



# TenCate BT250E-6 AS4C 3k-PW Fabric Gr 195 gsm 40% RC Qualification Material Property Data Report

FAA Special Project Number: TD03019RC-R

NCAMP Test Report Number: CAM-RP-2015-039 N/C

Report Date: October 24, 2017

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**REVISIONS:**

Rev	By	Date	Pages Revised or Added
N/C	Vinsensius Tanoto	10/24/2017	Document Initial Release

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## 1. Introduction

### 1.1 Scope

The test methods and results described in this document are intended to provide basic composite properties essential to most methods of analysis and are consistent with CMH-17-1G—Composite Materials Handbook for Polymer Matrix Composites. This report contains material property data of common usefulness to wide range of projects. The lamina material property data have been generated with FAA oversight through FAA Special Project Number TD03019RC-R, and also meet the requirements of NCAMP Standard Operating Procedure NSP100; the test panels and test specimens have been inspected by an FAA Designated Airworthiness Representative (DAR) and the testing has been witnessed by an FAA Designated Engineering Representative (DER). However, the data may not fulfill all the needs of any specific company's program; specific properties, environments, laminate architecture, and loading situations may require additional testing.

The use of NCAMP material and process specifications does not guarantee material or structural performance. Material users should be actively involved in evaluating material performance and quality including, but not limited to, performing regular purchaser quality control tests, performing periodic equivalency/additional testing, participating in material change management activities, conducting statistical process control, and conducting regular supplier audits.

The applicability of NCAMP material property data, material allowables, and specifications must be evaluated on a case-by-case basis by aircraft companies and certifying agencies. NCAMP assumes no liability whatsoever, expressed or implied, related to the use of the material property data, material allowables, and specifications.

This report contains material property data only. Statistical analysis of the data including the calculations of b-basis values is given in a separate report, TenCate Advance Composites AS4C 3k-PW Fabric with BT250E-6 Resin Material Allowables Statistical Analysis Report, NCP-RP-2015-020 Rev N/C. The qualification material was procured to Erickson Air-Crane (EAC) Material Specification ES0095 Revision B dated May 22, 2013. An equivalent NCAMP Material Specification NMS 250/2 Rev Initial Release dated January 2, 2018 has been created, which contains specification limits that are derived from guidelines in DOT/FAA/AR-03/19. The qualification test panels were cured in accordance with Erickson Air-Crane (EAC) Process Specification ES0098 Revision A dated June 15, 2011. An equivalent NCAMP Process Specification NPS 81250 baseline "C" Cure Cycle Rev Initial Release dated October 20, 2017 has been created. The Erickson Air-Crane (EAC) test plan EAC2028 Rev C was used for this qualification program.

Part fabricators that wish to utilize the material property data, allowables, and specifications may be able to do so by demonstrating the capability to reproduce the

original material properties; a process known as equivalency. More information about this equivalency process including the test statistics and its limitations can be found in Section 6 of DOT/FAA/AR-03/19 and Section 8.4.1 of CMH-17-1G. The applicability of equivalency process must be evaluated on program-by-program basis by the applicant and certifying agency. The applicant and certifying agency must agree that the equivalency test plan along with the equivalency process described in Section 6 of DOT/FAA/AR-03/19 and Section 8.4.1 of CMH-17-1G are adequate for the given program.

Aircraft companies should not use the data published in this report without specifying NCAMP Material Specification NMS 250/2. NMS 250/2 may have additional requirements that are listed in its prepreg process control document (PCD), fiber specification, fiber PCD, and other raw material specifications and PCDs which impose essential quality controls on the raw materials and raw material manufacturing equipment and processes. *Aircraft companies and certifying agencies should assume that the material property data published in this report is not applicable when the material is not procured to NMS 250/2.* NMS 250/2 is a free, publicly available, non-proprietary aerospace industry material specification.

The data in this report is intended for general distribution to the public, either freely or at a price that does not exceed the cost of reproduction (e.g. printing) and distribution (e.g. postage).

## 1.2 Symbols Used

$\nu_{12}^t$	major Poisson's ratio, tension
$\mu\varepsilon$	micro-strain
$E_1^c$	compressive modulus, longitudinal / warp direction
$E_1^t$	tensile modulus, longitudinal / warp direction
$E_2^c$	compressive modulus, transverse / fill direction
$E_2^t$	tensile modulus, transverse / fill direction
$F_1^{cu}$	ultimate compressive strength, longitudinal / warp direction
$F_1^{tu}$	ultimate tensile strength, longitudinal / warp direction
$F_2^{cu}$	ultimate compressive strength, transverse / fill direction
$F_2^{tu}$	ultimate tensile strength, transverse / fill direction
$\nu_{12}^c$	major Poisson's Ratio, compression
$\nu_{21}^c$	minor Poisson's Ratio, compression
$F_{12}^{s5\% \text{ strain}}$	in-plane shear strength at 5% strain
$F_{12}^{smax}$	in-plane shear peak strength before 5% strain
$F_{12}^{s0.2\%}$	in-plane shear strength at 0.2% offset
$G_{12}^s$	in-plane shear modulus

**Superscripts**

c	compression
cu	compression ultimate
s	shear
su	shear ultimate
t	tension
tu	tension ultimate

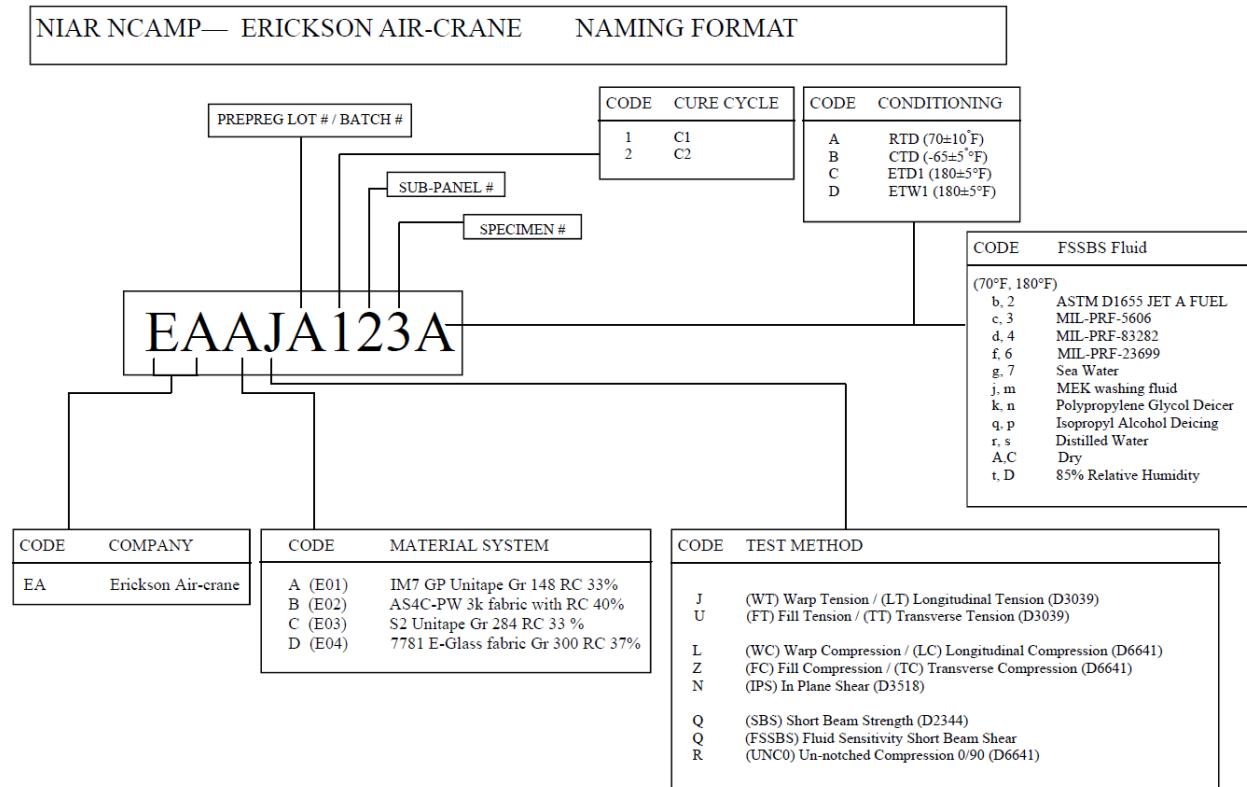
**Subscripts**

1	axis; longitudinal / warp direction (parallel to warp direction of reinforcement)
2	axis; transverse / fill direction (parallel to fill direction of reinforcement)
12	in-plane

**Acronyms and Definitions**

ASTM	American Society for Testing and Materials
B – Basis	95% lower confidence limit on the tenth population percentile
CV	Coefficient of variation
CTD	cold temperature dry
CPT	cured ply thickness
ETD	elevated temperature dry
ETW	elevated temperature wet
Gr/Ep	graphite/epoxy
norm	normalized
RTD	room temperature dry
SACMA	Suppliers of Advanced Composite Materials Association
SRM	SACMA Recommended Method
Tply	thickness divided by the number of plies provides the thickness average per specimen
wet	specimen with an “equilibrium” moisture content
T, RH	temperature, relative humidity

### 1.3 NIAR-Specimen Naming Format



**Figure 1-1: Naming Format**

## 1.4 References

### **ASTM Standards**

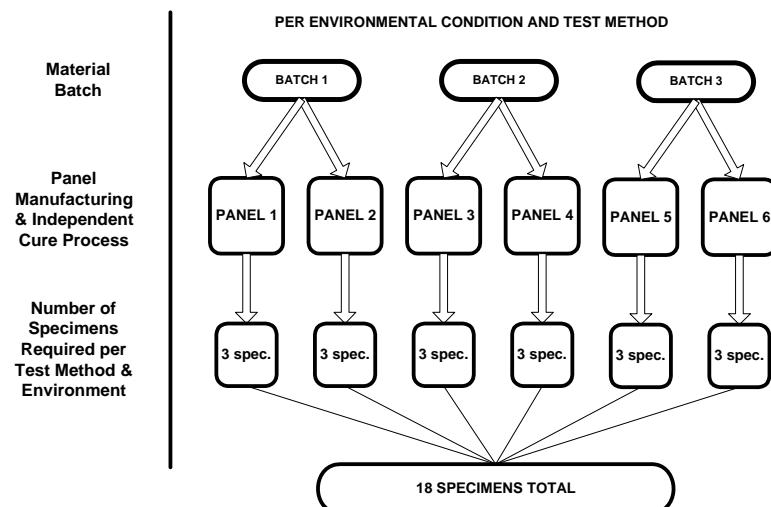
All testing was in accordance with nationally recognized standards, methods and procedures. Specific mechanical property test methods applicable to the test program in this document include:

- ASTM D2344/D2344M-00(2006) – Standard Test Method for Short-Beam Strength of Polymer Matrix Composite Materials and Their Laminates
- ASTM D3039/D3039M-08 – Standard Test Method for Tensile Properties of Polymer Matrix Composite Materials
- ASTM D3518/D3518M-94(2007) – Standard Test Method for In-Plane Shear Response of Polymer Matrix Composite Materials by Tensile Test of a  $\pm 45^\circ$  Laminate In-Plane Shear Strength and Modulus
- ASTM D6641/D6641M-01e1(2009) – Standard Test Method for Determining the Compressive Properties of Polymer Matrix Composite Laminates Using a Combined Loading Compression (CLC) Test Fixture
- ASTM D7028-07e1 – Standard Test Method for Glass Transition Temperature (DMA Tg) of Polymer Matrix Composites by Dynamic Mechanical Analysis (DMA)

## 1.5 Methodology

### 1.5.1 Process Definition

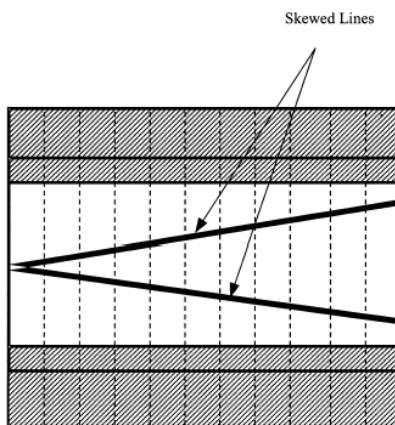
For each combination of test, batch and condition, the specimens were selected from minimum two separate panels cured separately as shown in Figure 1-2 unless otherwise specified.



**Figure 1-2: Specimen Selection Methodology**

All panels were fabricated in accordance with Erickson Air-Crane (EAC) Process Specification ES0098 which is equivalent to NCAMP Process Specification NPS 81250 baseline "C" Cure Cycle, caul plate is used.

In order to facilitate individual specimen traceability, individual specimen numbering and/or skewed lines were written or drawn across each sub-panel as shown in Figure 1-3.



**Figure 1-3: Specimen Traceability Line**

## 1.5.2 Specimen & Testing Details

### 1.5.2.1 Tabbing

No tabs were used for this program.

### 1.5.2.2 Specimen Dimensions & Test Configuration

For SBS specimens, a span of 4.5T and 4T were used for CTD, RTD and ETD conditions and a span of 4T was used for ETW condition, where T was the average thickness of six qualification panels. The same T was used to compute the width and length of the specimen.

Unless otherwise specified, a tolerance of  $\pm 5^{\circ}\text{F}$  applied to all temperature conditions specified in this document.

### 1.5.2.3 Specimen Strain Device Used

Corresponding Gage ID can be obtained from Appendix A of EAC2028 Rev C.

**Uniaxial gages** were used on:

All CTD Tension specimens

Two RTD Tension specimens for obtaining full stress strain curves

All conditions of combined loading compression specimens

**Biaxial gages** were used on:

All conditions of IPS specimens

**Uniaxial Extensometers** were used on:

All RTD and ETW Tension specimens

### 1.5.3 Test Matrix

The table below shows the lay-ups and test matrices used for lamina level testing.

Layup	Test Type and Direction	Property	Number of Batches x No. of Panels x No. of Specimens			
			Test Temperature/Moisture Condition			
			CTD	RTD (4)	ETD	ETW
[0] <sub>15</sub>	ASTM D3039 0° Tension	Strength, Modulus and Poisson's Ratio	3x2x3	3x2x3		3x2x3
[0] <sub>15</sub>	ASTM D6641 0° Compression	Strength and Modulus	3x2x3	3x2x3 (1)		3x2x3
[90] <sub>15</sub>	ASTM D3039 90° Tension	Strength and Modulus	3x2x3	3x2x3		3x2x3
[90] <sub>15</sub>	ASTM D6641 90° Compression	Strength and Modulus	3x2x3	3x2x3 (1)	1x2x3	3x2x3 (3)
[45/-45] <sub>3S</sub>	ASTM D3518 In-Plane Shear (2)	Strength and Modulus	3x2x3	3x2x3		3x2x3
[0] <sub>32</sub>	ASTM D2344 Short Beam	Strength	3x2x3	3x2x3	3x2x3	3x2x3

**Table 1-1: Lamina Level Test Matrix**

**Note 1:** Back-to-back strain gages are needed on the first two specimens of each environment. If no buckling is observed, the remaining modulus specimens will require a strain gage on one side of the specimens only. An appropriate extensometer may be used in place of the strain gage.

**Note 2:** Gripped (tab) length is  $1.5 \pm 0.5$ " on each end of the 10" long specimen. Once the samples have reached the 5% strain level, the actuator/crosshead displacement rate can be increased by four times the initial rate. Continue testing at the higher strain rate until ultimate failure is observed.

**Note 3:** If strain gage is used for modulus measurement, a separate un-gaged specimen must be used for strength measurement; because the strain gage and its protective coating may prevent moisture absorption in the gage area.

**Note 4:** At least two specimens must be gaged to obtain full stress-strain curve to failure. An appropriate extensometer may be used in place of the strain gage for the remaining specimens.

### 1.5.4 Cured Laminate Physical Testing

The properties in Table 1-2 were determined for each panel used for test coupons with the exception of Tg by DMA which were conducted on one laminate per batch from each oven cure conducted where that batch is present. The tests were performed by the National Institute for Aviation Research (NIAR) Composites Laboratory under the supervision of NCAMP.

Property	Condition/Method (Note 1)	Min Replicates per panel
Cured Ply Thickness	ASTM D3171-06 (Method II)	All data from mechanical test specimens
Laminate Density	ASTM D792-08	3
Fiber Volume, % by Volume	ASTM D3171-06 (Note 2)	3
Resin Content, % by Weight	ASTM D3171-06 (Note 2)	3
Ultrasonic Through Transmission, C-Scan	CP6121 (Note 3)	1
Glass Transition Temperature, Tg by DMA flexural loading	Dry and Wet – ASTM D7028	1 Dry, 1 Wet

**Table 1-2: Physical Testing Matrix**

**Note 1:** Where the applicable standard allows variations in specimen form or test method, the specific parameters to be used will be specified in the test work instructions and reported in the final test report.

**Note 2:** Method II, except for laminates of materials where actual fiber weight is not accurately known prior to impregnation, as in the case for unidirectional materials. For these materials, in order to verify Method II is accurate, a minimum of 12 samples per batch shall be tested by Method I, Procedure B.

**Note 3:** CP6121 is equivalent to MIL-HDBK-787A. Five MHz is preferred for solid laminates. Panels with anomaly should be segregated. Microscopy images may be taken from questionable areas. NCAMP must be involved in the review of all C-scans.

### 1.5.5 Environmental Conditioning

The following tests were performed by the NIAR Composites Laboratory under the supervision of NCAMP.

$$\begin{aligned} \text{CTD} &= -65 \pm 5^\circ\text{F}, \text{ dry} \\ \text{RTD} &= 70 \pm 10^\circ\text{F}, \text{ dry} \\ \text{ETD} &= 180 \pm 5^\circ\text{F}, \text{ dry} \\ \text{ETW} &= 180 \pm 5^\circ\text{F}, \text{ wet} \end{aligned}$$

Within each test method and test environment, the failure mode was evaluated immediately after each test by a delegated FAA DER or ACO engineer or an NCAMP staff engineer. All tested specimens were digitally photographed after each test in order to pictorially document failure modes.

For dry testing, specimens were dried at  $160^\circ\text{F} \pm 5^\circ\text{F}$  for 120 to 130 hours. After drying, specimens were kept in a desiccator until mechanical testing. Alternatively, the specimens may have been left in ambient laboratory condition for a maximum of 14 days until mechanical testing (no drying was required if specimens were tested within 14 days from the date they were cured). Ambient laboratory condition is defined as  $70^\circ\text{F} \pm 10^\circ\text{F}$ . Since moisture absorption and desorption rate for epoxy is very slow at ambient temperature, there was no requirement to maintain relative humidity levels.

For wet conditioning, specimens were dried at  $160^\circ\text{F} \pm 5^\circ\text{F}$  for 120 to 130 hours before being conditioned to equilibrium at  $160^\circ\text{F} \pm 5^\circ\text{F}$  and  $85\% \pm 5\%$ . Effective moisture equilibrium was achieved when the average moisture content of the traveler specimen changed by less than 0.02% for two consecutive readings which are  $7 \pm 0.5$  days apart and may be expressed by:

$$\frac{W_i - W_{i-1}}{W_b} < 0.0002$$

Where:

- $W_i$  = weight at current time
- $W_{i-1}$  = weight at previous time
- $W_b$  = baseline weight prior to conditioning

When representative specimens could not be measured to determine the moisture content (due to size, fastener and tab effects), traveler coupons of at least 1" by 1" by specimen thickness and weighing at least 15 grams were used to establish weight gain measurements. If the specimens or traveler coupons pass the criteria for two consecutive readings which are  $7 \pm 0.5$  days apart, the specimens were kept in the environmental chamber for up to an additional 60 days. Alternatively, the specimens may have been removed from the environmental chamber and placed in a sealed plastic bag along with a moist cotton towel for a maximum of 14 days until mechanical testing. Strain-gaged specimens were removed from the controlled environment for a maximum of 2 hours for application of gages in ambient laboratory conditions.

### 1.5.6 Non-ambient Testing

The chamber was of adequate size so that all test fixtures and load frame grips were contained within the chamber. For elevated temperature testing, the temperature chamber, test fixture, and grips were preheated to the specified temperature. Each specimen was heated to the required test temperature as verified by a thermocouple in direct contact with and taped to the specimen gage section. The heat-up time of the specimen did not exceed 5 minutes, unless otherwise specified in individual test summary sheets. The test was started  $5^{+1}_{-0}$  minutes after the specimen reached the test temperature. During the test, the temperature, as measured on the specimen, was within  $\pm 5^{\circ}\text{F}$  of the required test temperature.

For subzero temperature testing, each specimen was cooled to the required test temperature as verified by a thermocouple in direct contact with and taped to the specimen gage section. The test started  $5^{+1}_{-0}$  minutes after the specimen reached the test temperature. During the test, the temperature, as measured on the specimen, was within  $\pm 5^{\circ}\text{F}$  of the required test temperature.

For wet specimens, the moisture loss was determined by subjecting representative specimens to the same amount of time required to heat-up and fail the specimens. For filled-hole or bearing specimens, fasteners were removed prior to conducting moisture loss measurements. For tabbed specimens, representative coupons without tabs and having the same number of plies were used to conduct the moisture loss measurements. A minimum of one specimen or representative coupon was used to measure the moisture loss for every combination of test temperature and stacking sequence.

### **1.5.7 Normalization Procedures**

Most lamina level tension and compression strength and modulus properties, and all laminate level properties were normalized according to nominal cured ply thickness. Lamina level properties that were not normalized include 90° tensile strength and modulus (unidirectional only), 90° compressive strength and modulus (unidirectional only), in-plane shear strength and modulus, Poisson's ratio, SBS, and ILT. After normalizing, data scatter reduced or remained the same. If data scatter increased significantly after normalizing, the reason was investigated. Wherever properties are normalized, both measured and normalized data were reported.

The theoretically calculated cured ply thickness was 0.0075 inches. The experimentally Out-of-Autoclave measured cured ply thickness of 0.0085 inches has been used as the nominal cured ply thickness (CPT) for normalization purpose. This has been done at the request of the material supplier. The following normalization formula was used:

$$\text{Normalized Value} = \text{Measured Value} \times \text{Measured CPT} / \text{Nominal CPT}.$$

### **1.5.8 Inspection Verification**

The 3-batch qualification panels have been fabricated according to the requirements of the test plan and conformed by an FAA DAR. The test specimens and test setup have also been conformed by an FAA DAR.

Testing was witnessed by FAA DER. Mechanical testing was carried out at the National Institute for Aviation Research, Wichita State University.

### **1.5.9 Material Pedigree Information**

The PMC Data Collection Template includes the material pedigree information required, such as material and batch information, as well as panel fabrication record, environmental conditioning, test equipment, and test procedures.

## 2. Test Results

### 2.1 Lamina Level Test Summary

<b>Prepreg Material:</b>	TenCate BT250E-6 AS4C 3k-PW Fabric 195gsm 40% RC				<b>TenCate BT250E-6 AS4C 3k-PW Fabric 195gsm 40% RC Lamina Properties Summary</b>			
<b>Material Specification:</b>	NMS 250/2							
<b>Process Specification:</b>	NPS 81250							
<b>Fiber:</b>	AS4C 3k-PW	<b>Resin:</b>	TenCate BT250E-6					
<b>Tg(dry):</b>	275.46 °F	<b>Tg(wet):</b>	249.55 °F	<b>Tg METHOD:</b> ASTM D7028				
Date of fiber manufacture	Batch 1	Batch 2	Batch 3	Date of testing	Nov 2011 - Apr 2012			
Date of resin manufacture *	May 5, 2011	July 14, 2009	April 5, 2011	Date of data submittal	May 12, 2012			
Date of prepreg manufacture	March 8, 2011	May 10, 2011	May 10, 2011					
Date of composite manufacture	June 13, 2011	May 31, 2011	May 31, 2011					
	Sept 2011 - Oct 2011							
<b>LAMINA MECHANICAL PROPERTY SUMMARY</b> Data reported as: Normalized & Measured (Normalized by CPT=0.0085 inch)								
	CTD Mean		RTD Mean		ETD Mean		ETW Mean	
	Normalized	Measured	Normalized	Measured	Normalized	Measured	Normalized	Measured
<b>F<sub>1</sub><sup>tu</sup> [ksi]</b>	125.113	124.754	132.294	131.913			124.698	123.157
<b>E<sub>1</sub><sup>t</sup> [Msi]</b>	8.631	8.592	8.560	8.536			8.558	8.452
<b>v<sub>12</sub><sup>t</sup></b>		0.058		0.055				0.051
<b>F<sub>2</sub><sup>tu</sup> [ksi]</b>	120.271	120.324	124.751	125.323			110.388	110.164
<b>E<sub>2</sub><sup>t</sup> [Msi]</b>	8.641	8.615	8.530	8.569			8.604	8.585
<b>F<sub>1</sub><sup>cu</sup> [ksi]</b>	102.760	101.147	93.092	92.800			56.398	55.442
<b>E<sub>1</sub><sup>c</sup> [Msi]</b>	8.060	7.937	7.905	7.890			7.987	7.973
<b>F<sub>2</sub><sup>cu</sup> [ksi]</b>	93.260	93.152	85.122	85.240	73.543	72.056	50.327	49.864
<b>E<sub>2</sub><sup>c</sup> [Msi]</b>	8.111	8.109	7.862	7.882	7.896	7.724	7.906	7.886
<b>F<sub>12</sub><sup>s0.2%</sup> [ksi]</b>		8.396		6.403				3.688
<b>F<sub>12</sub><sup>s5%strain</sup> [ksi]</b>		13.246		10.335				5.733
<b>G<sub>12</sub><sup>s</sup> [Msi]</b>		0.662		0.578				0.379
<b>SBS [ksi]</b>		8.466		8.157		7.109		5.044

\* Three unique resin lots were used, two of the resin lots were produced on the same day.

