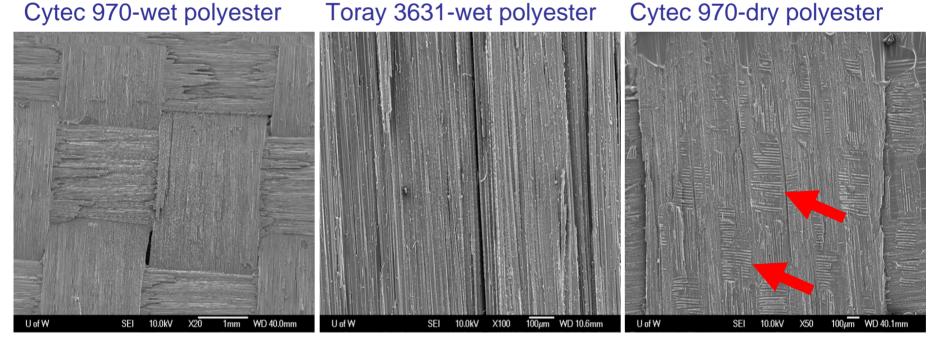
# SEM: post-RAT fracture

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- The images below are taken after the adhesively bonded samples are failed by the Mode I fracture Rapid Adhesion Test
- Prepregs which had been prepared with the Henkel EA-9895 epoxy-preimpregnated polyester peel ply all demonstrated the highly desirable 100% cohesive character in their failure mode





# Task 2 Discussion



- The Henkel EA-9895 peel ply-prepared surfaces performed well in actual adhesive bond quality and also contained the adhesive compounds within the wetting envelopes; surfaces after peel ply removal exhibited fractured epoxy regions and no visible fiber remnants
- Although surfaces prepared with the dry polyester peel ply contained the adhesives well within their wetting envelopes, they did not show substantial cohesive character in failure
- Cytec 970 prepared with the wet nylon peel ply and 3M AF 555 adhesive showed cohesive failure even though the adhesive was outside the boundaries of the wetting envelope
- Surfaces which had visible peel ply contamination when observed by SEM did not produce strong bonds



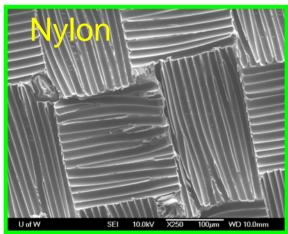
- A given peel ply surface preparation that works with one prepreg-adhesive system will not necessarily work with any other prepreg-adhesive system; each combination yields its own unique characteristics
- Henkel EA9895 epoxy-preimpregnated polyester peel ply produced high quality bonds in all of the systems investigated
- Surface wetting is a necessary but insufficient condition for the formation of strong adhesive bonds in the composites tested
- High O/C or N/C ratio's did not correlate to bond quality.

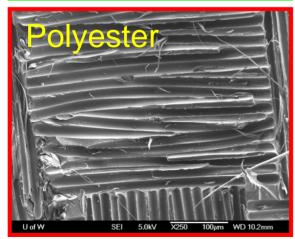
# JMS Peel Ply Surface Prep. - SEM Results Summary



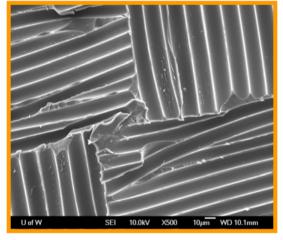
### Composite surface after removal of dry peel plies:

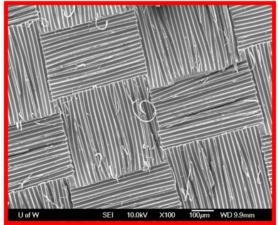
260 F cure GFRP



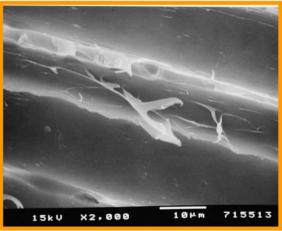


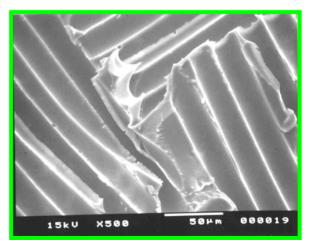
#### Cytec 970 (360F)