**Table 2-1: Lamina Summary Data**

## 2.2 Individual Test Summaries

### 2.2.1 Warp Tension Properties (WT)

Material:	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%				<b>Tension, 1-axis</b>				
Resin content:	39.33 % wt	Comp. density: 1.485 g/cc				TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [0]15			
Fiber volume:	50.62 % vol								
Ply count:	15								
Test method:	ASTM D 3039-08	Modulus calculation: 1000 to 3000 microstrain							
Normalized by:	0.0085 in. CPT								
	CTD	RTD	ETW						
Test Temperature [°F]	-65	70	180						
Moisture Conditioning	Dry	Dry	Equilibrium						
Equilibrium at T, RH			160 F, 85%						
Source code	EABJX XXXB	EABJX XXXA	EABJX XXXD						
	Normalized	Measured	Normalized	Measured	Normalized	Measured			
$F_t^{tu}$ [ksi]	Mean Minimum Maximum C.V.(%)	125.113 110.920 137.895 5.388	124.754 110.386 135.538 5.829	132.294 112.881 141.459 4.792	131.913 112.748 141.422 4.959	124.698 117.096 129.636 3.017	123.157 113.780 131.283 4.015		
	No. Specimens	20		22		25			
	No. Prepreg Lots	3		3		3			
$E_t^t$ [Ms]	Mean Minimum Maximum C.V.(%)	8.631 8.485 8.829 0.981	8.592 8.257 9.004 2.504	8.560 8.468 8.698 0.663	8.536 8.212 8.911 2.205	8.558 8.411 8.747 0.994	8.452 8.057 8.715 2.118		
	No. Specimens	18		22		28			
	No. Prepreg Lots	3		3		3			
$v_{12}^t$	Mean No. Specimens No. Prepreg Lots		0.058		0.055		0.051		
		13		21		15			
		3		3		3			

## 2.2.2 Fill Tension Properties (FT)

Material:	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%				<b>Tension, 2-axis</b>			
Resin content:	38.25 % wt		Comp. density:	1.489 g/cc				
Fiber volume:	51.64 % vol				TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [90]15			
Ply count:	15							
Test method:	ASTM D 3039-08		Modulus calculation: 1000 to 3000 microstrain					
Normalized by:	0.0085 in. CPT		CTD	RTD	ETW			
Test Temperature [°F]	-65		70	180				
Moisture Conditioning	Dry		Dry	Equilibrium				
Equilibrium at T, RH				160 F, 85%				
Source code	EABUX XXXB		EABUX XXXA	EABUX XXXD				
	Normalized	Measured	Normalized	Measured	Normalized	Measured		
<b>F<sub>2</sub><sup>tu</sup> [ksi]</b>	Mean Minimum Maximum C.V.(%)	120.271 111.238 132.730 4.488	120.324 109.120 133.218 5.194	124.751 114.261 133.012 3.967	125.323 114.351 132.998 4.373	110.388 99.509 121.328 5.533	110.164 98.594 122.561 6.081	
	No. Specimens	21		21		21		
	No. Prepreg Lots	3		3		3		
<b>E<sub>2</sub><sup>t</sup> [Ms]</b>	Mean Minimum Maximum C.V.(%)	8.641 8.482 8.783 1.021	8.615 8.404 8.746 0.978	8.530 8.215 8.737 1.385	8.569 8.195 9.003 1.983	8.604 8.421 8.790 1.286	8.585 8.363 8.871 1.618	
	No. Specimens	18		21		21		
	No. Prepreg Lots	3		3		3		

## 2.2.3 Warp Compression Properties (WC)

<b>Material:</b>	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%		<b>Compression, 1-axis</b>				
<b>Resin content:</b>	39.64 % wt	<b>Comp. density:</b>	1.484 g/cc	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [0]15			
<b>Fiber volume:</b>	50.33 % vol						
<b>Ply count:</b>	15						
<b>Test method:</b>	ASTM D 6641-09		<b>Modulus calculation:</b> 1000 to 3000 microstrain				
<b>Normalized by:</b>	0.0085	in. CPT					
	<b>CTD</b>		<b>RTD</b>				
<b>Test Temperature [°F]</b>	-65		70				
<b>Moisture Conditioning</b>	Dry		Dry				
<b>Equilibrium at T, RH</b>			Equilibrium 160 F, 85%				
<b>Source code</b>	EABLX XXXB		EABLX XXXA				
	<b>Normalized</b>	<b>Measured</b>	<b>Normalized</b>	<b>Measured</b>			
<b>F<sub>1</sub><sup>cu</sup> [ksi]</b>	Mean Minimum Maximum C.V.(%)	102.760 89.515 113.672 6.254	101.147 87.625 112.219 6.677	93.092 82.364 100.158 5.224	92.800 81.462 101.045 5.359	56.398 49.677 61.373 5.942	55.442 47.952 60.572 6.236
	<b>No. Specimens</b>	22		21		21	
	<b>No. Prepreg Lots</b>	3		3		3	
<b>E<sub>1</sub><sup>c</sup> [Ms]</b>	Mean Minimum Maximum C.V.(%)	8.060 7.819 8.391 1.572	7.937 7.699 8.214 1.519	7.905 7.783 8.067 1.019	7.890 7.690 8.106 1.593	7.987 7.622 8.172 1.760	7.973 7.550 8.282 2.294
	<b>No. Specimens</b>	18		18		21	
	<b>No. Prepreg Lots</b>	3		3		3	

## 2.2.4 Fill Compression Properties (FC)

Material:	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%				<b>Compression, 2-axis</b>			
Resin content:	39.10 % wt		Comp. density:	1.486 g/cc				
Fiber volume:	50.83 % vol			TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [90]15				
Ply count:	15							
Test method:	ASTM D 6641-09		Modulus calculation:	1000 to 3000 microstrain				
Normalized by:	0.0085 in. CPT							
	CTD		RTD	ETD		ETW		
Test Temperature [°F]	-65		70	180		180		
Moisture Conditioning	Dry		Dry	Dry		Equilibrium		
Equilibrium at T, RH						160 F, 85%		
Source code	EABZX XXXB		EABZX XXXA	EABZX XXXC		EABZX XXXD		
	Normalized	Measured	Normalized	Measured	Normalized	Measured		
<b>F<sub>2</sub><sup>cu</sup> [ksi]</b>	Mean	93.260	93.152	85.122	85.240	73.543		
	Minimum	82.771	82.690	78.582	79.502	64.866		
	Maximum	103.399	104.268	93.019	92.628	64.293		
	C.V. (%)	6.828	7.194	5.207	4.832	7.115		
	No. Specimens	21		21		7		
	No. Prepreg Lots	3		3		1		
<b>E<sub>2</sub><sup>c</sup> [Ms]</b>	Mean	8.111	8.109	7.862	7.882	7.724		
	Minimum	7.688	7.691	7.576	7.661	7.629		
	Maximum	8.439	8.526	8.450	8.596	8.157		
	C.V. (%)	2.181	2.873	2.269	2.676	3.329		
	No. Specimens	18		18		6		
	No. Prepreg Lots	3		3		1		
						21		
						3		

## 2.2.5 In-Plane Shear Properties (IPS)

Material:	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%			In-Plane Shear	
Resin content:	38.29 % wt			Comp. density: 1.465 g/cc	
Fiber volume:	50.73 % vol			TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [45/-45]3S	
Ply count:	12				
Test method:	ASTM D 3518-07			Modulus calculation: 2000 to 6000 microstrain	
Normalized by:	NA				
	CTD	RTD	ETW		
Test Temperature [°F]	-65	70	180		
Moisture Conditioning	Dry	Dry	Equilibrium		
Equilibrium at T, RH			160 F, 85%		
Source code	EABNX XXXB	EABNX XXXA	EABNX XXXD		
	Normalized	Measured	Normalized	Measured	
$F_{12}^{s0.2\%}$ [ksi]	Mean Minimum Maximum C.V.(%)	8.396 7.957 8.724 2.348	6.403 6.134 6.583 2.207	3.688 3.497 3.873 3.004	
	No. Specimens No. Prepreg Lots	21 3	21 3	21 3	
$F_{12}^{s5\%strain}$ [ksi]	Mean Minimum Maximum C.V.(%)	13.246 12.517 13.973 2.753	10.335 10.050 10.779 1.991	5.733 5.337 6.010 3.319	
	No. Specimens No. Prepreg Lots	21 3	21 3	20 3	
$G_{12}^s$ [Ms]	Mean Minimum Maximum C.V.(%)	0.662 0.629 0.692 2.460	0.578 0.550 0.597 2.387	0.379 0.364 0.398 2.760	
	No. Specimens No. Prepreg Lots	21 3	21 3	21 3	

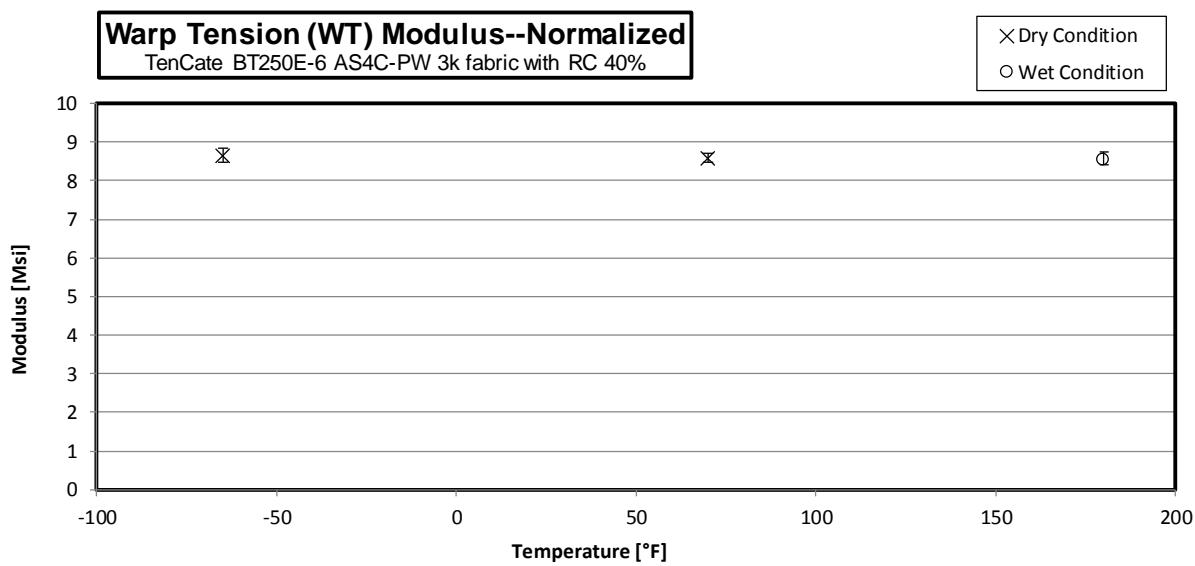
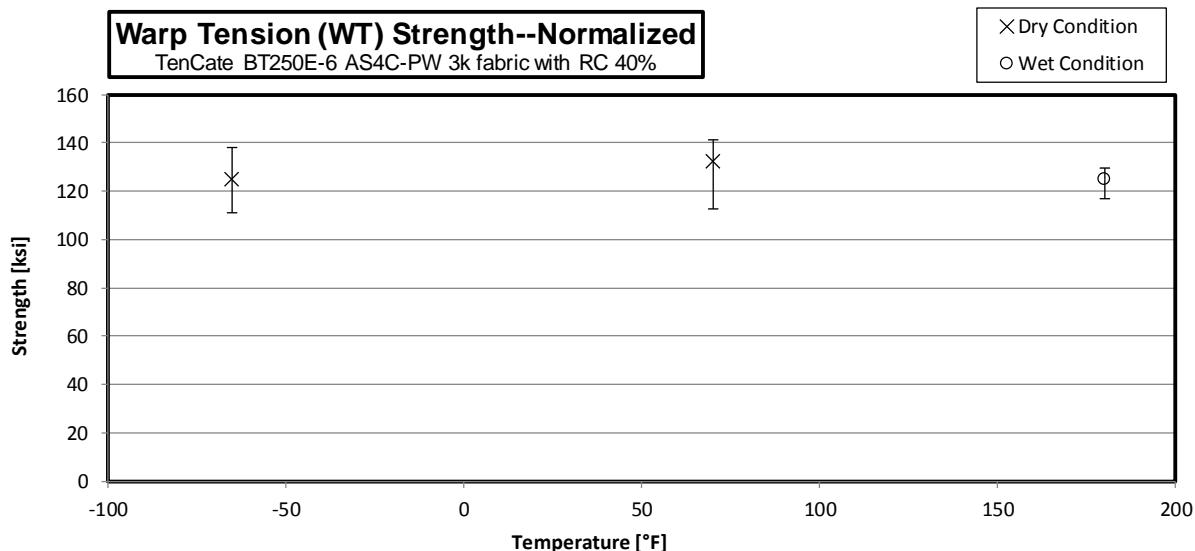
## 2.2.6 Lamina Short-Beam Strength Properties (SBS)

<b>Material:</b>	TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%				<b>Short-Beam Strength</b>	
<b>Resin content:</b>	39.04 % wt		<b>Comp. density:</b>	1.520 g/cc		TenCate BT250E-6 AS4C-PW 3k fabric with RC 40% [0]32
<b>Fiber volume:</b>	51.87 % vol					
<b>Ply count:</b>	32					
<b>Test method:</b>	ASTM D 2344-06					
<b>Normalized by:</b>	NA					
	<b>CTD</b>		<b>RTD</b>	<b>ETD</b>		<b>ETW</b>
<b>Test Temperature [°F]</b>	-65		70	180		180
<b>Moisture Conditioning</b>	Dry		Dry	Dry		Equilibrium
<b>Equilibrium at T, RH</b>						160 F, 85%
<b>Source code</b>	EABQX XXXB		EABQX XXXA	EABQX XXXC		EABQX XXXD
	<b>Normalized</b>	<b>Measured</b>	<b>Normalized</b>	<b>Measured</b>	<b>Normalized</b>	<b>Measured</b>
<b>SBS [ksi]</b>	Mean	8.466		8.157	7.109	5.044
	Minimum	7.885		7.745	6.443	4.597
	Maximum	8.948		8.896	7.554	5.630
	C.V. (%)	3.323		3.458	4.617	4.444
	No. Specimens	22		23	21	22
	No. Prepreg Lots	3		3	3	3

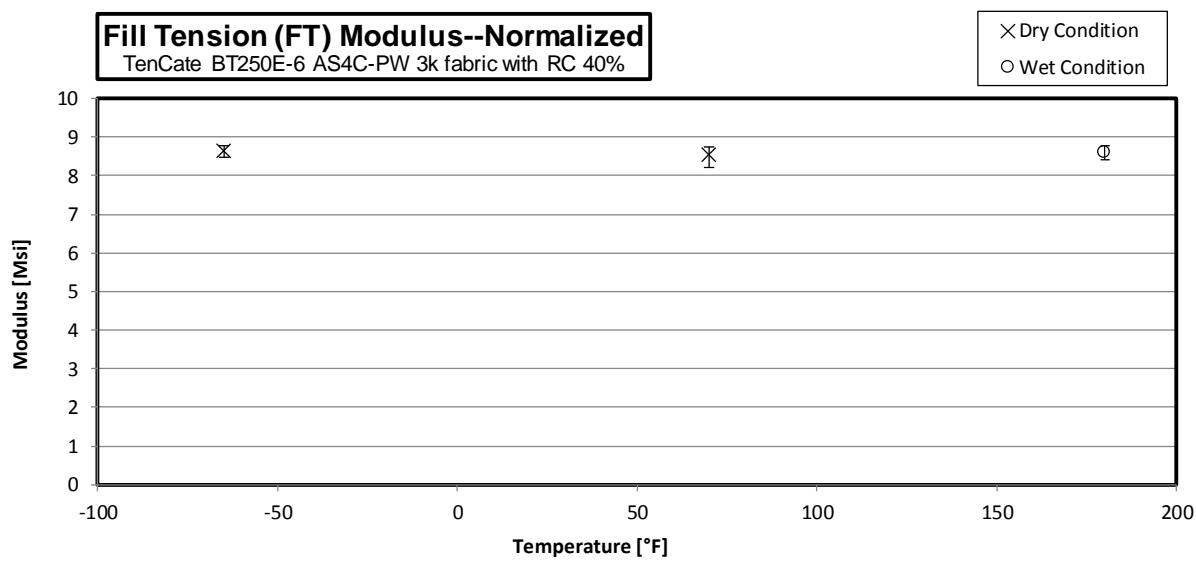
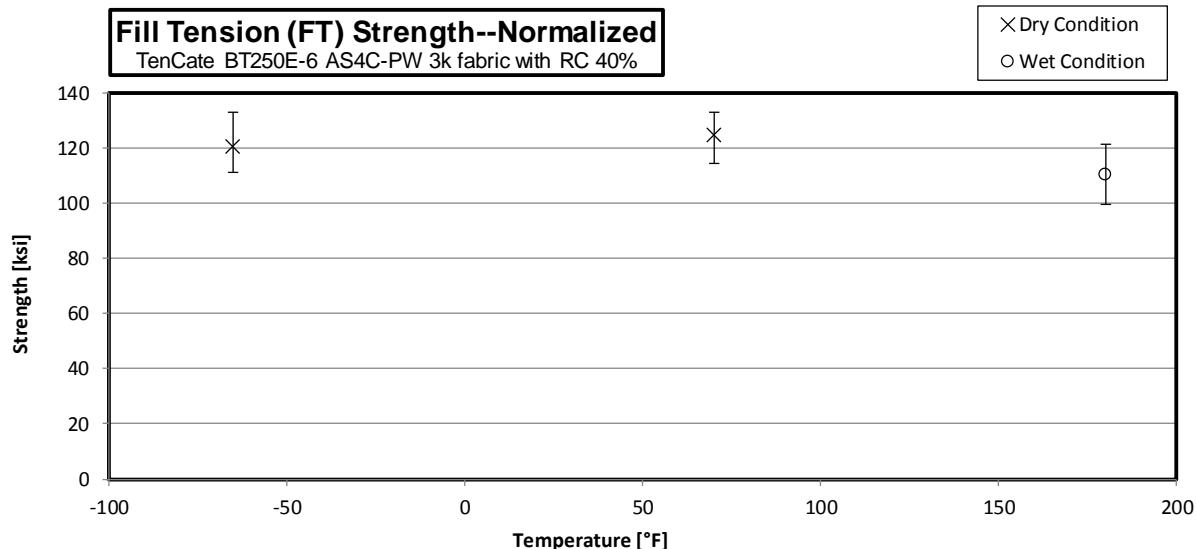
### **3. Individual Test Charts**

These charts combine all three batches of data and plot the minimum and maximum modulus and strength range based on the test temperature.

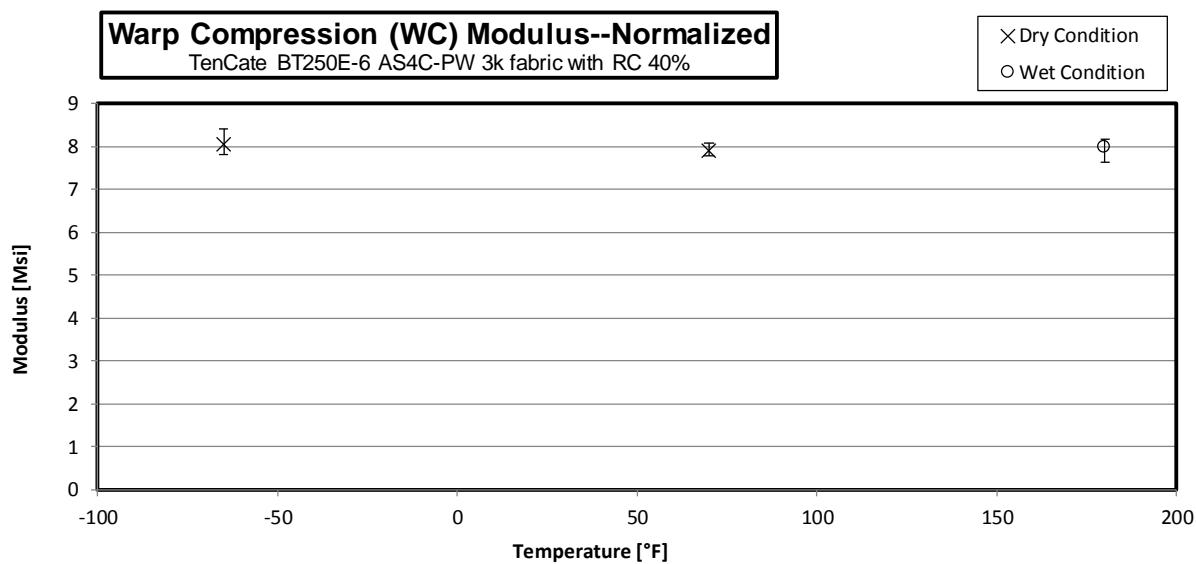
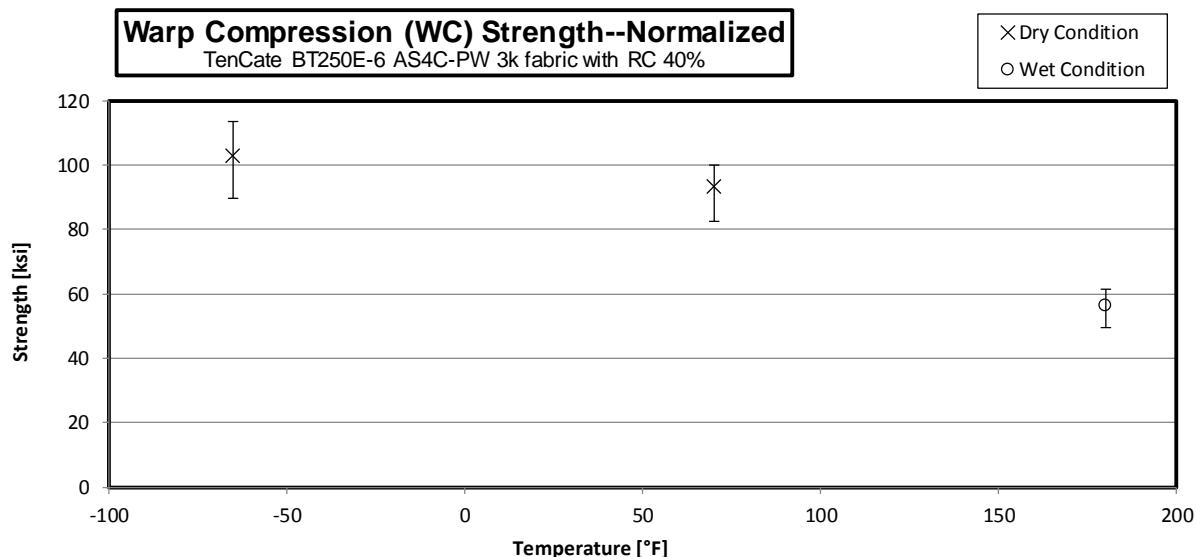
### 3.1 Warp Tension Properties (WT)



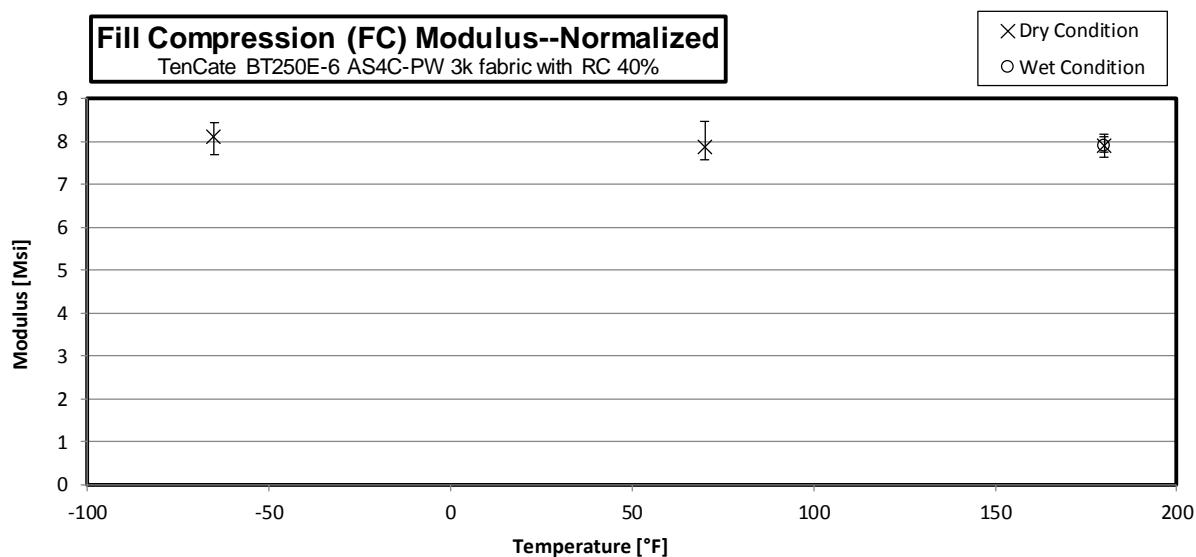
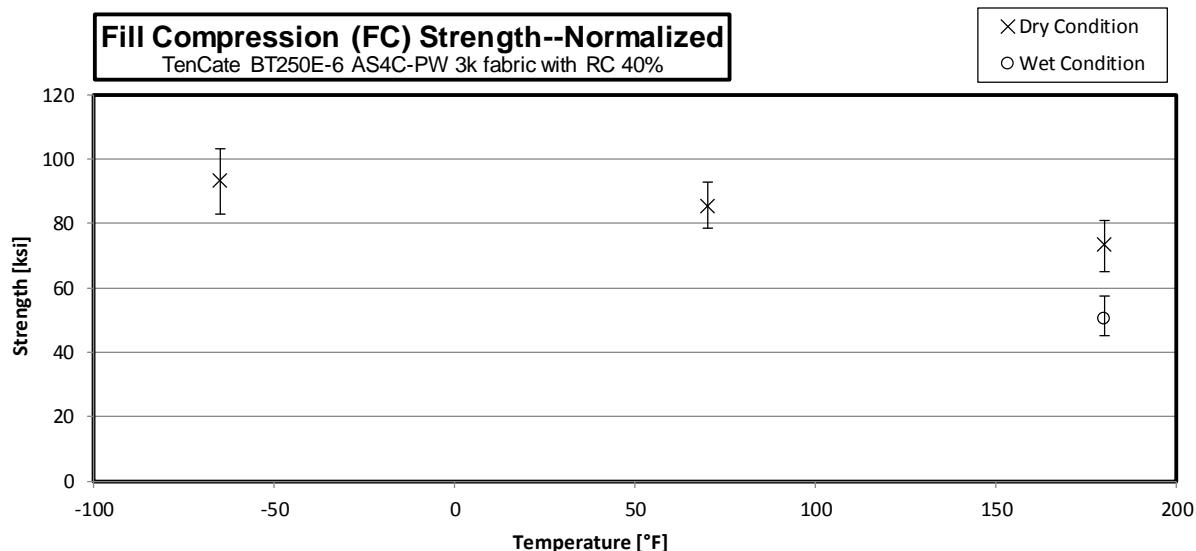
### 3.2 Fill Tension Properties (FT)