#### Toray 3900 (360 F)







- Bonding Depends on
  - Prepreg system (Resin and Fiber(?))
  - Peel Ply Material and Source
  - Adhesive
- Characterization Techniques (XPS, SEM and Surface Energy) provide useful information to help understand bonding requirements



- Further dissemination and acceptance of quick, inexpensive bond quality test- RAT method
- Initial stages of prepreg-peel ply-adhesive compatibility data base
- Contribute to fundamental understanding necessary to develop inspection techniques to determine the suitability of peel ply surfaces for bonding



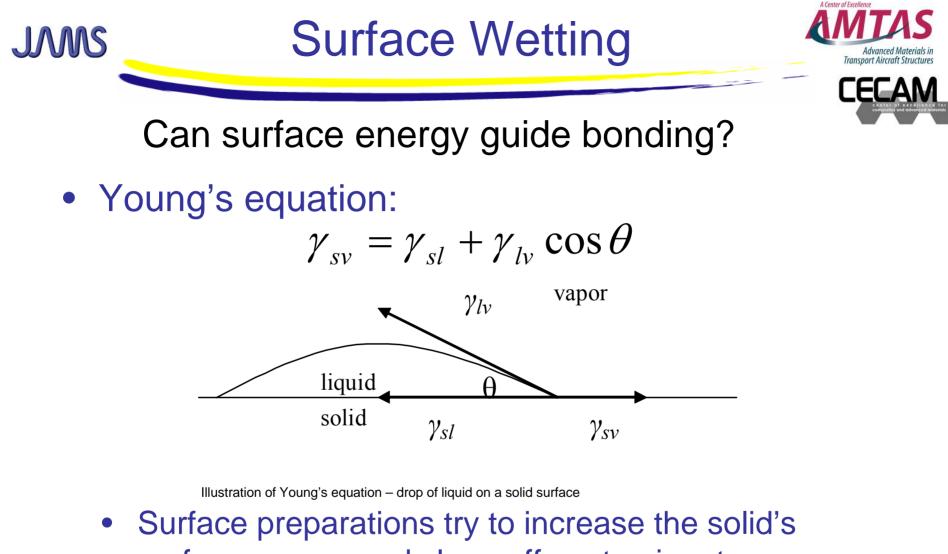
- Continued exploration of the correlations between surface characterization and the actual bond quality as determined experimentally is needed
- Each characterization method may provide criteria which are exclusive to the formation of good bonds though no single technique can currently explain whether or not a good bond will result
- Determining the quality of a bond using theoretical means with close correlation to experimental results is desired
- Even more tantalizing is the possibility of realizing new material combinations which could produce stronger bonds



- Benefit to Aviation
  - Better understanding of peel ply surface prep.
  - Guide development of QA methods for surface prep.
  - Greater confidence in adhesive bonds
- Future needs
  - Contact angle (wetting) vs. bond quality
  - Does fiber type (glass, pitch, PAN) effect bonding?
  - Peel ply-resin interactions
  - Applicability to other composite and adhesive (paste) systems
  - Model to guide bonding based on characterization, surface prep. and material properties



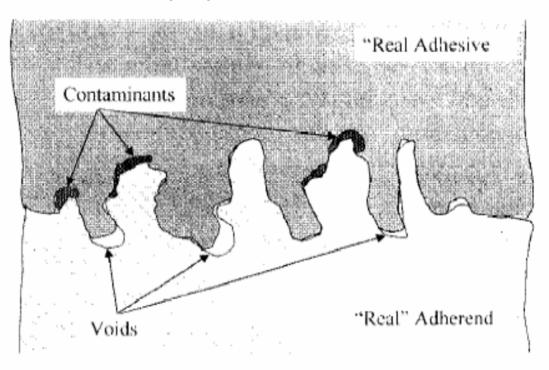
- Funding FAA JAMS-AMTAS
- Peter Van Voast & Will Grace at The Boeing Company
- Mark Tuttle for his technical input and "WET" software utility
- Material donations from Cytec-Cycom, Toray Composites America, Airtech International, Henkel, Richmond Aerospace, Yokahama and Precision Fabrics
- UW- MSE Undergraduates: Rockey Aye, Eric Brutke, Neil Golke, Dinda Padmasana



- surface energy and clean off contaminants
- Contaminants lower the solid's surface energy
- Complete wetting means θ approaches zero The Joint Advanced Materials and Structures Center of Excellence



Weak boundary layer due to voids and contaminants



## > Wetting required for good bond

Figure from Pocius, A., Adhesion and Adhesives Technology: An Introduction, 2nd ed., 2002, Hanser Gardner, New York.