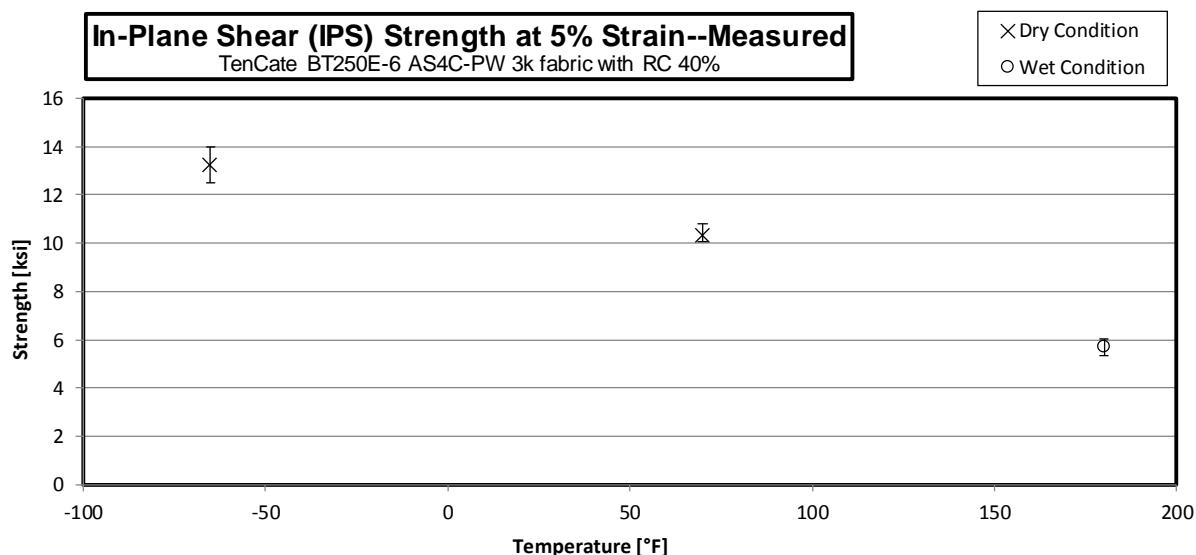
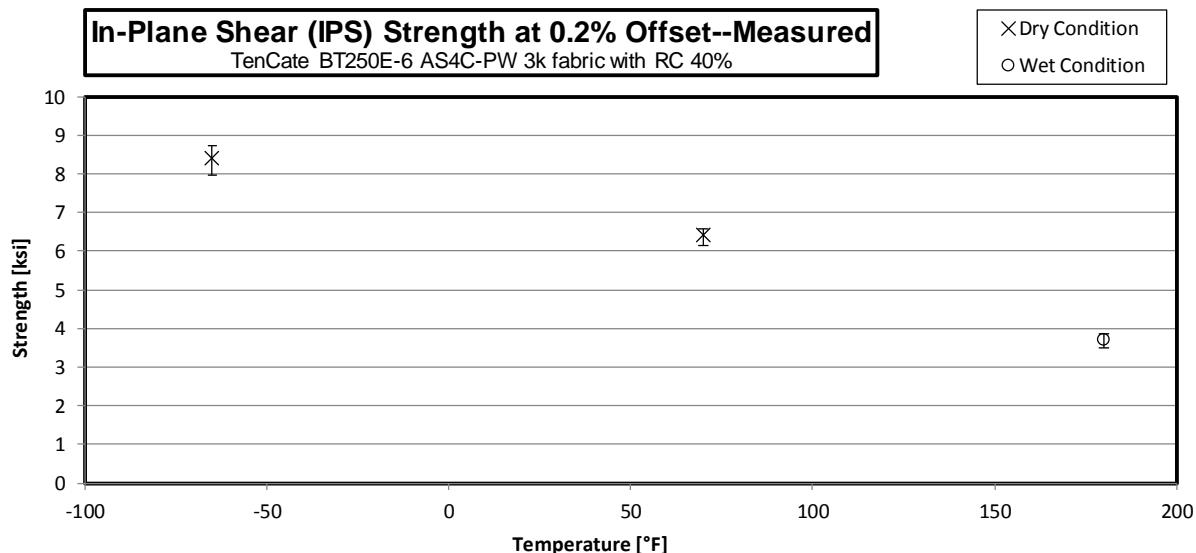
### 3.3 Warp Compression Properties (WC)

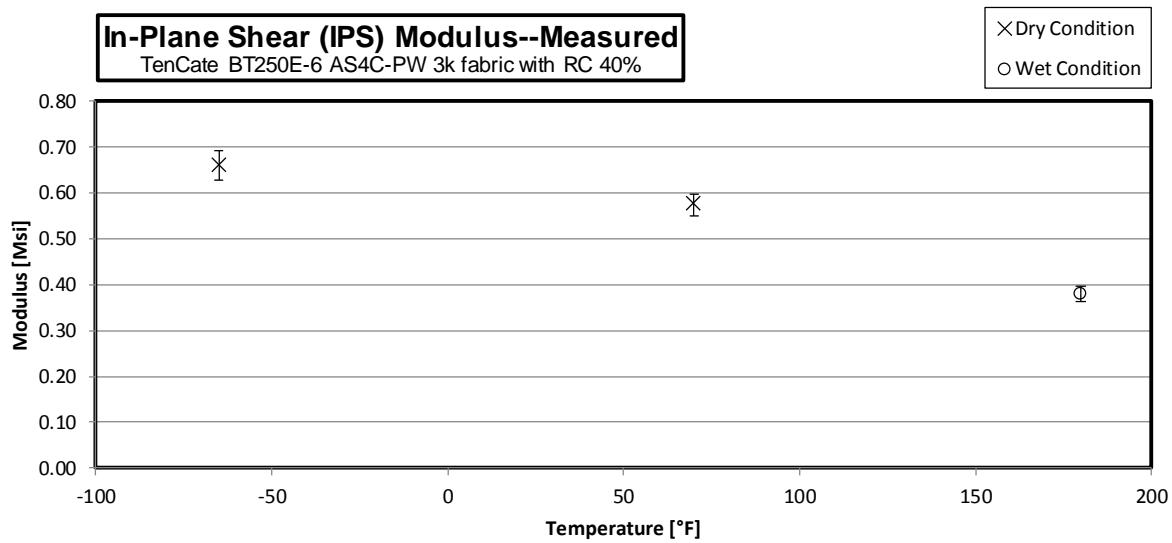


### 3.4 Fill Compression Properties (FC)

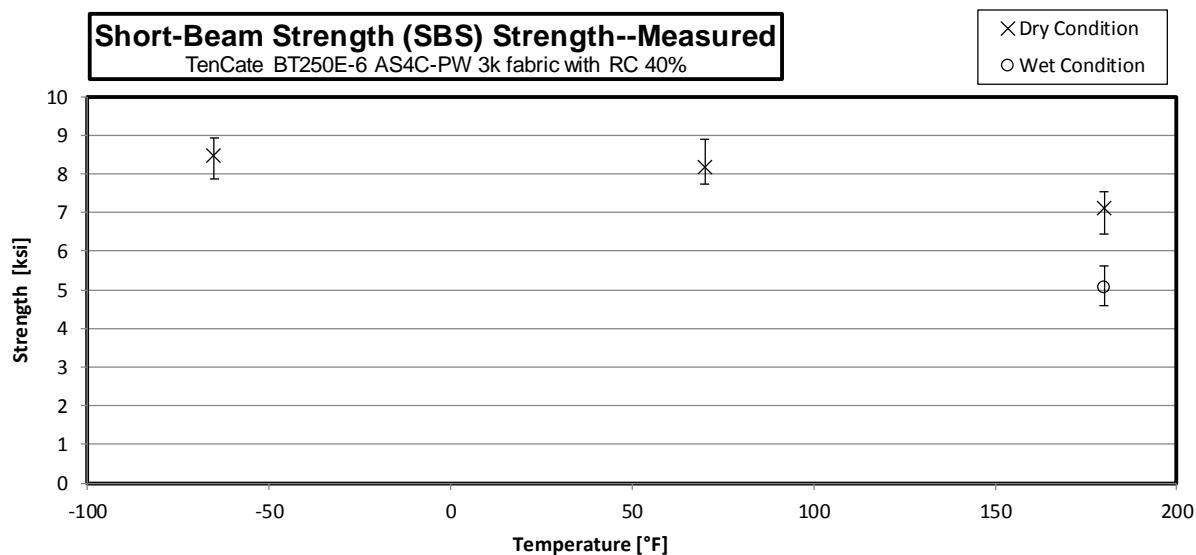


### 3.5 In-Plane Shear Properties (IPS)





### 3.6 Lamina Short-Beam Strength Properties (SBS)



## 4. Individual Test Data

### 4.1 Warp Tension Properties (WT)

**Warp Tension Properties (WT)--CTD**  
**Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
t<sub>ply</sub> [in]  
0.0085

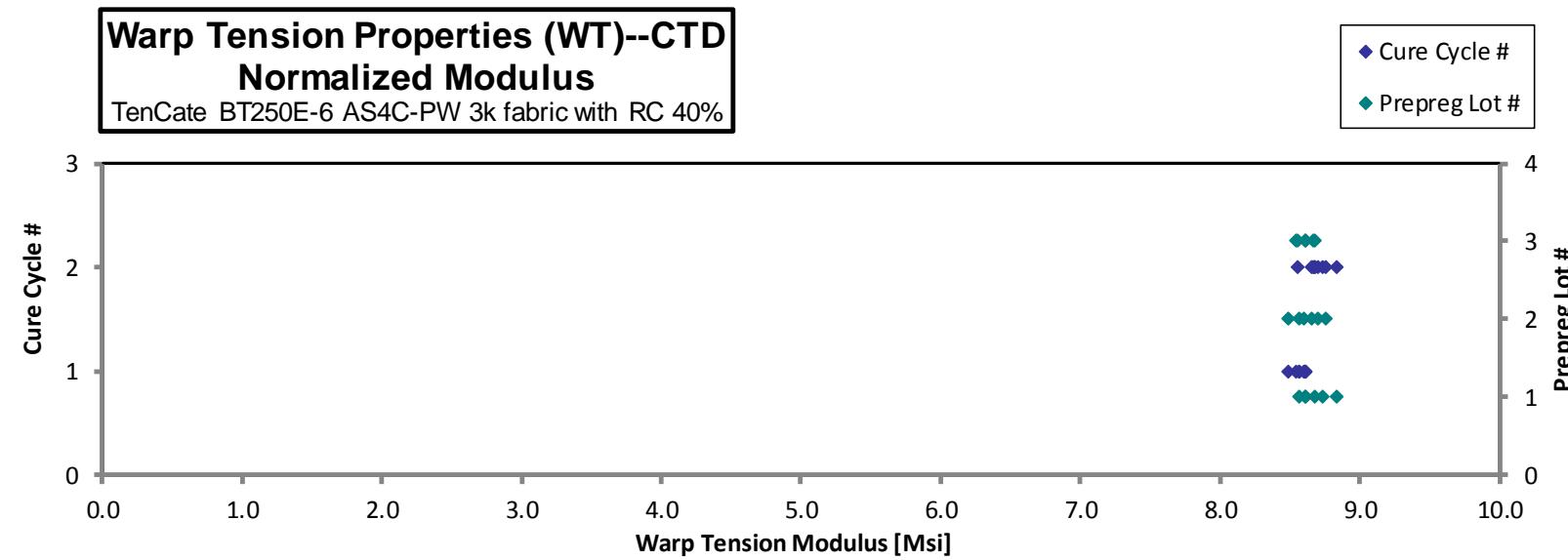
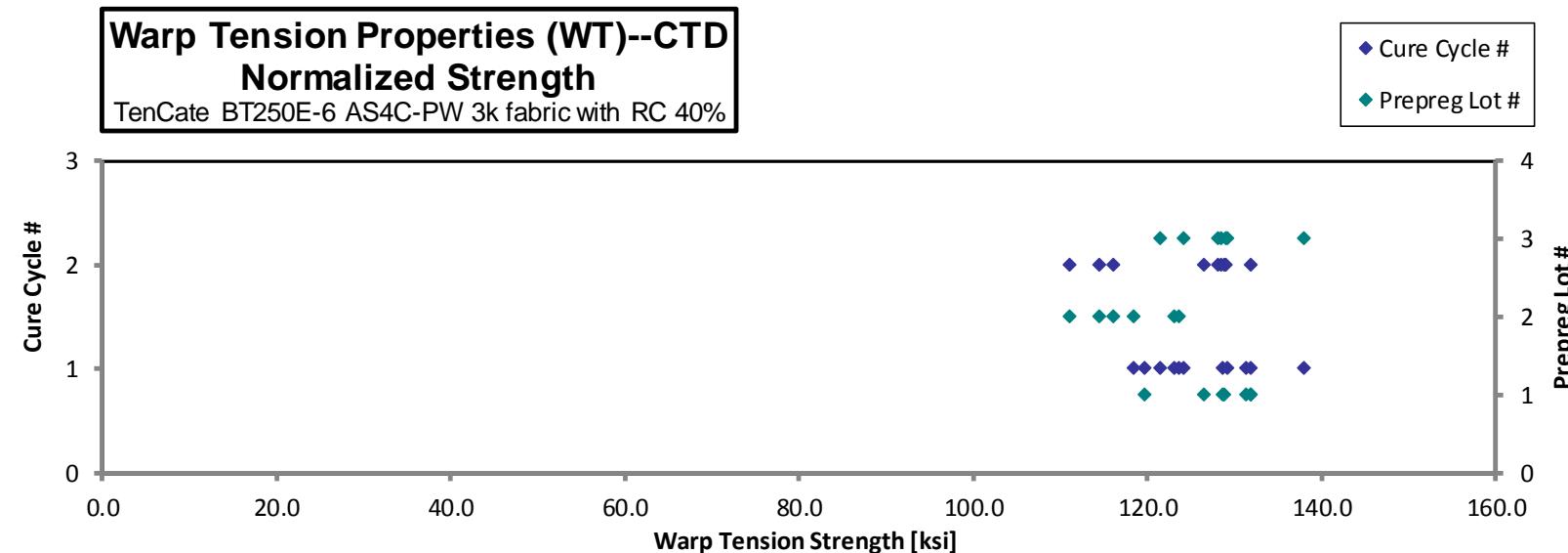
Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Poisson's Ratio	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABJA116B	A	C1	1	1	129.990	8.695	0.060	0.126	15	LGM/ LWB
EABJA117B	A	C1	1	1	134.359	8.760	***	0.125	15	LGM
EABJA118B	A	C1	1	1	122.974	8.843	***	0.124	15	LGM
EABJA119B*	A	C1	1	1	133.790			0.126	15	LGM/ LWB
EABJA215B	A	C2	1	2	126.493	8.684	***	0.127	15	LGM / LWT
EABJA216B	A	C2	1	2	132.440	8.764	0.066	0.127	15	LWT / LGM
EABJA217B	A	C2	1	2	131.243	9.004	0.066	0.125	15	LAB
EABJB116B	B	C1	2	1	120.123	8.364	0.059	0.131	15	LAT
EABJB117B	B	C1	2	1	115.311	8.268	***	0.131	15	LGM
EABJB118B	B	C1	2	1	**	8.402	0.062	0.130	15	LIT
EABJB119B*	B	C1	2	1	124.227			0.127	15	LGM
EABJB215B	B	C2	2	2	110.386	8.714	0.055	0.128	15	LAB
EABJB216B	B	C2	2	2	113.891	8.612	0.051	0.128	15	LAB
EABJB217B	B	C2	2	2	117.330	8.787	0.056	0.126	15	LAT / LWB
EABJC116B	C	C1	3	1	124.808	8.257	0.055	0.132	15	LWB / LAT
EABJC117B	C	C1	3	1	120.132	8.332	0.049	0.132	15	LGM / LWT
EABJC118B	C	C1	3	1	118.502	8.405	0.060	0.131	15	LWT/LWB
EABJC119B*	C	C1	3	1	135.538			0.130	15	LWT/LWB
EABJC215B	C	C2	3	2	127.558	8.604	***	0.128	15	LAB/LWT
EABJC216B	C	C2	3	2	127.926	8.489	0.050	0.128	15	LAB
EABJC217B	C	C2	3	2	128.056	8.663	0.060	0.128	15	LAB

\* Specimen was not gaged and tested for strength only.

\*\* Strength not reported due to unacceptable failure mode.

\*\*\* Poissons Ratio not reported due to non linear data.

Average	124.754	8.592	0.058	Average <sub>norm</sub>	0.0085	125.113	8.631
Standard Dev.	7.272	0.215	0.006	Standard Dev. <sub>norm</sub>		6.741	0.085
Coeff. of Var. [%]	5.829	2.504	9.745	Coeff. of Var. [%] <sub>norm</sub>		5.388	0.981
Min.	110.386	8.257	0.049	Min.	0.0083	110.920	8.485
Max.	135.538	9.004	0.066	Max.	0.0088	137.895	8.829
Number of Spec.	20	18	13	Number of Spec.	21	20	18



**Warp Tension Properties (WT)--RTD  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
 $t_{ply}$  [in]  
0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Poisson's Ratio	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABJA111A	A	C1	1	1	136.161	8.911	0.051	0.124	15	LWB/LWT
EABJA112A	A	C1	1	1	134.214	8.742	**	0.125	15	LWB/LWT
EABJA113A	A	C1	1	1	128.471	8.710	0.053	0.125	15	LWB/LWT
EABJA114A	A	C1	1	1	137.605	8.714	0.057	0.126	15	LWB/LWT
EABJA211A	A	C2	1	2	137.334	8.783	0.052	0.125	15	LAT/LWB
EABJA212A	A	C2	1	2	138.720	8.689	0.058	0.125	15	LAT/LWB
EABJA213A	A	C2	1	2	134.838	8.630	0.059	0.127	15	LAT/LWB
EABJB111A*	B	C1	2	1	112.748	8.576	0.054	0.128	15	LAT
EABJB112A	B	C1	2	1	127.685	8.534	0.062	0.128	15	LAB / LWT
EABJB113A	B	C1	2	1	125.372	8.486	0.059	0.130	15	LAB / LWT
EABJB114A	B	C1	2	1	126.589	8.352	0.051	0.130	15	LAT / LIB
EABJB115A	B	C1	2	1	127.674	8.340	0.054	0.131	15	LAB / LWT
EABJB211A	B	C2	2	2	128.999	8.589	0.055	0.126	15	LAB / LWT
EABJB212A	B	C2	2	2	128.212	8.569	0.055	0.128	15	LWT/LWB
EABJB213A	B	C2	2	2	125.391	8.532	0.055	0.128	15	LGM
EABJC111A	C	C1	3	1	134.765	8.353	0.054	0.130	15	LWB/LWT
EABJC112A	C	C1	3	1	132.440	8.242	0.057	0.131	15	LWB/LWT
EABJC113A	C	C1	3	1	133.707	8.228	0.061	0.131	15	LWB/LWT
EABJC114A	C	C1	3	1	131.694	8.212	0.060	0.132	15	LWB/LWT
EABJC211A	C	C2	3	2	140.517	8.572	0.051	0.128	15	LWT/LWB
EABJC212A	C	C2	3	2	141.422	8.473	0.054	0.128	15	LGM
EABJC213A	C	C2	3	2	137.533	8.564	0.053	0.127	15	LWT/LAB/LGM

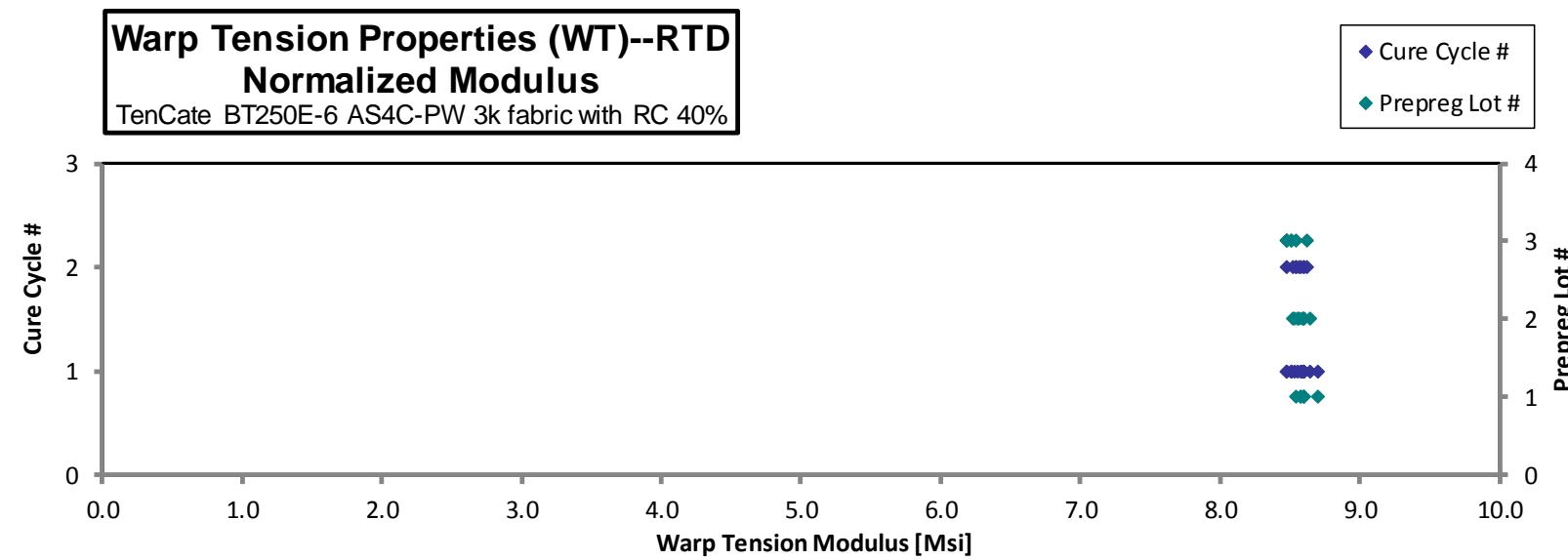
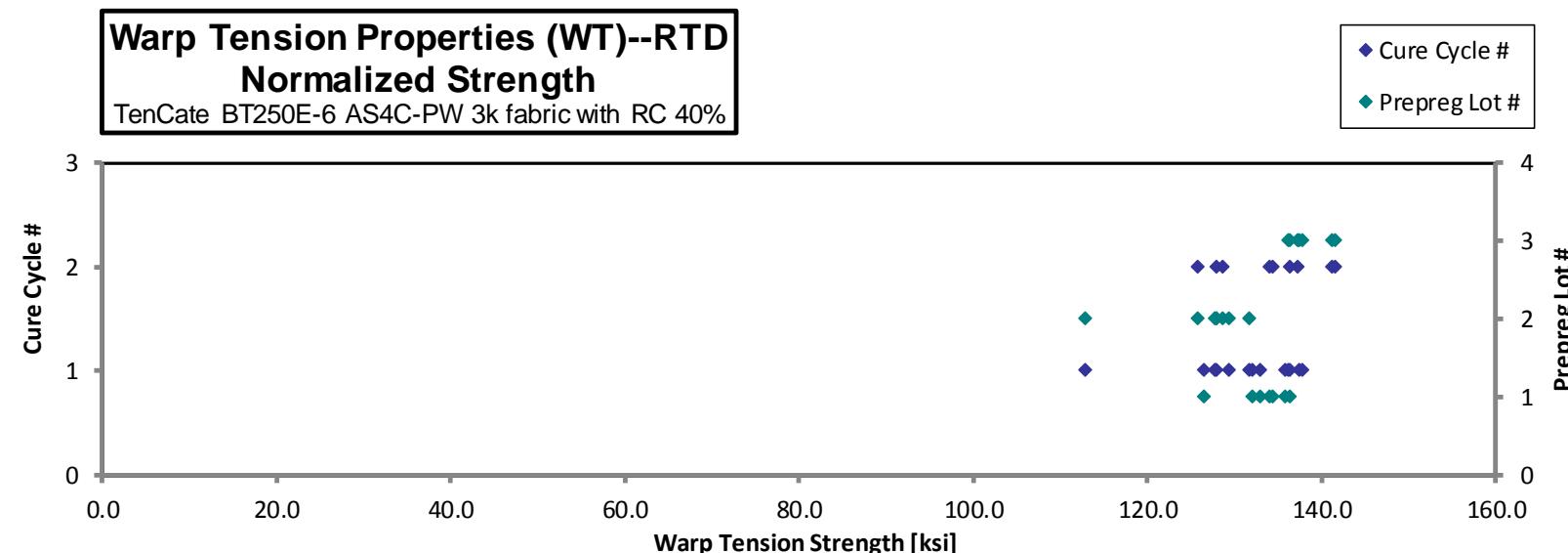
\* Data point investigated and no reason found to omit data.

\*\* Poissons Ratio not reported due to non linear data.

Average	131.913	8.536	0.055
Standard Dev.	6.541	0.188	0.003
Coeff. of Var. [%]	4.959	2.205	6.029
Min.	112.748	8.212	0.051
Max.	141.422	8.911	0.062
Number of Spec.	22	22	21

Avg. $t_{ply}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
0.0083	132.904	8.698
0.0084	131.951	8.594
0.0084	126.439	8.572
0.0084	135.716	8.595
0.0083	134.354	8.592
0.0084	136.345	8.541
0.0084	133.992	8.576
0.0085	112.881	8.586
0.0085	127.936	8.551
0.0087	127.601	8.637
0.0087	129.303	8.531
0.0088	131.596	8.596
0.0084	127.937	8.518
0.0085	128.580	8.594
0.0085	125.768	8.558
0.0087	137.284	8.510
0.0087	136.075	8.468
0.0088	137.675	8.472
0.0088	136.325	8.501
0.0085	141.196	8.614
0.0085	141.459	8.475
0.0085	137.155	8.540

Average	0.0085	132.294	8.560
Standard Dev. <sub>norm</sub>		6.340	0.057
Coeff. of Var. [%] <sub>norm</sub>		4.792	0.663
Min.	0.0083	112.881	8.468
Max.	0.0088	141.459	8.698
Number of Spec.	22	22	22



**Warp Tension Properties (WT)--ETW**  
**Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
 $t_{\text{ply}}$  [in]  
 0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Poisson's Ratio	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABJA11BD	A	C1	1	1	129.634	8.667	0.044	0.127	15	LGM/DGM
EABJA11CD	A	C1	1	1	126.059	8.571	**	0.128	15	LWT/LWB/DGT
EABJA11DD	A	C1	1	1	128.692	8.498	**	0.128	15	LGM/LWT/LWB
EABJA11ED	A	C1	1	1	*	8.544	0.056	0.127	15	LIT
EABJA11FD	A	C1	1	1	131.283	8.580	**	0.126	15	LWT/LWB
EABJA11GD	A	C1	1	1	129.172	8.701	0.056	0.125	15	LWT/LWB
EABJA219D	A	C2	1	2	127.503	8.715	0.041	0.127	15	DGM/LGM/LWT/LWB
EABJA21AD	A	C2	1	2	127.013	8.588	0.049	0.128	15	
EABJA21BD	A	C2	1	2	125.686	8.620	0.083	0.128	15	
EABJA21BD	A	C2	1	2	*	8.620	0.083	0.128	15	
EABJB11BD	B	C1	2	1	119.368	8.444	**	0.130	15	DGM/LGM/AWB
EABJB11CD	B	C1	2	1	113.780	8.500	**	0.131	15	LAB/AWT
EABJB11DD	B	C1	2	1	114.819	8.422	**	0.132	15	LWT/LWB
EABJB11ED	B	C1	2	1	116.409	8.517	0.039	0.130	15	LWT/LWB
EABJB11FD	B	C1	2	1	115.293	8.417	**	0.130	15	LGM
EABJB11GD	B	C1	2	1	119.178	8.393	0.044	0.129	15	LWB
EABJB219D	B	C2	2	2	*	8.602	0.050	0.128	15	LIT
EABJB21AD	B	C2	2	2	122.150	8.479	**	0.128	15	LAT/LWB
EABJB21BD	B	C2	2	2	122.189	8.523	**	0.128	15	LWT/LWB
EABJB21CD	B	C2	2	2	124.467	8.546	0.045	0.128	15	LWT/LWB
EABJC11BD	C	C1	3	1	*	8.209	0.132	0.132	15	LIT
EABJC11CD	C	C1	3	1	118.842	8.158	0.045	0.133	15	LGM/LWT
EABJC11DD	C	C1	3	1	122.198	8.057	0.046	0.134	15	LWT/LWB
EABJC11ED	C	C1	3	1	122.032	8.073	0.054	0.133	15	LGM/DGM
EABJC11FD	C	C1	3	1	120.121	8.197	0.050	0.133	15	LGM/DGM
EABJC219D	C	C2	3	2	123.790	8.539	**	0.128	15	LWT/LWB
EABJC21AD	C	C2	3	2	126.890	8.453	0.056	0.129	15	LWT/LWB
EABJC21BD	C	C2	3	2	125.168	8.348	**	0.129	15	LWB/DGM
EABJC21CD	C	C2	3	2	127.189	8.303	**	0.129	15	LGM/LWB

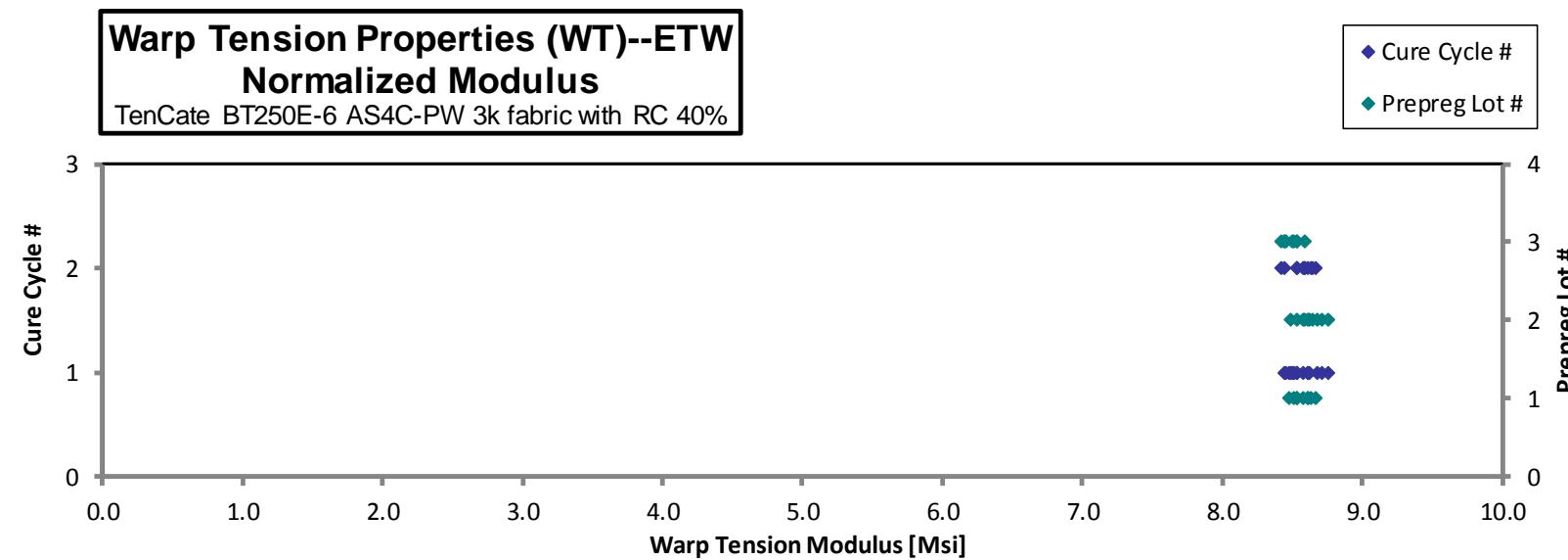
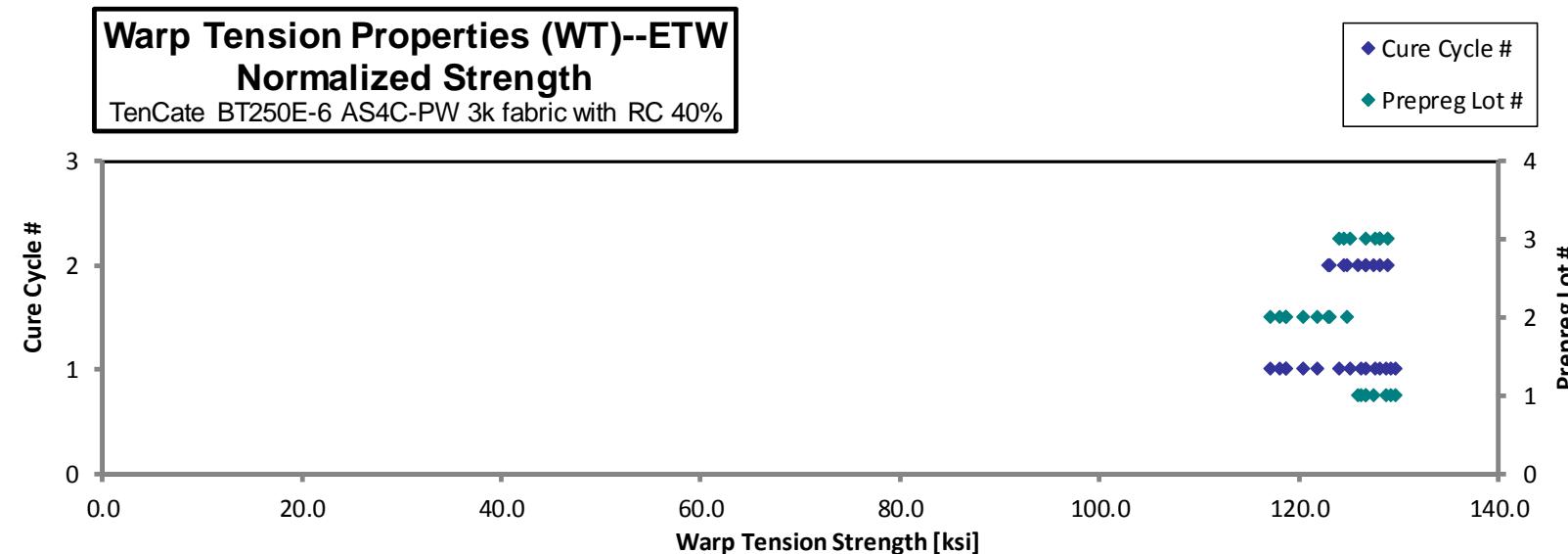
Avg. $t_{\text{ply}}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
0.0084	128.719	8.606
0.0085	126.141	8.577
0.0085	129.163	8.529
0.0085	129.636	8.506
0.0083	126.572	8.526
0.0084	126.653	8.657
0.0085	127.345	8.610
0.0085	125.768	8.626
0.0087	121.802	8.616
0.0087	117.096	8.747
0.0088	118.676	8.704
0.0087	118.585	8.677
0.0087	117.946	8.610
0.0086	120.408	8.479
0.0085	122.837	8.527
0.0086	123.036	8.582
0.0085	124.776	8.567
0.0088	123.922	8.507
0.0089	128.061	8.443
0.0089	127.504	8.434
0.0088	125.005	8.530
0.0085	124.388	8.580
0.0086	128.018	8.528
0.0086	126.559	8.441
0.0086	128.834	8.411

\* Strength not reported due to unacceptable failure mode.

\*\* Poissons Ratio not reported due to non linear data.

Average	123.157	8.452	0.051
Standard Dev.	4.944	0.179	0.011
Coeff. of Var. [%]	4.015	2.118	20.962
Min.	113.780	8.057	0.039
Max.	131.283	8.715	0.083
Number of Spec.	25	28	15

Average <sub>norm</sub>	0.0086	124.698	8.558
Standard Dev. <sub>norm</sub>	3.762	0.085	
Coeff. of Var. [%] <sub>norm</sub>	3.017	0.994	
Min.	0.0083	117.096	8.411
Max.	0.0089	129.636	8.747
Number of Spec.	28	25	28



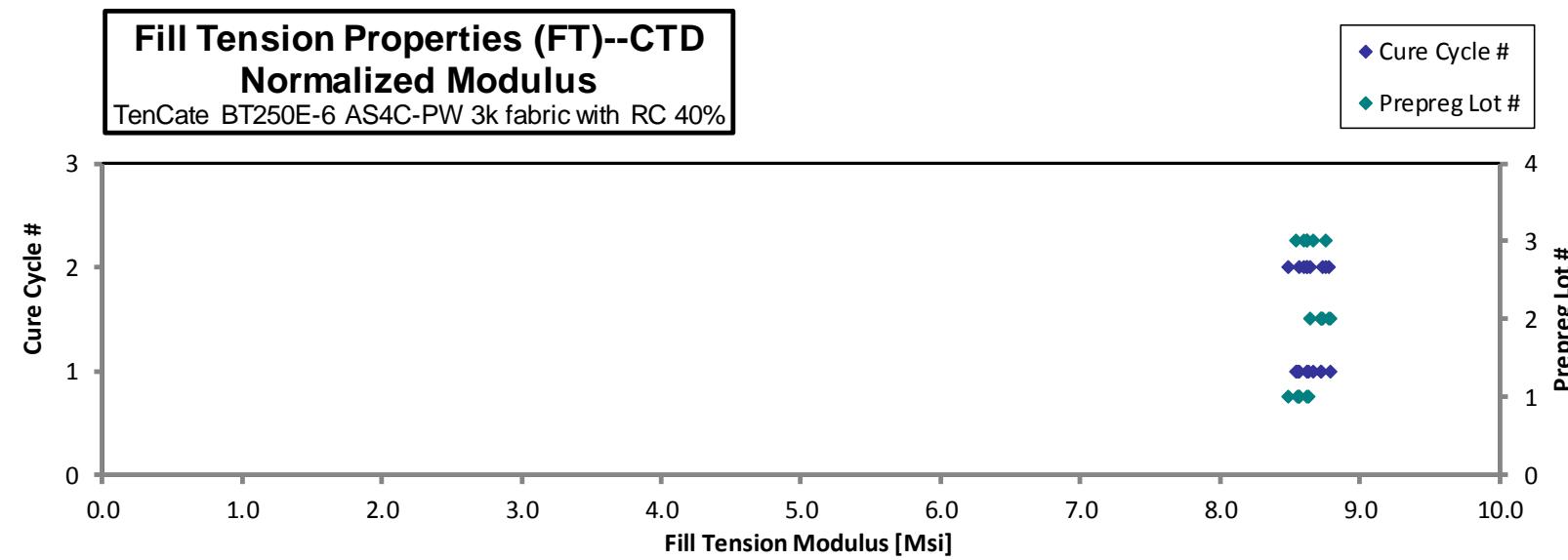
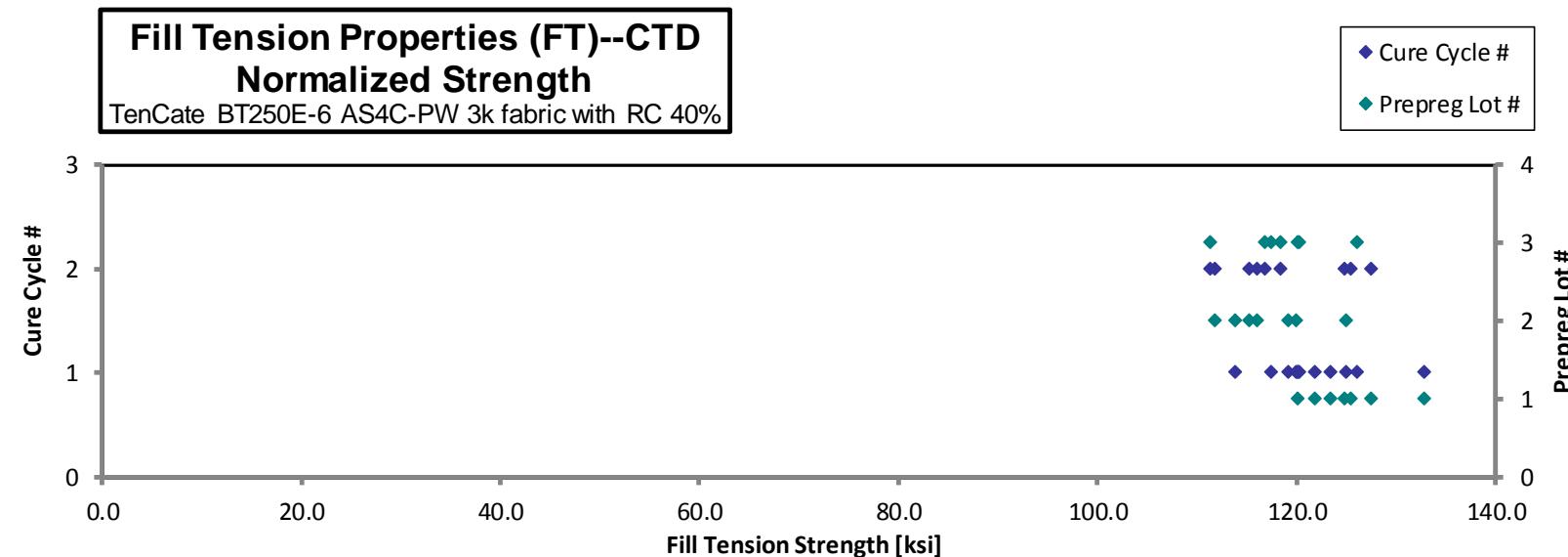
## 4.2 Fill Tension Properties (FT)

Fill Tension Properties (FT)--CTD Strength & Modulus									normalizing $t_{\text{ply}}$ [in]
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%									0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABUA116B	A	C1	1	1	133.218	8.664	0.127	15	LGM
EABUA117B	A	C1	1	1	122.223	8.596	0.127	15	LWT
EABUA118B	A	C1	1	1	124.625	8.636	0.126	15	LWT
EABUA119B*	A	C1	1	1	122.935		0.124	15	LWB
EABUA215B	A	C2	1	2	123.536	8.404	0.129	15	LWB
EABUA216B	A	C2	1	2	126.473	8.498	0.128	15	LWB
EABUA217B	A	C2	1	2	125.745	8.639	0.127	15	LAB
EABUB116B	B	C1	2	1	117.783	8.571	0.130	15	LAT/LWB
EABUB117B	B	C1	2	1	112.145	8.597	0.129	15	LGM
EABUB118B	B	C1	2	1	117.070	8.637	0.130	15	LGM
EABUB119B*	B	C1	2	1	124.853		0.127	15	LGM
EABUB215B	B	C2	2	2	109.120	8.566	0.131	15	LAB
EABUB216B	B	C2	2	2	113.526	8.536	0.130	15	LAB
EABUB217B	B	C2	2	2	115.424	8.659	0.127	15	LAB
EABUC116B	C	C1	3	1	121.165	8.681	0.127	15	LAB
EABUC117B	C	C1	3	1	121.228	8.746	0.126	15	LGM
EABUC118B	C	C1	3	1	118.259	8.609	0.127	15	LAT / LWB
EABUC119B*	C	C1	3	1	130.128		0.123	15	LGM / LWB
EABUC216B	C	C2	3	2	116.288	8.719	0.128	15	LGM
EABUC217B	C	C2	3	2	120.237	8.726	0.126	15	LWT / LWB
EABUC218B	C	C2	3	2	110.832	8.587	0.128	15	LAB

\* Specimen was not gaged and tested for strength only.

Average	120.324	8.615	Average <sub>norm</sub>	0.0085	120.271	8.641
Standard Dev.	6.250	0.084	Standard Dev. <sub>norm</sub>	5.398	0.088	
Coeff. of Var. [%]	5.194	0.978	Coeff. of Var. [%] <sub>norm</sub>	4.488	1.021	
Min.	109.120	8.404	Min.	0.0082	111.238	8.482
Max.	133.218	8.746	Max.	0.0087	132.730	8.783
Number of Spec.	21	18	Number of Spec.	21	21	18



October 24, 2017

CAM-RP-2015-039 N/C

**Fill Tension Properties (FT)-RTD  
Strength & Modulus**

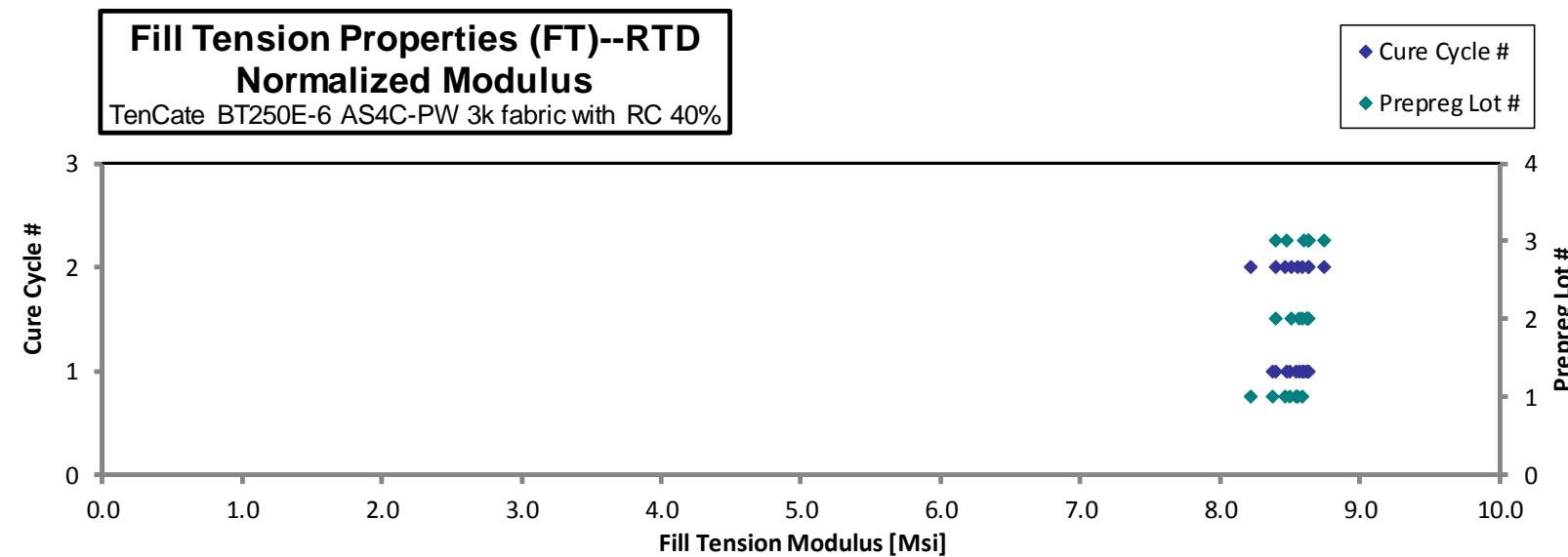
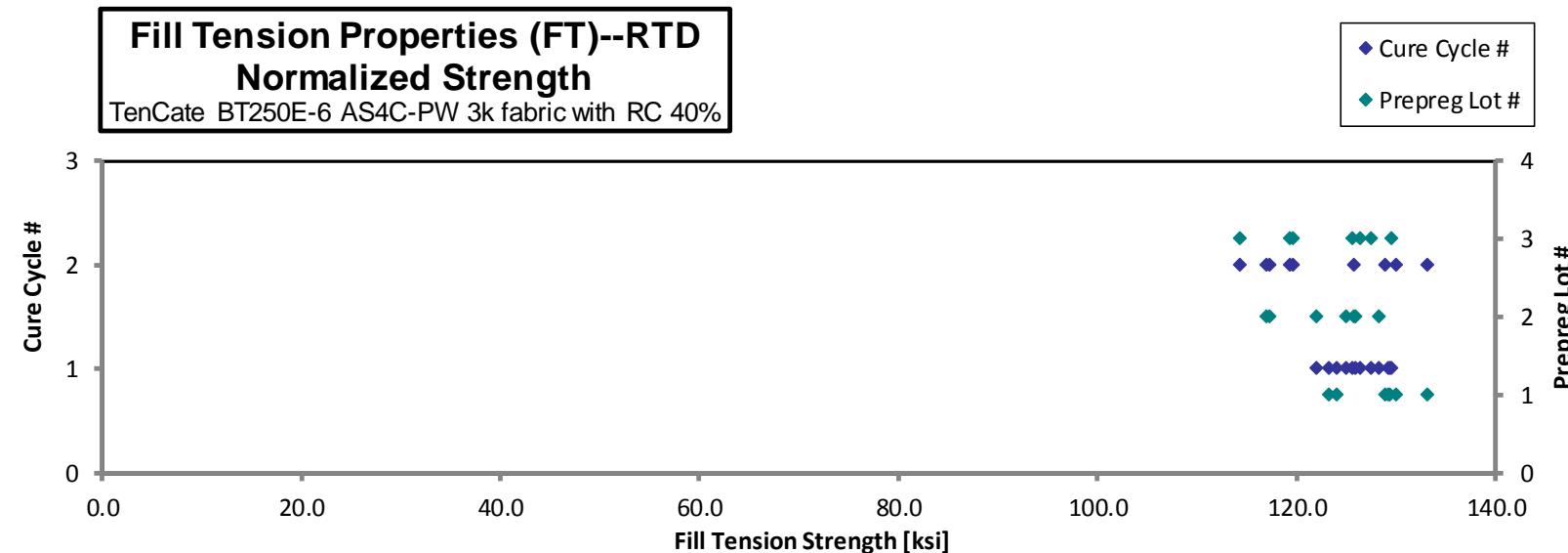
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
 $t_{\text{ply}}$  [in]  
0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABUA111A	A	C1	1	1	132.998	8.737	0.124	15	LGM
EABUA112A	A	C1	1	1	131.022	8.709	0.126	15	LWB/LWT
EABUA113A	A	C1	1	1	124.515	8.622	0.126	15	LWT/LGM
EABUA114A	A	C1	1	1	124.611	8.414	0.127	15	LGM
EABUA211A	A	C2	1	2	129.997	8.465	0.127	15	LGM
EABUA212A	A	C2	1	2	132.682	8.195	0.128	15	LWB
EABUA213A	A	C2	1	2	128.390	8.518	0.128	15	LGM
EABUB111A	B	C1	2	1	126.855	8.460	0.126	15	LAT / LWB
EABUB112A	B	C1	2	1	121.865	8.549	0.128	15	LWB
EABUB113A	B	C1	2	1	123.808	8.535	0.129	15	LGM
EABUB114A	B	C1	2	1	126.051	8.474	0.130	15	LAB / LAT
EABUB211A	B	C2	2	2	117.123	8.510	0.127	15	LGM
EABUB212A	B	C2	2	2	124.933	8.576	0.128	15	LGM
EABUB213A	B	C2	2	2	116.206	8.505	0.129	15	LWB
EABUC111A	C	C1	3	1	131.454	9.003	0.122	15	LWB / LWT
EABUC112A	C	C1	3	1	128.968	8.727	0.126	15	LGM
EABUC113A	C	C1	3	1	129.874	8.500	0.127	15	LWB / LWT
EABUC114A	C	C1	3	1	126.387	8.632	0.127	15	LGM
EABUC211A	C	C2	3	2	120.837	8.829	0.126	15	LGM
EABUC212A	C	C2	3	2	114.351	8.396	0.127	15	LWB
EABUC213A	C	C2	3	2	118.858	8.592	0.128	15	LGM