- Is there a relation between surface energy and bond quality?
- Measuring the contact angles of multiple test liquids on the prepared surfaces allowed calculation of the substrates' surface energy
- The two-parameter Owens and Wendt model of surface energy was used, with a polar and dispersive component

$$\gamma = \gamma^{d} + \gamma^{p}$$
  

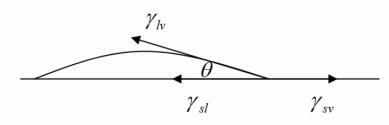
$$\gamma_{sl} = \gamma_{sv} + \gamma_{lv} - 2\sqrt{\gamma_{sv}^{d}\gamma_{lv}^{d}} - 2\sqrt{\gamma_{sv}^{p}\gamma_{lv}^{p}}$$

• Peel-ply prepared surfaces were measured (before adhesives were added)

# Some Examination by contact angle

## Contact angle analysis by goniometer

- Various fluids are used in a Ramé-Hart Tilting Contact Angle Goniometer, model 100-00 115, to form small droplets on the peel ply-prepared surfaces
- The contact angle these fluids form with the surface is recorded and applied to Owen and Wendt's surface energy model to generate a Kaelble plot
- The Kaelble plot allows determination of the polar and dispersive surface energy components of the solid by linear regression; a wettability envelope may then be generated



 $\frac{\sigma_l(\cos\theta+1)}{2\sqrt{\sigma_l^d}} = \sqrt{\sigma_s^p} \left(\sqrt{\frac{\sigma_l^p}{\sigma_l^d}}\right) + \sqrt{\sigma_s^d}$ 



- Subsequent to the development of the Kaelble plots, wettability envelopes were generated with BKCWet 1.1, a program initially devised by Mark Tuttle and modified by Brian Clark, both of the University of Washington
- It is supposed that any fluid whose dispersive and polar surface energies plot its point within the wetting envelope of a solid will wet out on the surface
- Points excluded from a wetting envelope are assumed not to spontaneously wet out on the surface
- The reality is that the break-even point of energetic favorability represented by the wetting envelope is not always a guaranteed predictor of bond quality



A liquid epoxy adhesive is on each surface.

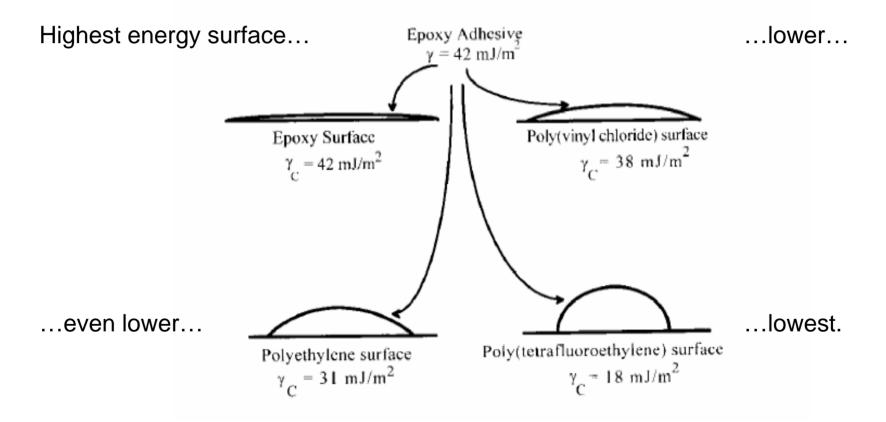


Figure from Pocius, A., Adhesion and Adhesives Technology: An Introduction, 2nd ed., 2002, Hanser Gardner, New York. The Joint Advanced Materials and Structures Center of Excellence 51