Avg. $t_{\text{ply}}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
0.0083	129.243	8.490
0.0084	129.086	8.580
0.0084	123.229	8.533
0.0085	123.911	8.366
0.0085	129.912	8.460
0.0085	133.012	8.215
0.0085	128.893	8.552
0.0084	125.827	8.392
0.0085	121.992	8.558
0.0086	124.957	8.614
0.0086	128.127	8.613
0.0085	116.970	8.499
0.0085	125.635	8.624
0.0086	117.224	8.579
0.0081	125.475	8.593
0.0084	127.417	8.622
0.0085	129.381	8.468
0.0085	126.304	8.627
0.0084	119.589	8.737
0.0085	114.261	8.389
0.0085	119.324	8.626

Average	125.323	8.569	Average <sub>norm</sub>	0.0085	124.751	8.530
Standard Dev.	5.480	0.170	Standard Dev. <sub>norm</sub>	4.949	0.118	
Coeff. of Var. [%]	4.373	1.983	Coeff. of Var. [%] <sub>norm</sub>	3.967	1.385	
Min.	114.351	8.195	Min.	0.0081	114.261	8.215
Max.	132.998	9.003	Max.	0.0086	133.012	8.737
Number of Spec.	21	21	Number of Spec.	21	21	21



**Fill Tension Properties (FT)--ETW  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

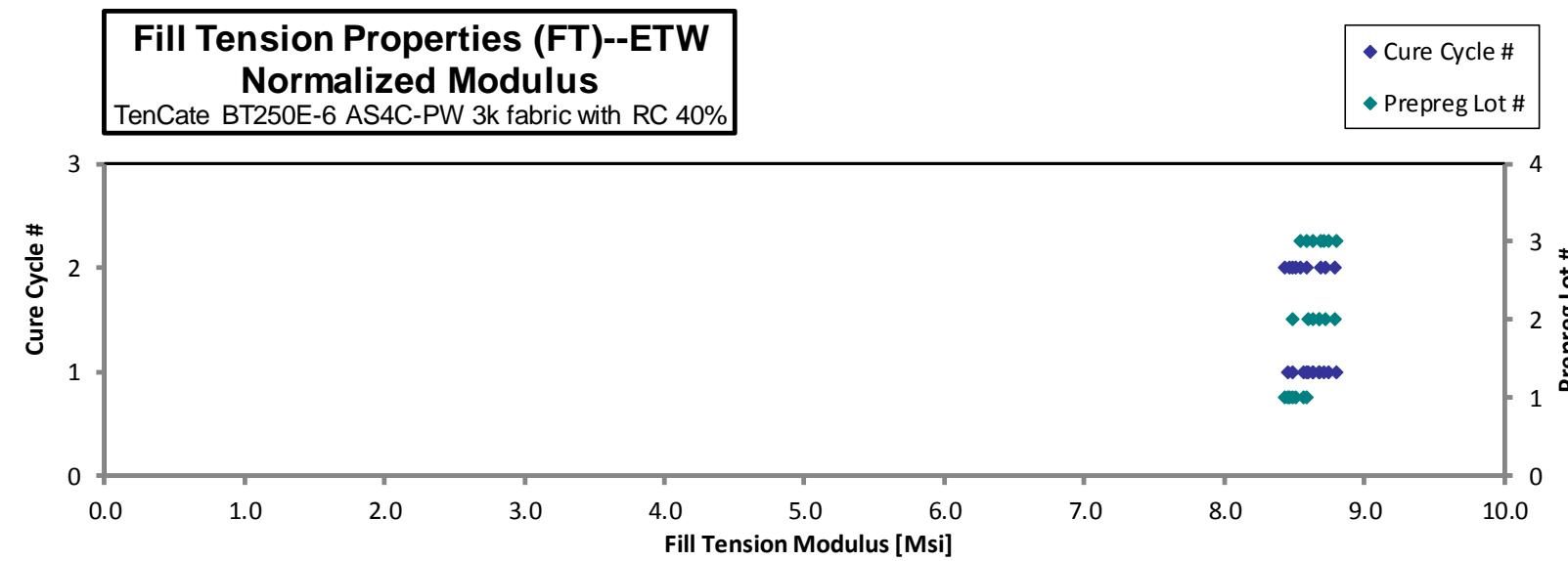
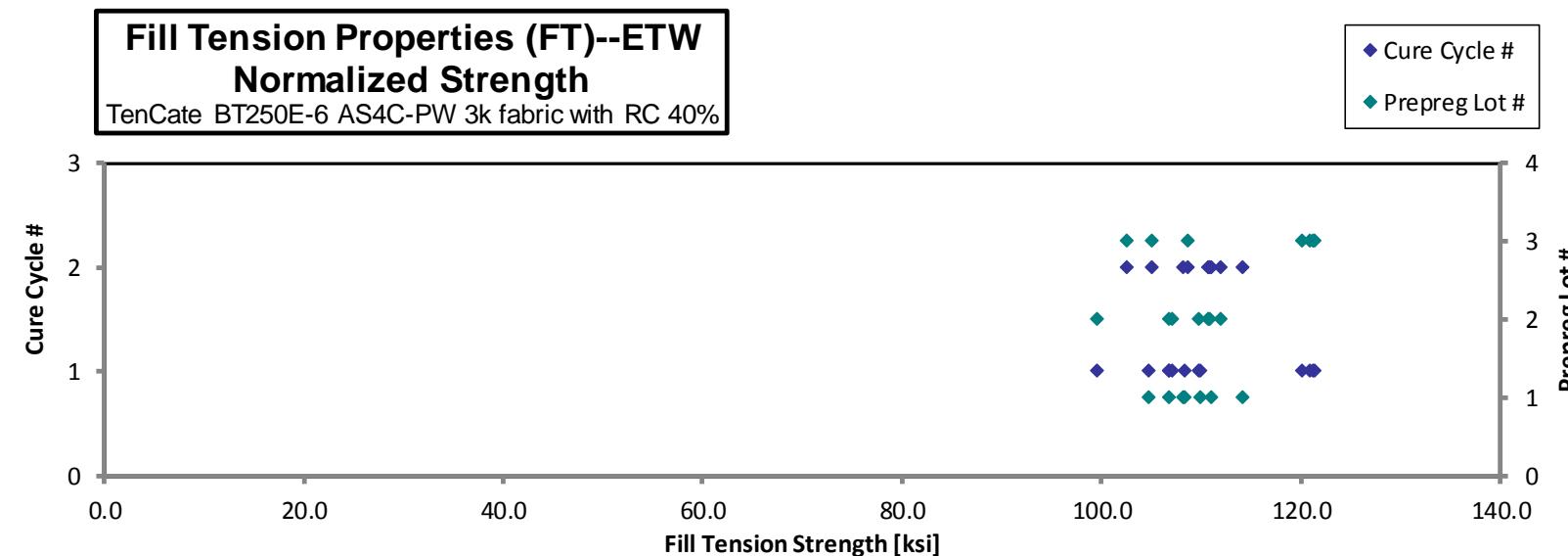
 normalizing  
 $t_{ply}$  [in]  
 0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABUA11BD	A	C1	1	1	105.781	8.653	0.126	15	LGM
EABUA11CD	A	C1	1	1	109.023	8.635	0.127	15	LGM
EABUA11DD	A	C1	1	1	106.818	8.469	0.127	15	LWB
EABUA11ED	A	C1	1	1	110.082	8.501	0.127	15	LGM/LWB
EABUA219D	A	C2	1	2	107.596	8.459	0.128	15	LGM
EABUA21AD	A	C2	1	2	110.091	8.399	0.129	15	LWT/LWB/DGM
EABUA21BD	A	C2	1	2	113.251	8.363	0.128	15	LGM
EABUB11BD	B	C1	2	1	109.038	8.552	0.128	15	LGM
EABUB11CD	B	C1	2	1	106.289	8.614	0.128	15	LGM
EABUB11DD	B	C1	2	1	105.986	8.564	0.128	15	LWB/DGM
EABUB11ED	B	C1	2	1	98.594	8.592	0.129	15	LGM
EABUB219D	B	C2	2	2	110.058	8.428	0.128	15	LGM
EABUB21AD	B	C2	2	2	109.443	8.694	0.129	15	LWT
EABUB21BD	B	C2	2	2	109.874	8.551	0.130	15	LGM
EABUC11BD	C	C1	3	1	121.105	8.871	0.126	15	LGM/DGM
EABUC11CD	C	C1	3	1	122.067	8.713	0.126	15	LGM/DGM
EABUC11DD	C	C1	3	1	122.561	8.832	0.126	15	LGM/DGM
EABUC11ED	C	C1	3	1	122.182	8.790	0.126	15	LGM/DGM
EABUC219D	C	C2	3	2	101.223	8.569	0.129	15	LGM
EABUC21AD	C	C2	3	2	103.909	8.456	0.129	15	LWB
EABUC21BD	C	C2	3	2	108.479	8.571	0.128	15	LGM

Avg. $t_{ply}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
0.0084	104.689	8.564
0.0084	108.324	8.580
0.0085	106.622	8.453
0.0085	109.909	8.488
0.0085	108.172	8.505
0.0086	110.955	8.465
0.0086	114.035	8.421
0.0085	109.637	8.599
0.0086	106.984	8.671
0.0086	106.707	8.622
0.0086	99.509	8.671
0.0086	110.806	8.485
0.0086	110.573	8.784
0.0087	111.943	8.712
0.0084	119.997	8.790
0.0084	120.823	8.624
0.0084	121.328	8.743
0.0084	121.080	8.711
0.0086	102.533	8.680
0.0086	104.955	8.541
0.0085	108.564	8.578

Average	110.164	8.585
Standard Dev.	6.699	0.139
Coeff. of Var. [%]	6.081	1.618
Min.	98.594	8.363
Max.	122.561	8.871
Number of Spec.	21	21

Average <sub>norm</sub>	0.0085	110.388	8.604
Standard Dev. <sub>norm</sub>		6.108	0.111
Coeff. of Var. [%] <sub>norm</sub>		5.533	1.286
Min.	0.0084	99.509	8.421
Max.	0.0087	121.328	8.790
Number of Spec.	21	21	21



### 4.3 Warp Compression Properties (WC)

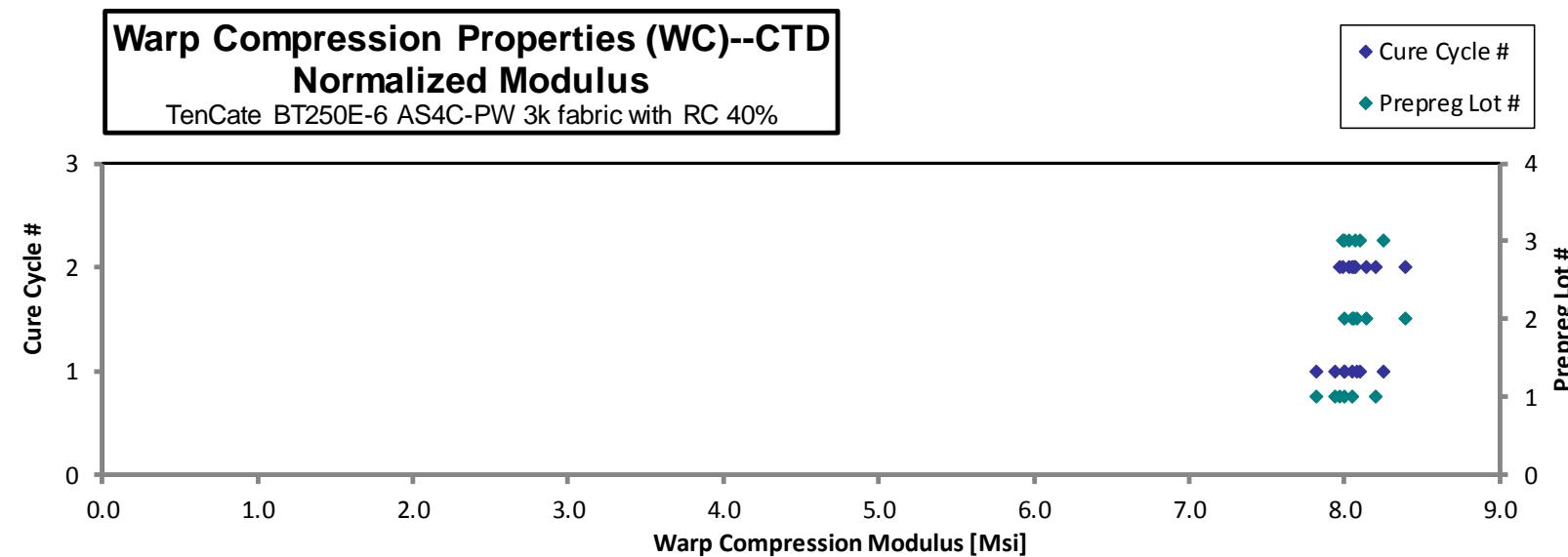
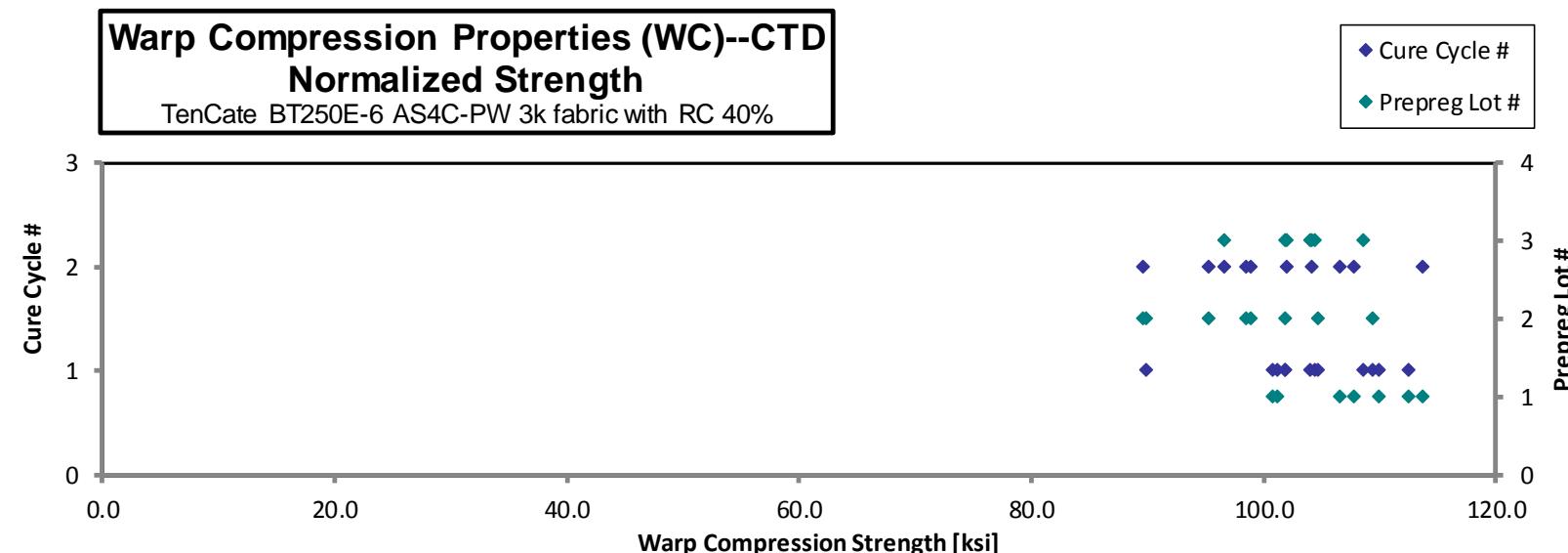
Warp Compression Properties (WC)--CTD Strength & Modulus									normalizing $t_{ply}$ [in] 0.0085			
Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode	Avg. $t_{ply}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
EABLA116B	A	C1	1	1	108.167	7.699	0.129	15	BGM	0.0086	109.853	7.819
EABLA117B	A	C1	1	1	111.242	7.917	0.129	15	HAT / BGM	0.0086	112.398	8.000
EABLA118B	A	C1	1	1	100.321	7.874	0.128	15	BGM	0.0086	101.098	7.935
EABLA119B*	A	C1	1	1	99.390		0.129	15	BGM	0.0086	100.725	
EABLA215B	A	C2	1	2	106.110	7.850	0.129	15	BAB	0.0086	107.670	7.966
EABLA216B	A	C2	1	2	112.219	8.093	0.129	15	BGM	0.0086	113.672	8.198
EABLA217B	A	C2	1	2	105.048	7.937	0.129	15	BGM	0.0086	106.459	8.043
EABLBB16B	B	C1	2	1	107.266	7.929	0.130	15	BGM	0.0087	109.296	8.079
EABLBB117B	B	C1	2	1	102.502	7.875	0.130	15	BGM	0.0087	104.693	8.043
EABLBB118B	B	C1	2	1	99.893	7.842	0.130	15	BGM	0.0087	101.813	7.992
EABLBB119B*	B	C1	2	1	87.832		0.130	15	BGM	0.0087	89.838	
EABLBB215B	B	C2	2	2	87.625	8.214	0.130	15	BGM	0.0087	89.515	8.391
EABLBB216B	B	C2	2	2	95.906	7.930	0.131	15	BGM	0.0087	98.398	8.136
EABLBB217B	B	C2	2	2	96.140	7.836	0.131	15	BGM/HIT	0.0087	98.836	8.055
EABLBB218B	B	C2	2	2	92.343		0.131	15	BGM	0.0088	95.231	
EABLC116B	C	C1	3	1	103.368	8.021	0.129	15	BGM	0.0086	104.311	8.094
EABLC117B	C	C1	3	1	108.172	7.972	0.128	15	BGM	0.0085	108.532	7.998
EABLC118B	C	C1	3	1	100.526	8.141	0.129	15	BGM	0.0086	101.866	8.249
EABLC119B*	C	C1	3	1	103.123		0.129	15	BGM	0.0086	103.932	
EABLC215B	C	C2	3	2	103.131	7.955	0.129	15	BGM	0.0086	104.102	8.030
EABLC216B	C	C2	3	2	100.588	7.878	0.129	15	BGM	0.0086	101.959	7.986
EABLC217B	C	C2	3	2	**	7.904	0.130	15	HIT/BGM	0.0087		8.065
EABLC218B*	C	C2	3	2	94.331		0.130	15	BGM	0.0087	96.514	

\* Specimen was not gaged and tested for strength only.

\*\* Strength was omitted because unacceptable failure mode is more prominent.

Average	101.147	7.937
Standard Dev.	6.753	0.121
Coeff. of Var. [%]	6.677	1.519
Min.	87.625	7.699
Max.	112.219	8.214
Number of Spec.	22	18

Average <sub>norm</sub>	0.0086	102.760	8.060
Standard Dev. <sub>norm</sub>	6.426	0.127	
Coeff. of Var. [%] <sub>norm</sub>	6.254	1.572	
Min.	0.0085	89.515	7.819
Max.	0.0088	113.672	8.391
Number of Spec.	23	22	18



**Warp Compression Properties (WC)–RTD  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

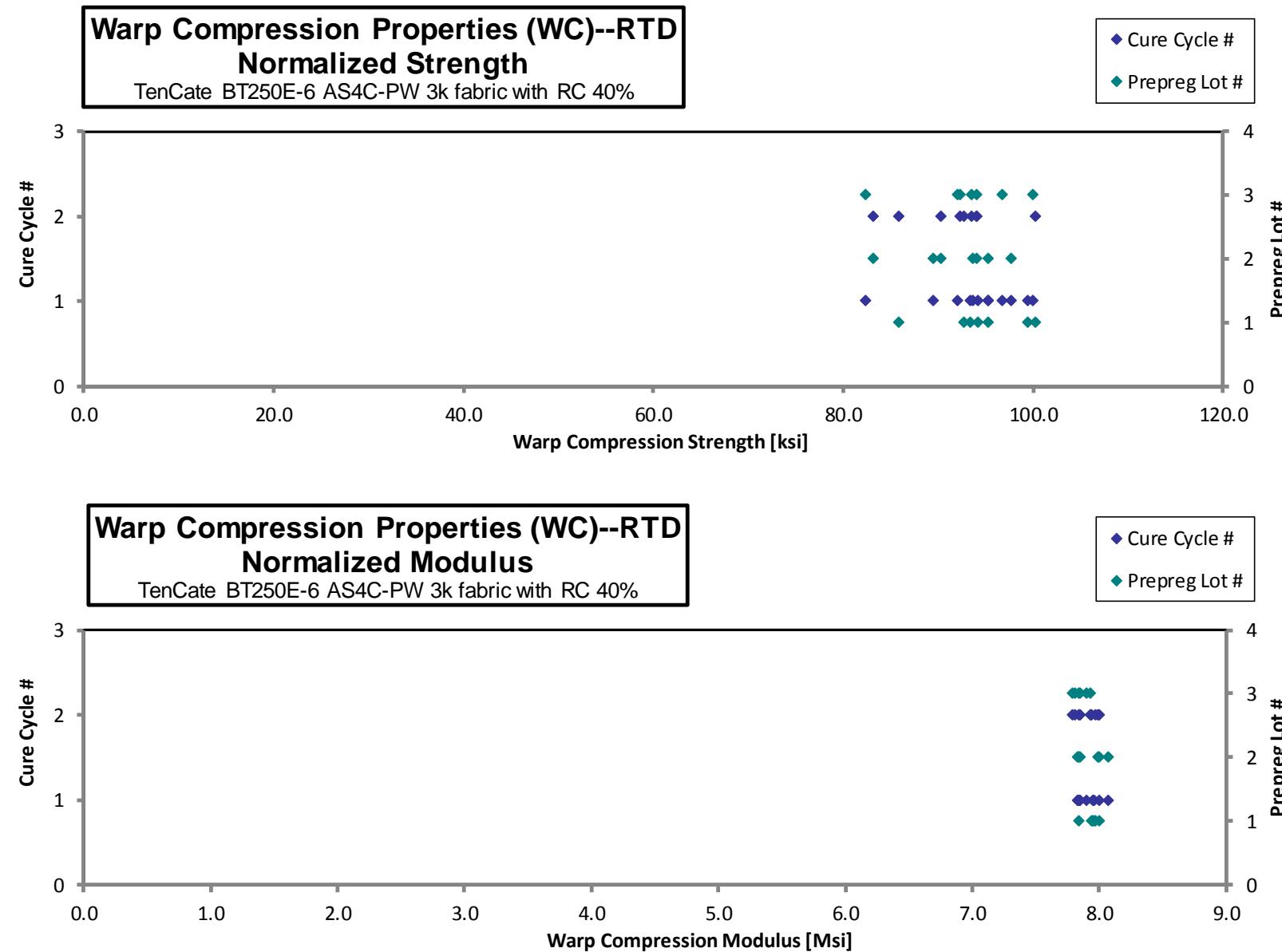
normalizing  
 $t_{\text{ply}}$  [in]  
0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABLA111A	A	C1	1	1	101.045	8.087	0.125	15	BGM/HIB
EABLA112A	A	C1	1	1	94.066	8.058	0.127	15	BGM/HIT
EABLA113A	A	C1	1	1	94.613	7.890	0.128	15	BGM
EABLA114A*	A	C1	1	1	93.390		0.129	15	BGM/HIB
EABLA211A	A	C2	1	2	92.378	7.936	0.128	15	HGM/BGM
EABLA212A	A	C2	1	2	84.839	7.847	0.129	15	BGM/HAB
EABLA213A	A	C2	1	2	99.147	7.757	0.129	15	HGM/BGM
EABLBB111A	B	C1	2	1	98.078	8.106	0.127	15	HIB/BGM
EABLBB112A	B	C1	2	1	89.464	7.821	0.128	15	HAB/BGM
EABLBB113A	B	C1	2	1	92.936	7.781	0.128	15	HGM/HIT
EABLBB114A*	B	C1	2	1	94.147		0.129	15	HIT/BGM
EABLBB211A	B	C2	2	2	94.939	8.080	0.126	15	BGM
EABLBB212A	B	C2	2	2	83.294	7.864	0.127	15	BGM
EABLBB213A	B	C2	2	2	89.515	7.917	0.129	15	BGM/HIB
EABLC111A	C	C1	3	1	91.313	7.771	0.129	15	BGM/HAB
EABLC112A	C	C1	3	1	97.948	7.690	0.130	15	HIT/HGM
EABLC113A	C	C1	3	1	81.462	7.812	0.129	15	HGM
EABLC114A*	C	C1	3	1	95.506		0.129	15	BGM
EABLC211A	C	C2	3	2	94.661	7.882	0.126	15	BGM
EABLC212A	C	C2	3	2	94.247	7.949	0.127	15	BGM
EABLC213A	C	C2	3	2	91.821	7.765	0.128	15	BGM

\* Specimen was not gaged and tested for strength only.

Average	92.800	7.890
Standard Dev.	4.974	0.126
Coeff. of Var. [%]	5.359	1.593
Min.	81.462	7.690
Max.	101.045	8.106
Number of Spec.	21	18

Average <sub>norm</sub>	0.0085	93.092	7.905
Standard Dev. <sub>norm</sub>	4.863	0.081	
Coeff. of Var. [%] <sub>norm</sub>	5.224	1.019	
Min.	0.0084	82.364	7.783
Max.	0.0087	100.158	8.067
Number of Spec.	21	21	18



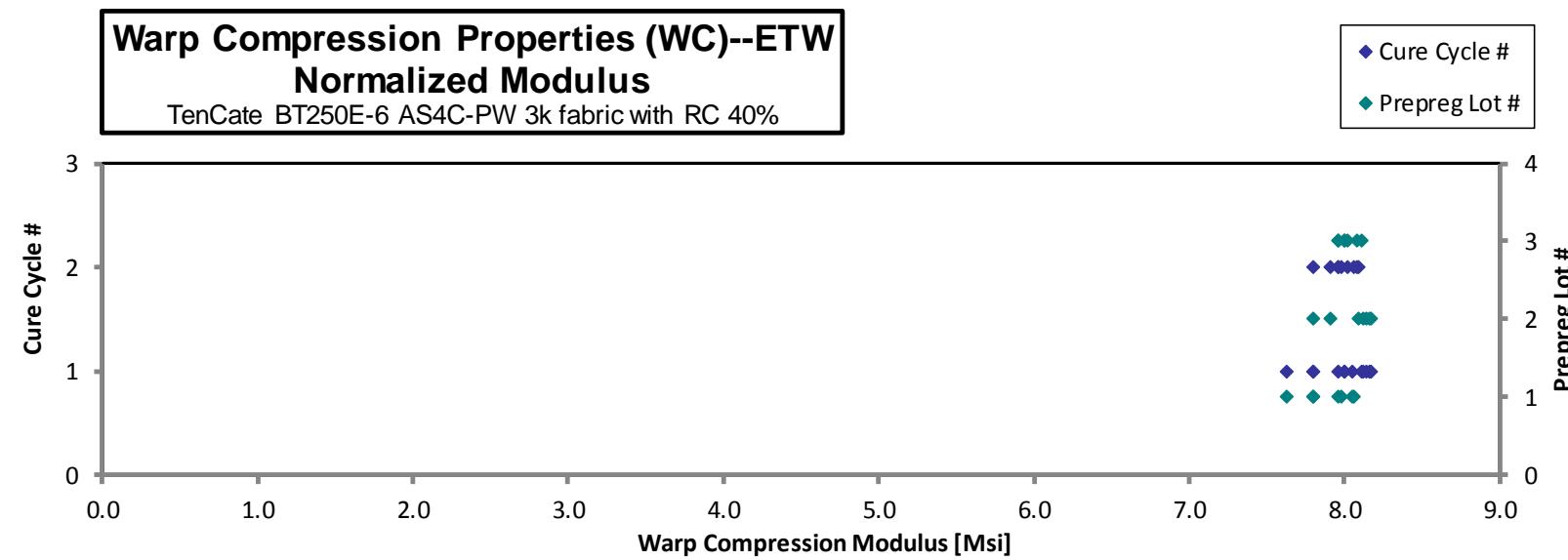
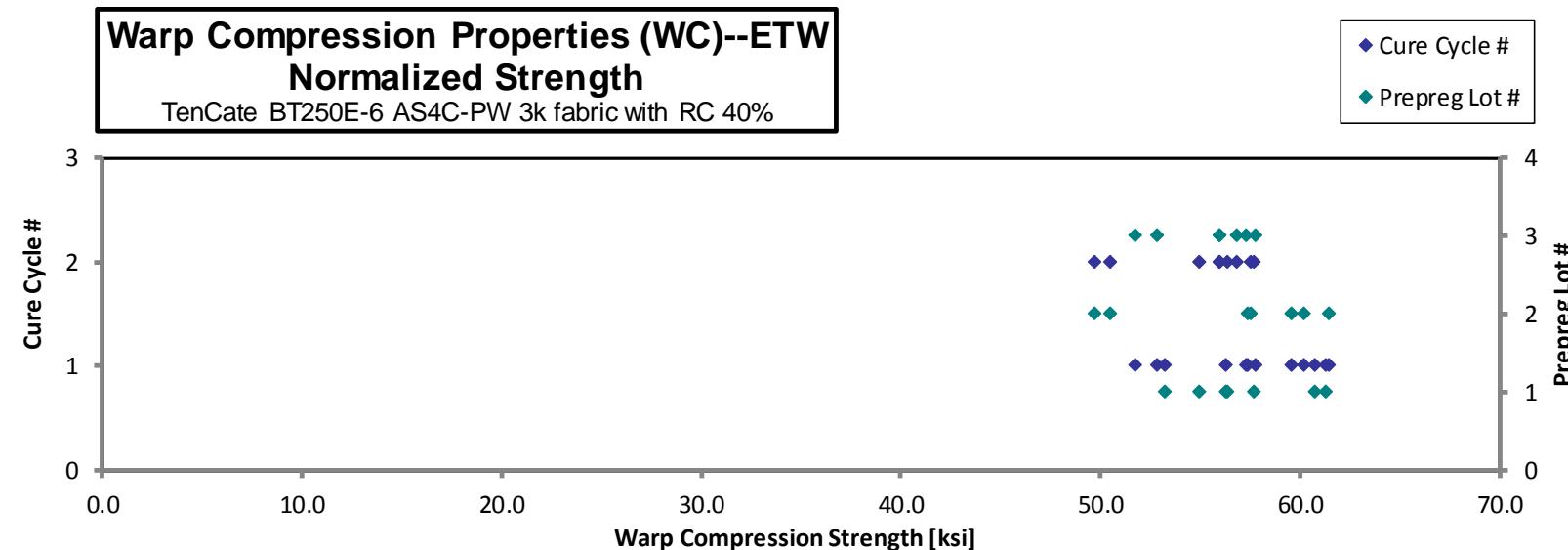
**Warp Compression Properties (WC)--ETW  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
 $t_{\text{ply}}$  [in]  
0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode	Avg. $t_{\text{ply}}$ [in]	Strength <sub>norm</sub> [ksi]	Modulus <sub>norm</sub> [Msi]
EABLA11BD	A	C1	1	1		8.179	0.125	15	HGM	0.0084	8.049	
EABLA11CD	A	C1	1	1		7.835	0.127	15	BGM	0.0085	7.790	
EABLA11DD	A	C1	1	1		7.787	0.128	15	BGM	0.0085	7.800	
EABLA11ED	A	C1	1	1		7.550	0.129	15	HGM	0.0086	7.622	
EABLA11FD	A	C1	1	1	60.544		0.129	15	BGM	0.0086	61.238	
EABLA11GD	A	C1	1	1	60.161		0.129	15	BGM	0.0086	60.692	
EABLA11HD	A	C1	1	1	55.968		0.128	15	HGM	0.0085	56.226	
EABLA11ID	A	C1	1	1	53.306		0.127	15	HAT	0.0085	53.196	
EABLA219D	A	C2	1	2		7.890	0.129	15	BGM	0.0086		7.955
EABLA21AD	A	C2	1	2		7.978	0.129	15	HGM	0.0086		8.058
EABLA21BD	A	C2	1	2		7.850	0.130	15	HGM	0.0086		7.977
EABLA21CD	A	C2	1	2	53.958		0.130	15	HGM	0.0086	54.904	
EABLA21DD	A	C2	1	2	55.325		0.130	15	HAT	0.0087	56.307	
EABLA21ED	A	C2	1	2	56.636		0.130	15	HGM	0.0086	57.624	
EABLB11BD	B	C1	2	1		8.282	0.126	15	BGM	0.0084		8.158
EABLB11CD	B	C1	2	1		8.212	0.126	15	BGM	0.0084		8.113
EABLB11DD	B	C1	2	1		8.191	0.127	15	HGM	0.0085		8.172
EABLB11ED	B	C1	2	1		8.091	0.128	15	HGM	0.0085		8.137
EABLB11FD	B	C1	2	1	59.720		0.128	15	BGM	0.0086	60.153	
EABLB11GD	B	C1	2	1	60.572		0.129	15	HAB	0.0086	61.373	
EABLB11HD	B	C1	2	1	56.181		0.130	15	BAT	0.0087	57.376	
EABLB11ID	B	C1	2	1	58.058		0.131	15	HGM	0.0087	59.498	
EABLB219D	B	C2	2	2		8.186	0.126	15	HAB	0.0084		8.084
EABLB21AD	B	C2	2	2		7.812	0.127	15	BGM	0.0085		7.800
EABLB21BD	B	C2	2	2		7.750	0.130	15	BGM	0.0087		7.901
EABLB21CD	B	C2	2	2	49.155		0.131	15	HGM	0.0087	50.470	
EABLB21DD	B	C2	2	2	47.952		0.132	15	HGM	0.0088	49.677	
EABLB21FD	B	C2	2	2	55.569		0.132	15	BGM	0.0088	57.498	
EABLC11BD	C	C1	3	1		7.996	0.127	15	HGM	0.0085		7.993
EABLC11CD	C	C1	3	1		7.944	0.128	15	BGM	0.0086		7.993
EABLC11DD	C	C1	3	1		8.063	0.128	15	BAB	0.0085		8.105
EABLC11ED	C	C1	3	1		7.844	0.129	15	HGM	0.0086		7.958
EABLC11FD	C	C1	3	1	51.031		0.129	15	HGM	0.0086	51.691	
EABLC11GD	C	C1	3	1	51.945		0.130	15	HGM	0.0086	52.842	
EABLC11HD	C	C1	3	1	56.451		0.130	15	HGM	0.0087	57.702	
EABLC11JD	C	C1	3	1	56.301		0.130	15	HGM	0.0086	57.251	
EABLC219D	C	C2	3	2		8.061	0.127	15	HGM	0.0085		8.020
EABLC21AD	C	C2	3	2		7.926	0.128	15	BGM	0.0085		7.958
EABLC21BD	C	C2	3	2		7.998	0.129	15	HGM	0.0086		8.077
EABLC21CD	C	C2	3	2	55.923		0.129	15	BGM/HGM	0.0086	56.778	
EABLC21DD	C	C2	3	2	54.879		0.130	15	BGM/HGM	0.0087	55.928	
EABLC21ED	C	C2	3	2	54.641		0.131	15	HGM	0.0087	55.938	

Average	55.442	7.973	Average <sub>norm</sub>	0.0086	56.398	7.987
Standard Dev.	3.457	0.183	Standard Dev. <sub>norm</sub>		3.351	0.141
Coeff. of Var. [%]	6.236	2.294	Coeff. of Var. [%] <sub>norm</sub>		5.942	1.760
Min.	47.952	7.550	Min.	0.0084	49.677	7.622
Max.	60.572	8.282	Max.	0.0088	61.373	8.172
Number of Spec.	21	21	Number of Spec.	42	21	21



## 4.4 Fill Compression Properties (FC)

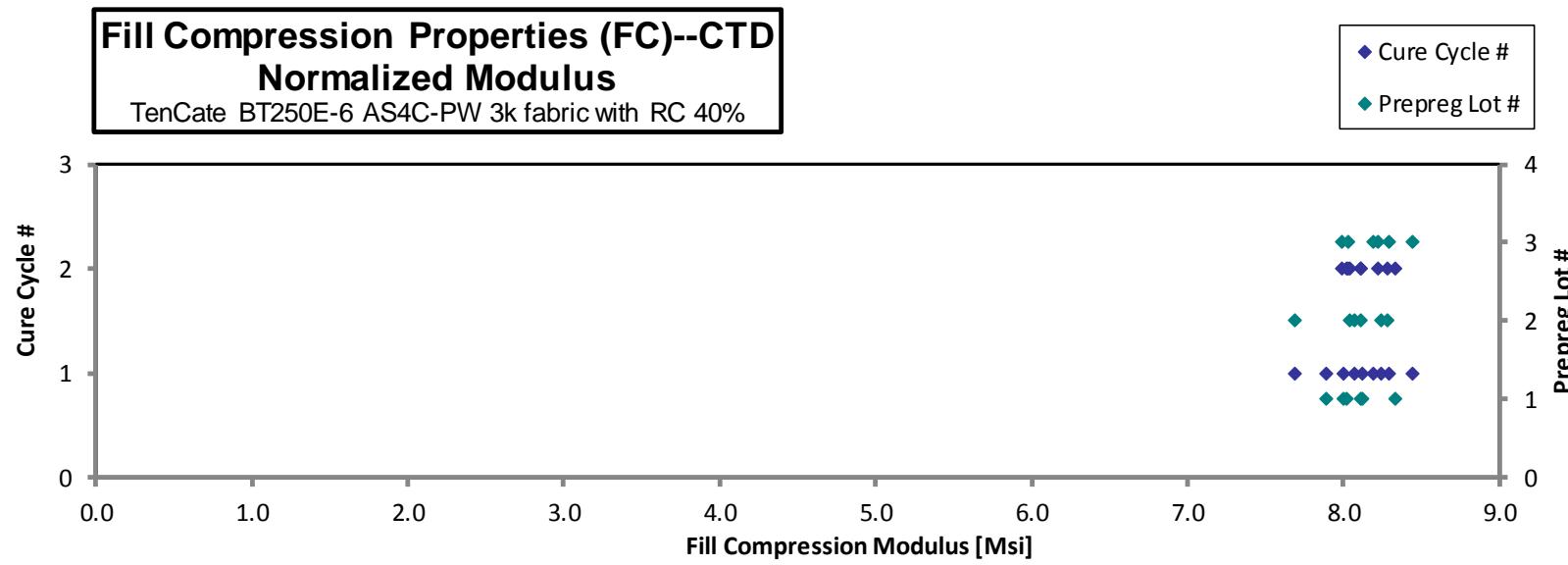
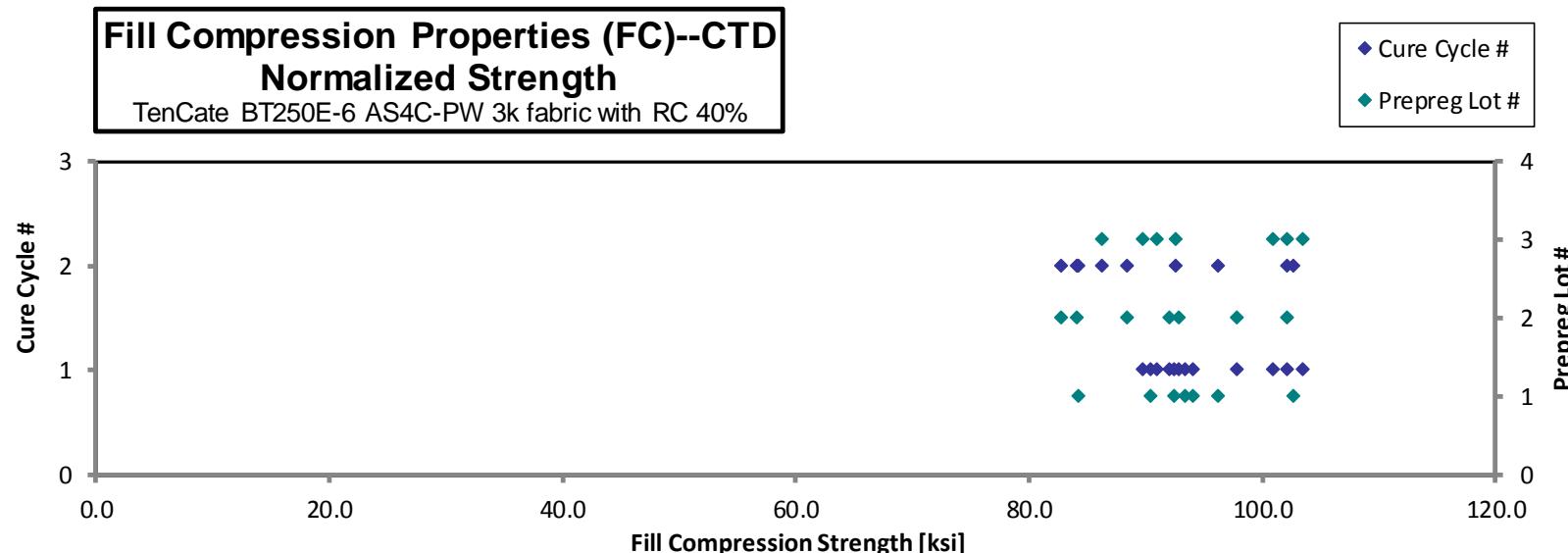
Fill Compression Properties (FC)--CTD Strength & Modulus									normalizing $t_{\text{ply}}$ [in]
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%									0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABZA116B	A	C1	1	1	91.100	7.691	0.131	15	BGM
EABZA117B	A	C1	1	1	87.790	7.771	0.131	15	BGM
EABZA118B	A	C1	1	1	89.410	7.850	0.132	15	BGM
EABZA119B*	A	C1	1	1	90.841		0.132	15	BGM
EABZA215B	A	C2	1	2	96.903	8.394	0.127	15	BGM
EABZA216B	A	C2	1	2	102.989	8.054	0.127	15	BGM
EABZA217B	A	C2	1	2	84.579	8.143	0.127	15	BGM
EABZB116B*	B	C1	2	1	92.839		0.127	15	BGM
EABZB117B	B	C1	2	1	92.286	8.268	0.127	15	BGM
EABZB118B	B	C1	2	1	102.512	8.094	0.127	15	BGM
EABZB119B	B	C1	2	1	98.392	7.736	0.127	15	BGM
EABZB215B	B	C2	2	2	**	8.303	0.127	15	HIT/BGM
EABZB216B	B	C2	2	2	84.317	8.061	0.127	15	BGM
EABZB217B	B	C2	2	2	88.592	8.128	0.127	15	BGM
EABZB218B*	B	C2	2	2	82.690		0.128	15	BGM
EABZC116B	C	C1	3	1	104.268	8.360	0.126	15	BGM
EABZC117B	C	C1	3	1	91.625	8.245	0.127	15	BGM
EABZC118B	C	C1	3	1	101.843	8.526	0.126	15	BGM
EABZC119B*	C	C1	3	1	91.089		0.126	15	BGM
EABZC215B	C	C2	3	2	86.781	8.070	0.127	15	BGM
EABZC216B	C	C2	3	2	102.676	8.036	0.127	15	BGM
EABZC217B	C	C2	3	2	92.664	8.230	0.127	15	BGM

\* Specimen was not gaged and tested for strength only.

\*\* Strength not reported due to dominant bad failure mode.

Average	93.152	8.109	Average <sub>norm</sub>	0.0085	93.260	8.111
Standard Dev.	6.701	0.233	Standard Dev. <sub>norm</sub>	6.368	0.177	
Coeff. of Var. [%]	7.194	2.873	Coeff. of Var. [%] <sub>norm</sub>	6.828	2.181	
Min.	82.690	7.691	Min.	0.0084	82.771	7.688
Max.	104.268	8.526	Max.	0.0088	103.399	8.439
Number of Spec.	21	18	Number of Spec.	22	21	18



**Fill Compression Properties (FC)--RTD  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

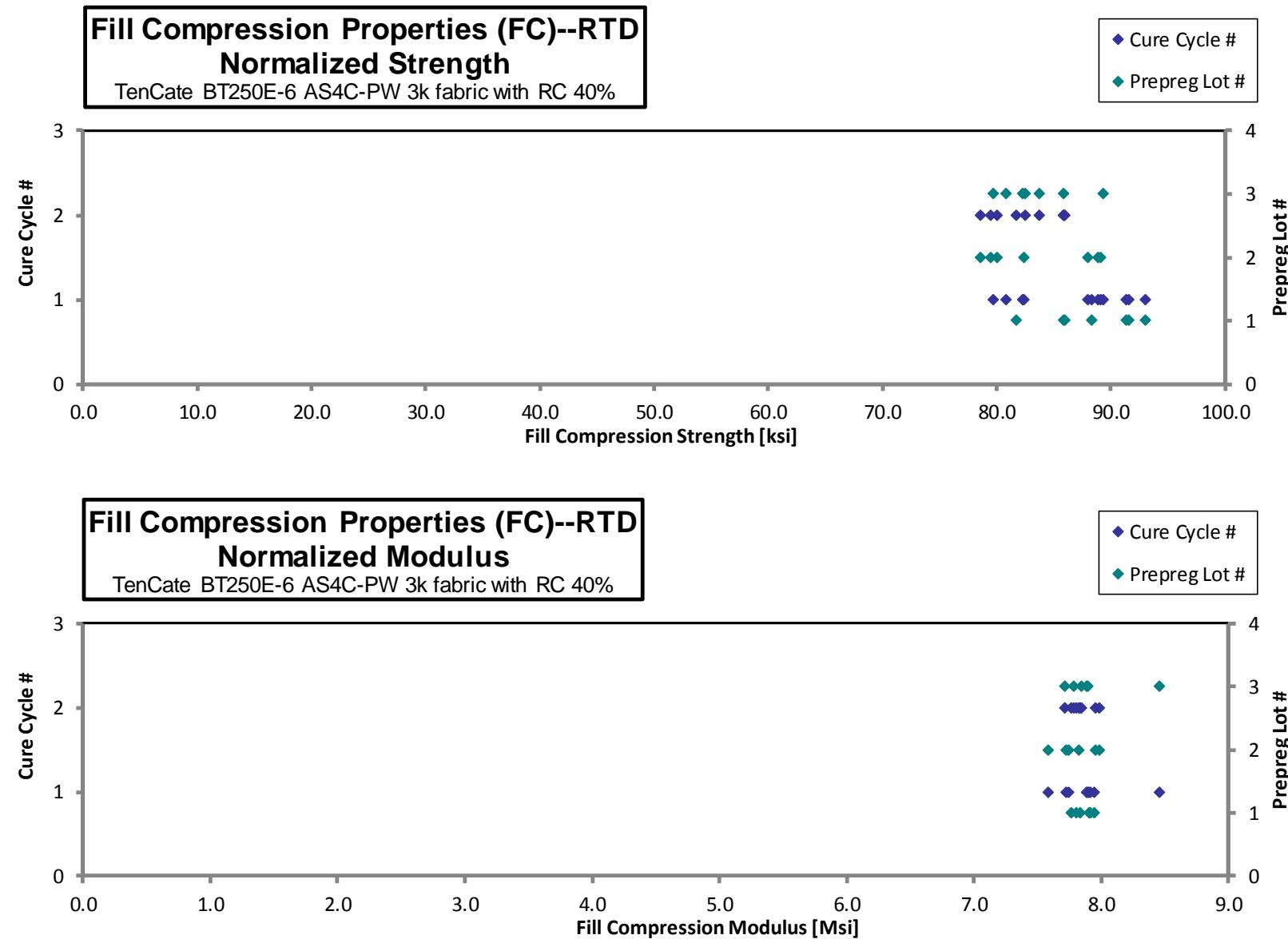
 normalizing  
 $t_{\text{ply}}$  [in]  
 0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABZA111A	A	C1	1	1	92.628	7.875	0.128	15	BGM
EABZA112A	A	C1	1	1	87.261	7.851	0.129	15	BGM/HIB
EABZA113A	A	C1	1	1	89.835	7.780	0.130	15	BGM
EABZA114A*	A	C1	1	1	89.504		0.130	15	BGM
EABZA211A	A	C2	1	2	81.136	7.711	0.128	15	BAT
EABZA212A	A	C2	1	2	85.660	7.790	0.128	15	BGM
EABZA213A	A	C2	1	2	85.888	7.836	0.127	15	BGM
EABZB111A	B	C1	2	1	89.817	7.661	0.126	15	BGM
EABZB112A	B	C1	2	1	89.648	7.799	0.127	15	BGM
EABZB113A	B	C1	2	1	82.748	7.758	0.127	15	BGM
EABZB114A*	B	C1	2	1	88.162		0.127	15	BGM
EABZB211A	B	C2	2	2	79.502	7.916	0.126	15	BGM/HGM
EABZB212A	B	C2	2	2	**	8.009	0.127	15	BGM/HIB
EABZB213A	B	C2	2	2	80.159	8.001	0.127	15	HGM
EABZB214A*	B	C2	2	2	79.634		0.127	15	BGM
EABZC111A	C	C1	3	1	82.008	8.014	0.126	15	BGM
EABZC112A	C	C1	3	1	**	8.596	0.125	15	BGM/HIB
EABZC113A	C	C1	3	1	90.823	8.017	0.125	15	BGM/HGM
EABZC114A*	C	C1	3	1	83.452		0.126	15	BGM
EABZC115A*	C	C1	3	1	80.575		0.126	15	HGM
EABZC211A	C	C2	3	2	81.998	7.797	0.128	15	BGM/HGM
EABZC212A	C	C2	3	2	85.530	7.677	0.128	15	BGM
EABZC213A	C	C2	3	2	**	7.794	0.127	15	BGM/HIB
EABZC214A*	C	C2	3	2	84.075		0.127	15	BGM

\* Specimen was not gaged and tested for strength only.

\*\* Strength not reported due to dominant bad failure mode.

Average	85.240	7.882	Average <sub>norm</sub>	0.0085	85.122	7.862
Standard Dev.	4.118	0.211	Standard Dev. <sub>norm</sub>	4.433	0.178	
Coeff. of Var. [%]	4.832	2.676	Coeff. of Var. [%] <sub>norm</sub>	5.207	2.269	
Min.	79.502	7.661	Min.	0.0084	78.582	7.576
Max.	92.628	8.596	Max.	0.0087	93.019	8.450
Number of Spec.	21	18	Number of Spec.	24	21	18



**Fill Compression Properties (FC)--ETD  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

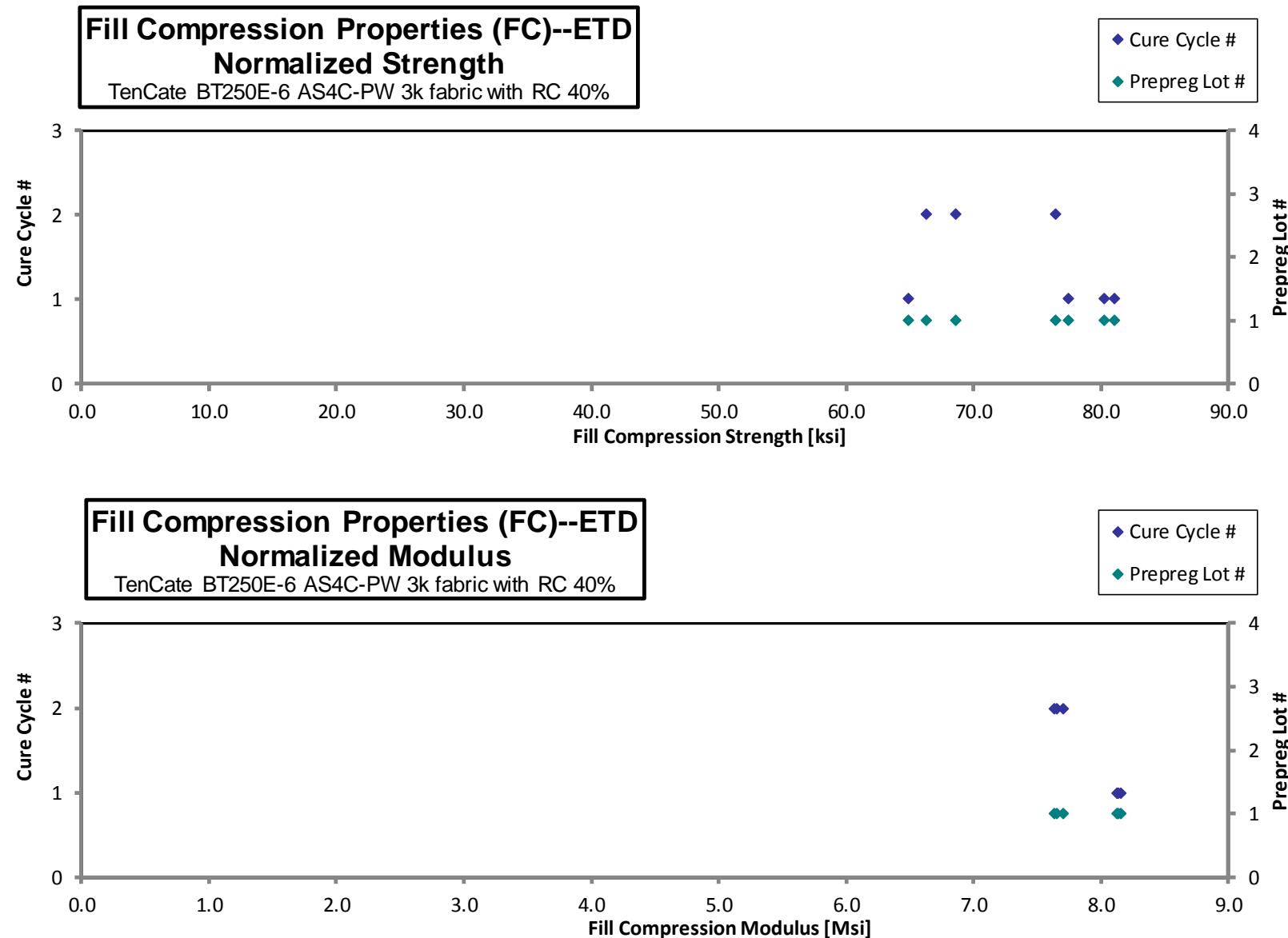
normalizing  
 $t_{\text{ply}}$  [in]  
0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABZA11BC	A	C1	1	1	77.377	7.868	0.132	15	BGM
EABZA11CC	A	C1	1	1	78.115	7.840	0.132	15	BGM/CIT
EABZA11DC	A	C1	1	1	74.576	7.819	0.132	15	BGM/HIB
EABZA11EC*	A	C1	1	1	64.293		0.129	15	BGM
EABZA219C	A	C2	1	2	65.575	7.555	0.129	15	BGM
EABZA21AC	A	C2	1	2	68.173	7.597	0.128	15	BGM
EABZA21BC	A	C2	1	2	**	7.663	0.128	15	HIT/BGM
EABZA21CC*	A	C2	1	2	76.281		0.128	15	BAB

\* Specimen was not gaged and tested for strength only.

\*\* Strength not reported due to dominant bad failure mode.

Average	72.056	7.724	Average <sub>norm</sub>	0.0087	73.543	7.896
Standard Dev.	5.868	0.135	Standard Dev. <sub>norm</sub>		6.795	0.263
Coeff. of Var. [%]	8.143	1.751	Coeff. of Var. [%] <sub>norm</sub>		9.239	3.329
Min.	64.293	7.555	Min.	0.0085	64.866	7.629
Max.	78.115	7.868	Max.	0.0088	81.002	8.157
Number of Spec.	7	6	Number of Spec.	8	7	6



**Fill Compression Properties (FC)--ETW**  
**Strength & Modulus**

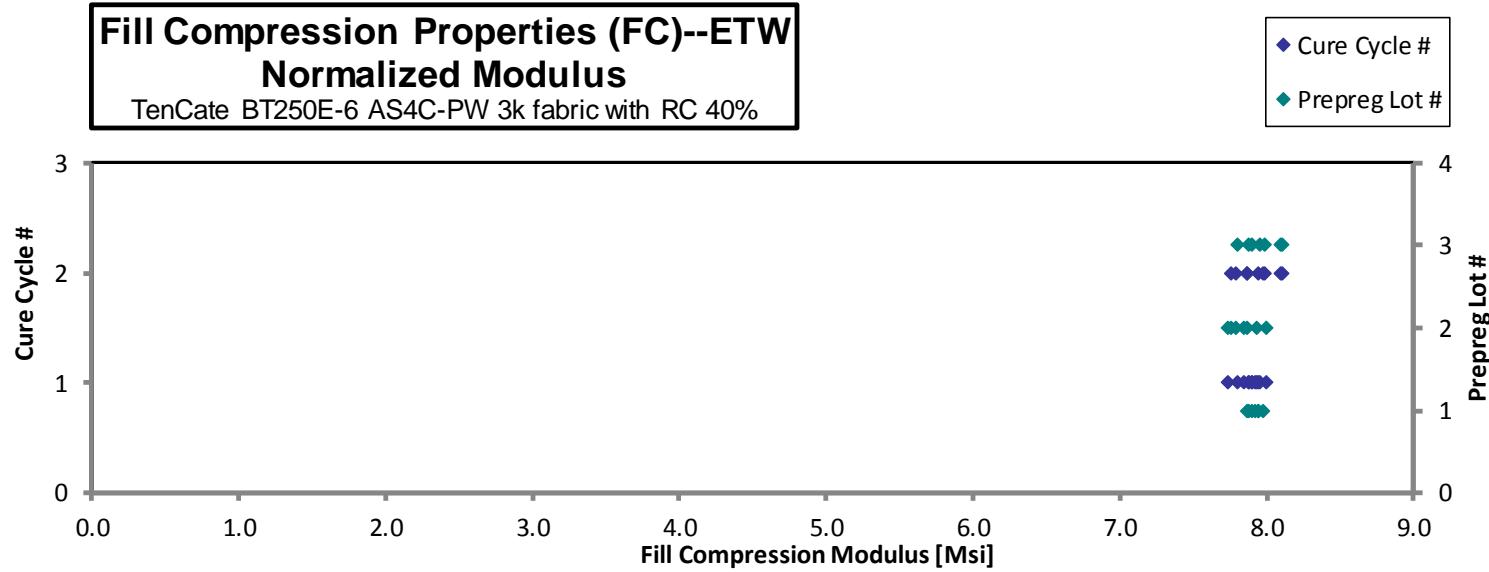
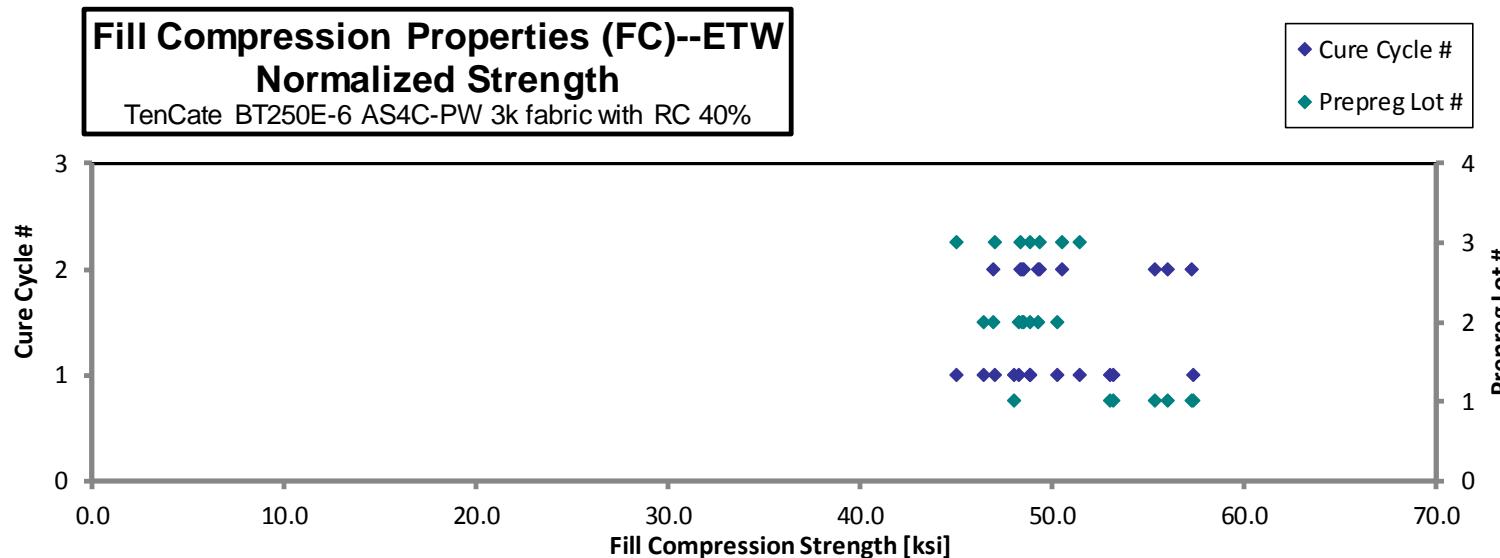
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

normalizing  
 $t_{\text{ply}}$  [in]  
 0.0085

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Failure Mode
EABZA11GD	A	C1	1	1		7.735	0.130	15	HGM
EABZA11HD	A	C1	1	1		7.682	0.131	15	HGM
EABZA11D	A	C1	1	1		7.731	0.131	15	HGM
EABZA11JD	A	C1	1	1		7.737	0.131	15	HGM
EABZA11KD	A	C1	1	1	51.567		0.131	15	HGM
EABZA11LD	A	C1	1	1	55.329		0.132	15	HGM
FABZA11MD	A	C1	1	1	46.341		0.132	15	HAB
EABZA11ND	A	C1	1	1	51.220		0.132	15	HGM
EABZA21DD	A	C2	1	2		7.908	0.128	15	BGM
EABZA21ED	A	C2	1	2		7.829	0.128	15	BGM
EABZA21FD	A	C2	1	2		7.930	0.128	15	HGM
EABZA21GD	A	C2	1	2	54.917		0.129	15	HGM
EABZA21HD	A	C2	1	2	55.730		0.131	15	HGM
EABZA21ID	A	C2	1	2	54.327		0.131	15	BGM
EABZB11BD	B	C1	2	1		7.728	0.128	15	HGM
EABZB11CD	B	C1	2	1		7.833	0.128	15	HGM
EABZB11DD	B	C1	2	1		7.971	0.128	15	HGM
EABZB11ED	B	C1	2	1		7.891	0.128	15	HGM
EABZB11FD	B	C1	2	1	46.106		0.128	15	HGM
EABZB11GD	B	C1	2	1	48.439		0.128	15	HGM
EABZB11HD	B	C1	2	1	49.902		0.128	15	HGM
EABZB11ID	B	C1	2	1	47.896		0.128	15	HGM
EABZB219D	B	C2	2	2		7.923	0.127	15	HGM
EABZB21BD	B	C2	2	2		7.833	0.127	15	HGM
EABZB21CD	B	C2	2	2	48.435		0.127	15	HGM
EABZB21DD	B	C2	2	2	49.434		0.127	15	HGM
EABZB21ED	B	C2	2	2	47.216		0.127	15	BGM
EABZB21FD*	B	C2	2	2	48.852	7.814	0.127	15	HGM
EABZC11BD	C	C1	3	1		7.964	0.125	15	BGM
EABZC11CD	C	C1	3	1		8.035	0.125	15	HGM
EABZC11DD	C	C1	3	1		7.978	0.126	15	BGM
EABZC11ED	C	C1	3	1		8.060	0.126	15	HGM
EABZC11FD	C	C1	3	1	45.571		0.126	15	HGM
EABZC11GD	C	C1	3	1	49.172		0.127	15	HGM
EABZC11HD	C	C1	3	1	47.295		0.127	15	HGM
EABZC11ID	C	C1	3	1	51.750		0.127	15	HGM
EABZC219D	C	C2	3	2		7.896	0.129	15	BGM
EABZC21AD	C	C2	3	2		8.052	0.128	15	HGM
EABZC21BD	C	C2	3	2		8.065	0.128	15	HGM
FABZC21CD	C	C2	3	2	48.169		0.128	15	HGM
EABZC21DD	C	C2	3	2	50.245		0.128	15	BGM
EABZC21ED	C	C2	3	2	49.104		0.128	15	BGM

\*Strength is reported because SG was gaged after conditioning with a room temp cure adhesive.

Average	49.864	7.886	Average <sub>norm</sub>	0.0086	50.327	7.906
Standard Dev.	3.007	0.120	Standard Dev. <sub>norm</sub>		3.553	0.097
Coeff. of Var. [%]	6.029	1.521	Coeff. of Var. [%] <sub>norm</sub>		7.059	1.222
Min.	45.571	7.682	Min.	0.0083	44.995	7.737
Max.	55.730	8.065	Max.	0.0088	57.347	8.101
Number of Spec.	22	21	Number of Spec.	42	22	21

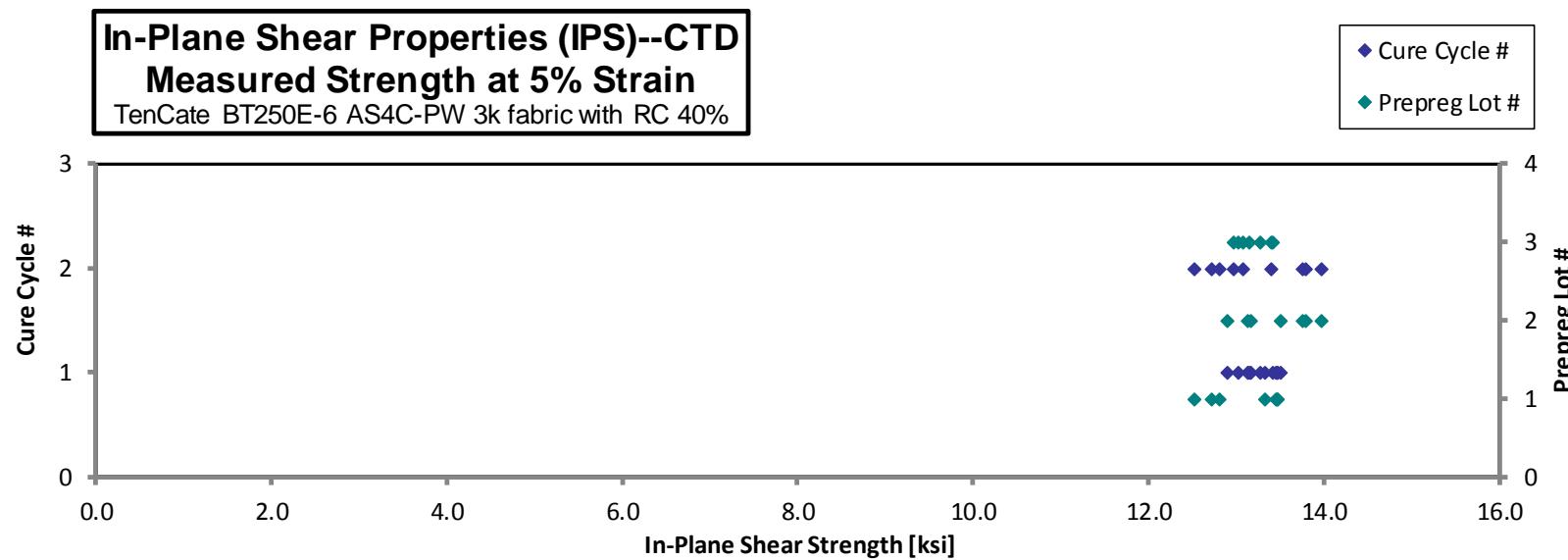
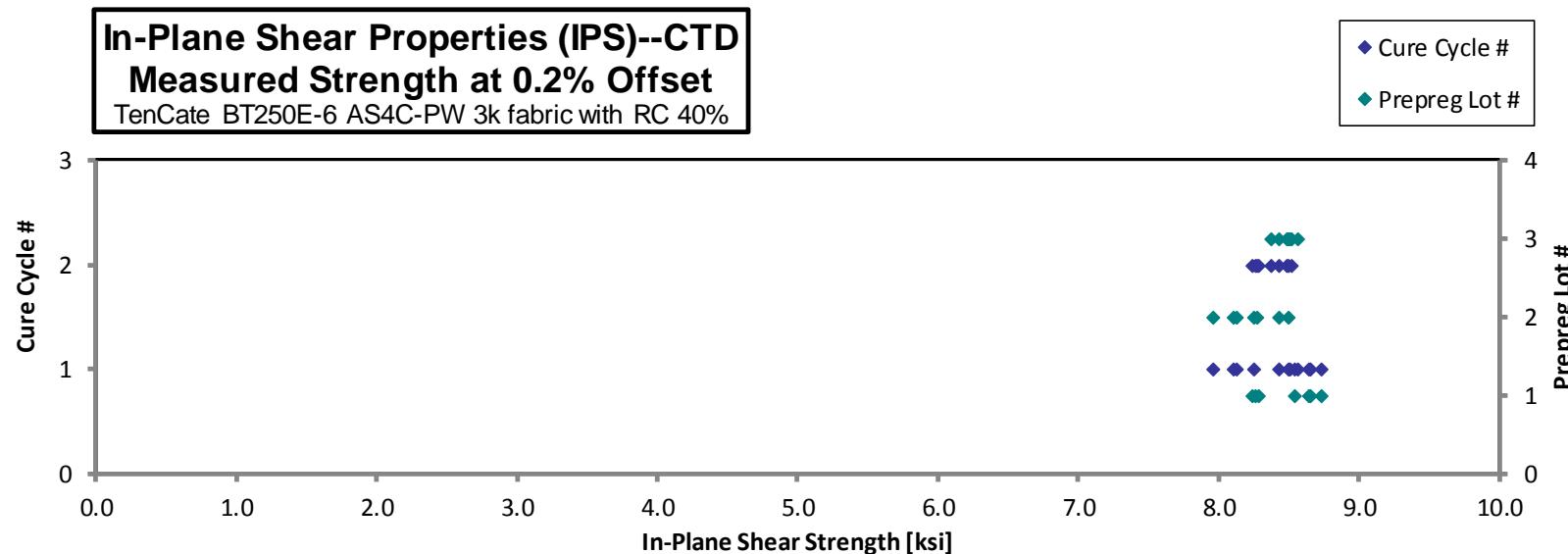


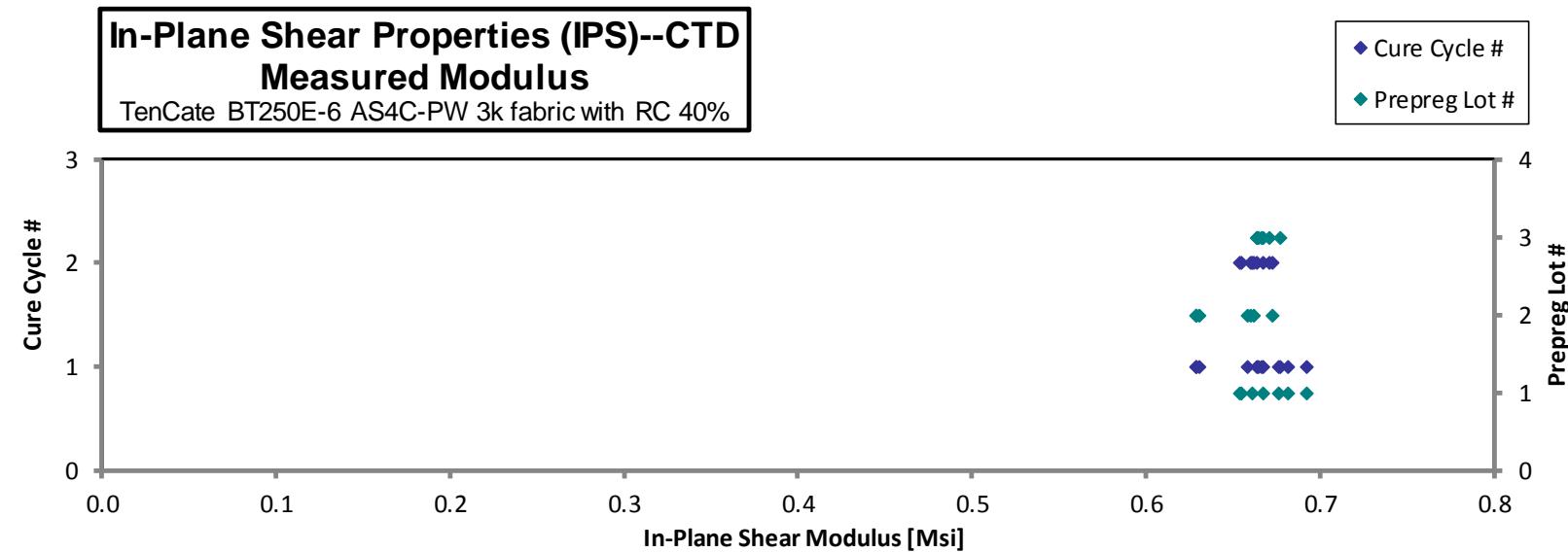
## 4.5 In-Plane Shear Properties (IPS)

In-Plane Shear Properties (IPS)--CTD Strength & Modulus									
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%									

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	0.2% Offset Strength [ksi]	Strength at 5% Strain [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. t <sub>ply</sub> [in]
EABNA116B	A	C1	1	1	8.538	13.444	0.676	0.102	12	0.0085
EABNA117B	A	C1	1	1	8.636	13.465	0.667	0.102	12	0.0085
EABNA118B	A	C1	1	1	8.646	13.472	0.681	0.102	12	0.0085
EABNA119B	A	C1	1	1	8.724	13.316	0.692	0.099	12	0.0082
EABNA215B	A	C2	1	2	8.285	12.717	0.660	0.105	12	0.0087
EABNA216B	A	C2	1	2	8.254	12.799	0.654	0.105	12	0.0087
EABNA217B	A	C2	1	2	8.237	12.517	0.655	0.103	12	0.0085
EABNB116B	B	C1	2	1	8.101	13.163	0.630	0.108	12	0.0090
EABNB117B	B	C1	2	1	8.129	13.133	0.629	0.107	12	0.0089
EABNB118B	B	C1	2	1	7.957	12.890	0.629	0.106	12	0.0088
EABNB119B	B	C1	2	1	8.245	13.497	0.658	0.105	12	0.0088
EABNB215B	B	C2	2	2	8.424	13.973	0.662	0.103	12	0.0085
EABNB216B	B	C2	2	2	8.274	13.786	0.659	0.103	12	0.0086
EABNB217B	B	C2	2	2	8.490	13.742	0.672	0.102	12	0.0085
EABNC116B	C	C1	3	1	8.506	13.010	0.666	0.102	12	0.0085
EABNC117B	C	C1	3	1	8.564	13.406	0.677	0.102	12	0.0085
EABNC118B	C	C1	3	1	8.494	13.137	0.664	0.102	12	0.0085
EABNC119B	C	C1	3	1	8.426	13.271	0.664	0.103	12	0.0085
EABNC215B	C	C2	3	2	8.373	12.955	0.670	0.103	12	0.0086
EABNC216B	C	C2	3	2	8.487	13.074	0.663	0.103	12	0.0086
EABNC217B	C	C2	3	2	8.515	13.392	0.667	0.103	12	0.0085

Average	8.396	13.246	0.662	Average	0.0086
Standard Dev.	0.197	0.365	0.016	Standard Dev.	
Coeff. of Var. [%]	2.348	2.753	2.460	Coeff. of Var. [%]	
Min.	7.957	12.517	0.629	Min.	0.0082
Max.	8.724	13.973	0.692	Max.	0.0090
Number of Spec.	21	21	21	Number of Spec.	21



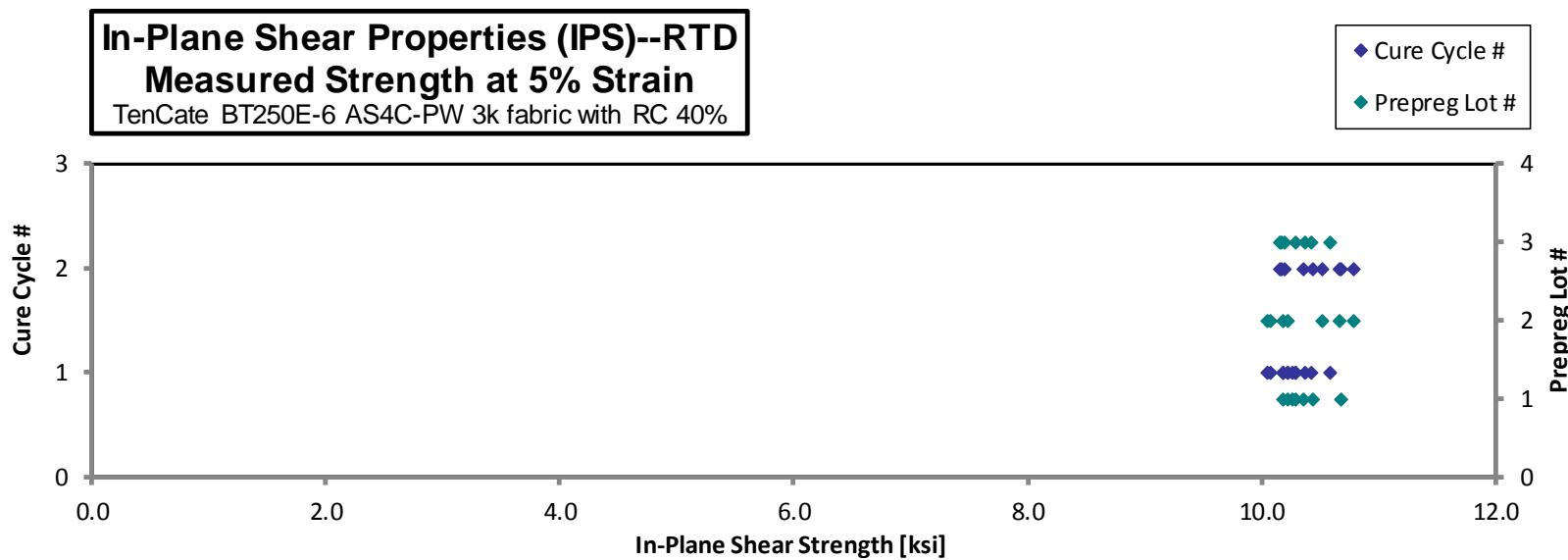
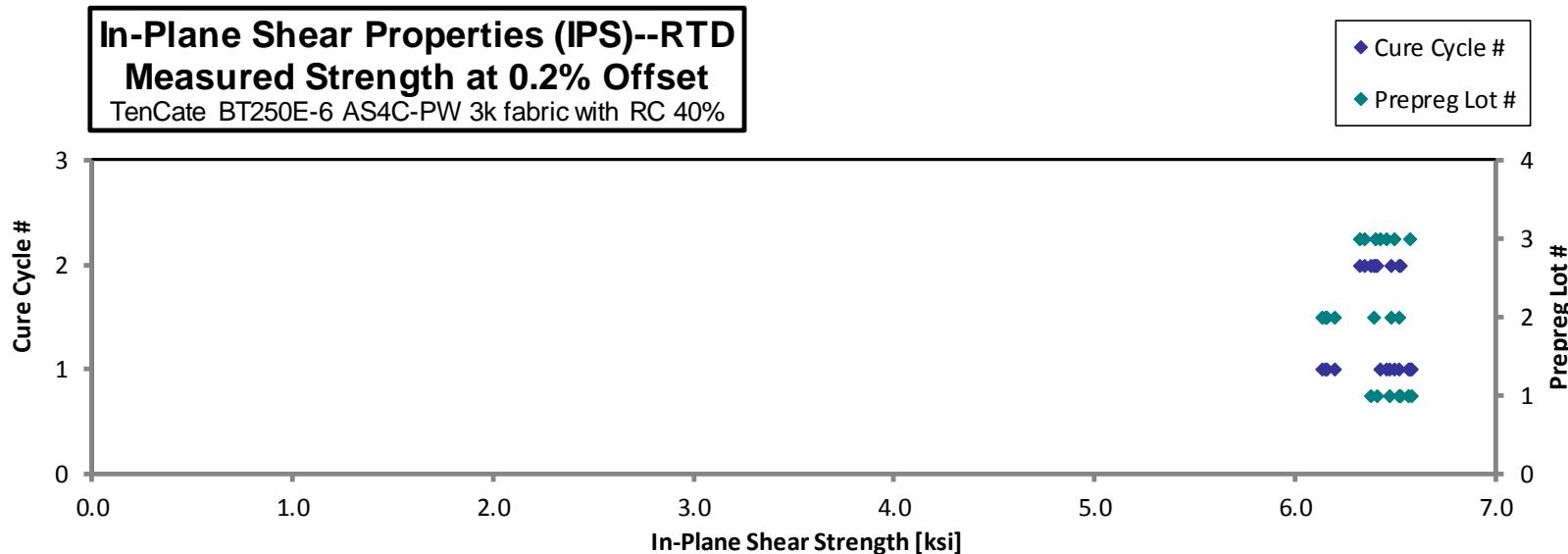


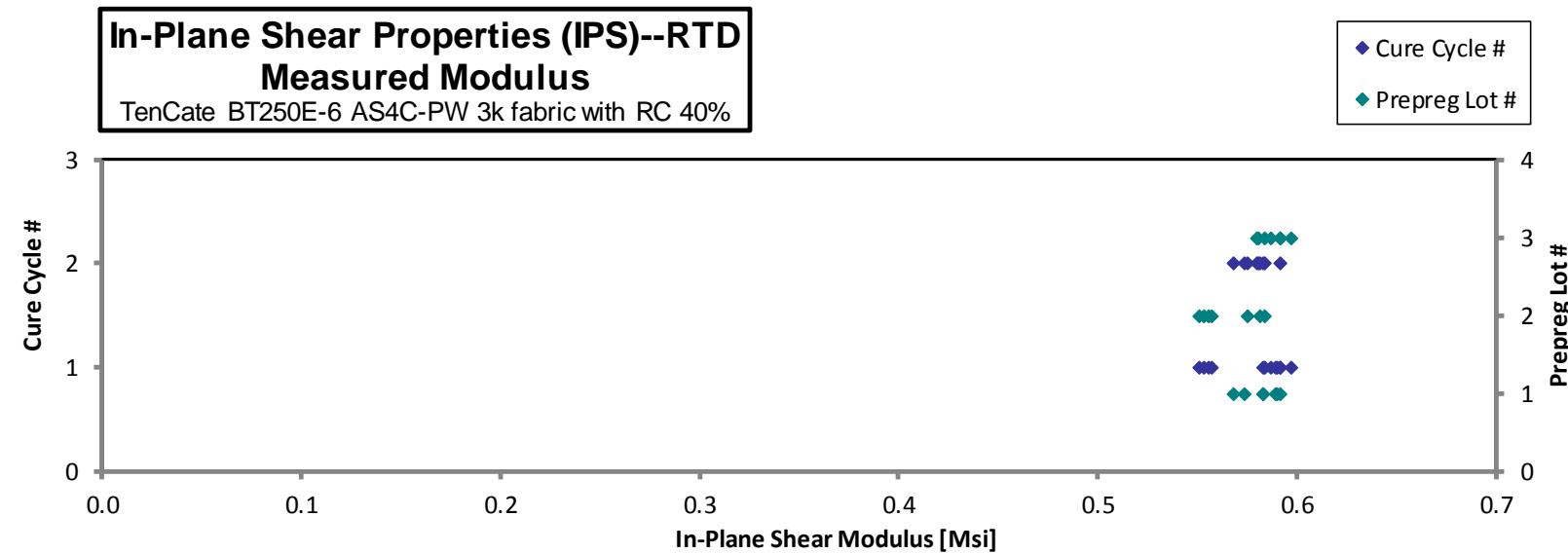
**In-Plane Shear Properties (IPS)--RTD  
Strength & Modulus**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	0.2% Offset Strength [ksi]	Strength at 5% Strain [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. t <sub>ply</sub> [in]
EABNA111A	A	C1	1	1	6.583	10.217	0.590	0.100	12	0.0083
EABNA112A	A	C1	1	1	6.472	10.185	0.583	0.101	12	0.0084
EABNA113A	A	C1	1	1	6.562	10.291	0.589	0.101	12	0.0084
EABNA114A	A	C1	1	1	6.515	10.260	0.591	0.101	12	0.0085
EABNA211A	A	C2	1	2	6.527	10.668	0.583	0.102	12	0.0085
EABNA212A	A	C2	1	2	6.373	10.430	0.568	0.105	12	0.0087
EABNA213A	A	C2	1	2	6.404	10.353	0.573	0.104	12	0.0087
EABNB111A	B	C1	2	1	6.159	10.050	0.553	0.105	12	0.0088
EABNB112A	B	C1	2	1	6.134	10.072	0.556	0.106	12	0.0088
EABNB113A	B	C1	2	1	6.198	10.218	0.557	0.107	12	0.0089
EABNB114A	B	C1	2	1	6.145	10.176	0.550	0.107	12	0.0089
EABNB211A	B	C2	2	2	6.393	10.520	0.575	0.102	12	0.0085
EABNB212A	B	C2	2	2	6.479	10.660	0.582	0.102	12	0.0085
EABNB213A	B	C2	2	2	6.516	10.779	0.584	0.102	12	0.0085
EABNC111A	C	C1	3	1	6.421	10.288	0.584	0.102	12	0.0085
EABNC112A	C	C1	3	1	6.489	10.418	0.591	0.101	12	0.0084
EABNC113A	C	C1	3	1	6.572	10.587	0.597	0.101	12	0.0084
EABNC114A	C	C1	3	1	6.453	10.360	0.586	0.102	12	0.0085
EABNC211A	C	C2	3	2	6.403	10.149	0.591	0.101	12	0.0084
EABNC212A	C	C2	3	2	6.342	10.170	0.580	0.103	12	0.0086
EABNC213A	C	C2	3	2	6.324	10.189	0.579	0.104	12	0.0086

Average	6.403	10.335	0.578	Average	0.0086
Standard Dev.	0.141	0.206	0.014	Standard Dev.	
Coeff. of Var. [%]	2.207	1.991	2.387	Coeff. of Var. [%]	
Min.	6.134	10.050	0.550	Min.	0.0083
Max.	6.583	10.779	0.597	Max.	0.0089
Number of Spec.	21	21	21	Number of Spec.	21





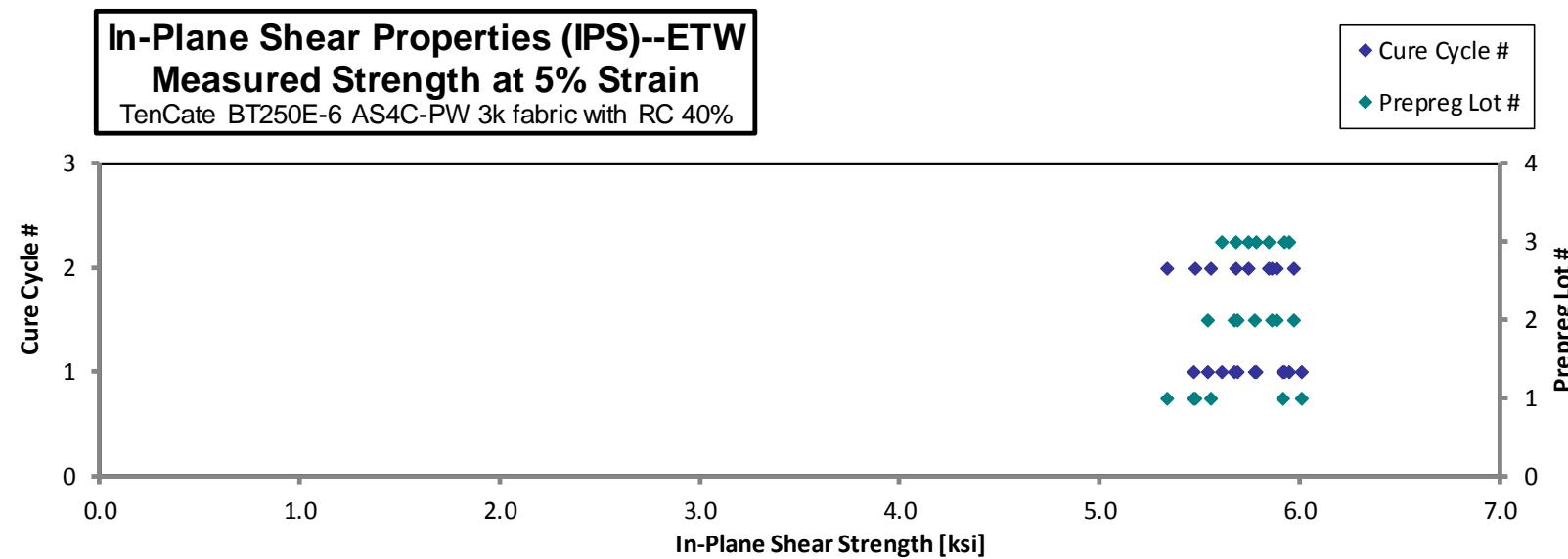
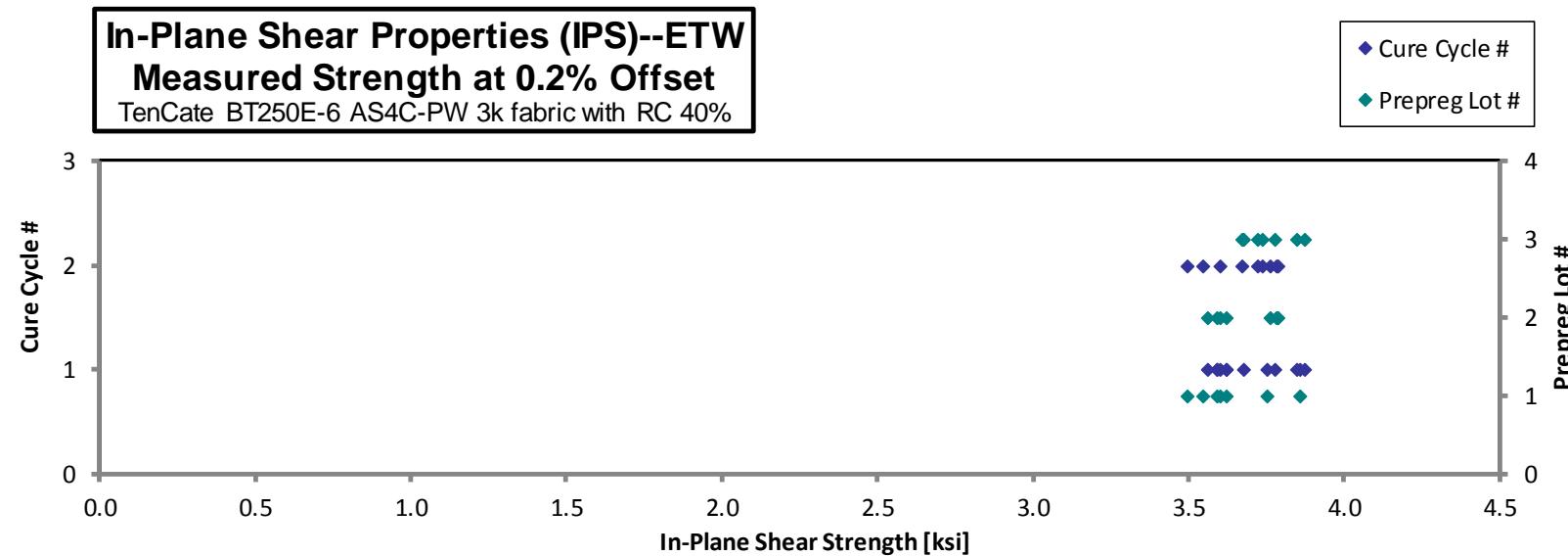
**In-Plane Shear Properties (IPS)--ETW****Strength & Modulus**

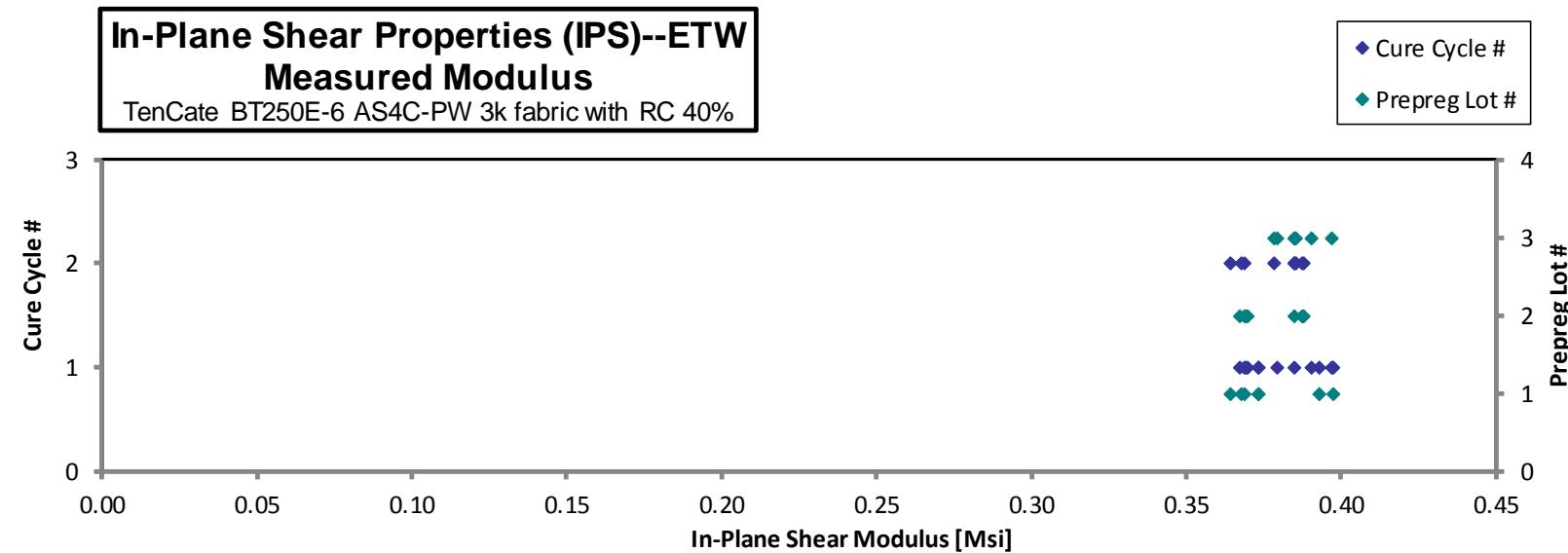
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	0.2% Offset Strength [ksi]	Strength at 5% Strain [ksi]	Modulus [Msi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. $t_{\text{ply}}$ [in]
EABNA11BD	A	C1	1	1	3.859	6.010	0.393	0.101	12	0.0084
EABNA11CD	A	C1	1	1	3.751	5.916	0.398	0.102	12	0.0085
EABNA11DD*	A	C1	1	1	3.590		0.373	0.102	12	0.0085
EABNA11ED	A	C1	1	1	3.619	5.466	0.373	0.102	12	0.0085
EABNA219D	A	C2	1	2	3.497	5.337	0.364	0.103	12	0.0086
EABNA21AD	A	C2	1	2	3.545	5.474	0.368	0.103	12	0.0086
EABNA21BD	A	C2	1	2	3.598	5.555	0.369	0.104	12	0.0087
EABNB11BD	B	C1	2	1	3.559	5.538	0.367	0.107	12	0.0089
EABNB11CD	B	C1	2	1	3.619	5.772	0.369	0.108	12	0.0090
EABNB11DD	B	C1	2	1	3.600	5.689	0.370	0.107	12	0.0089
EABNB11ED	B	C1	2	1	3.588	5.670	0.368	0.107	12	0.0089
EABNB219D	B	C2	2	2	3.780	5.882	0.388	0.102	12	0.0085
EABNB21AD	B	C2	2	2	3.785	5.967	0.387	0.102	12	0.0085
EABNB21BD	B	C2	2	2	3.761	5.862	0.385	0.101	12	0.0084
EABNC11BD	C	C1	3	1	3.846	5.918	0.397	0.101	12	0.0084
EABNC11CD	C	C1	3	1	3.679	5.612	0.379	0.102	12	0.0085
EABNC11DD	C	C1	3	1	3.873	5.944	0.390	0.103	12	0.0085
EABNC11ED	C	C1	3	1	3.776	5.784	0.385	0.103	12	0.0086
EABNC219D	C	C2	3	2	3.735	5.741	0.385	0.103	12	0.0086
EABNC21AD	C	C2	3	2	3.719	5.844	0.385	0.104	12	0.0087
EABNC21BD	C	C2	3	2	3.673	5.681	0.378	0.104	12	0.0087

\* Strength at 5% strain is not available because strain gage failed prior to reaching 5% strain.

Average	3.688	5.733	0.379	Average	0.0086
Standard Dev.	0.111	0.190	0.010	Standard Dev.	
Coeff. of Var. [%]	3.004	3.319	2.760	Coeff. of Var. [%]	
Min.	3.497	5.337	0.364	Min.	0.0084
Max.	3.873	6.010	0.398	Max.	0.0090
Number of Spec.	21	20	21	Number of Spec.	21





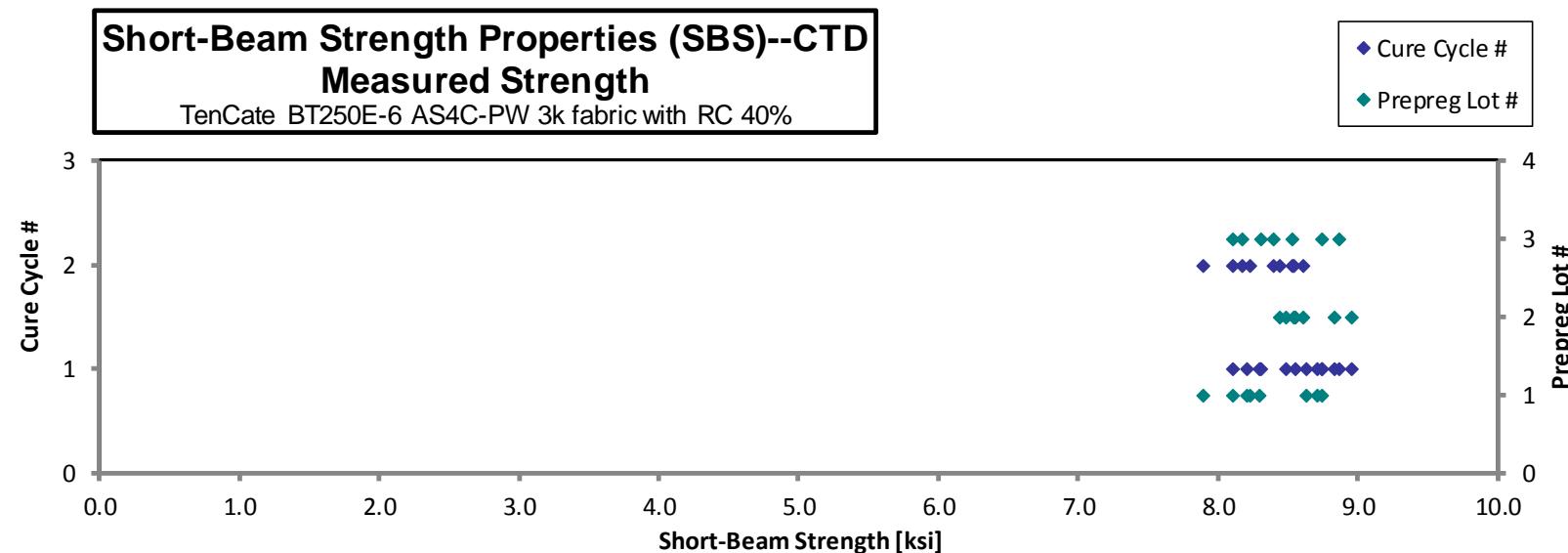
## 4.6 Lamina Short-Beam Strength Properties (SBS)

Short-Beam Strength Properties (SBS)--CTD Strength									
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%									
Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. $t_{\text{ply}}$ [in]	Failure Mode
EABQA116B*	A	C1	1	1	8.631	0.268	32	0.0084	COMPRESSION
EABQA117B	A	C1	1	1	8.207	0.270	32	0.0084	ILS
EABQA118B	A	C1	1	1	8.292	0.272	32	0.0085	ILS
EABQA119B	A	C1	1	1	8.702	0.273	32	0.0085	ILS
EABQA11AB	A	C1	1	1	8.736	0.273	32	0.0085	ILS
EABQA215B	A	C2	1	2	7.885	0.268	32	0.0084	1st: COMPRESSION/ 2nd: ILS
EABQA216B	A	C2	1	2	8.107	0.271	32	0.0085	ILS
EABQA217B	A	C2	1	2	8.230	0.273	32	0.0085	ILS
EABQB116B	B	C1	2	1	8.482	0.264	32	0.0083	1st: ILS/ 2nd: COMPRESSION
EABQB117B	B	C1	2	1	8.550	0.267	32	0.0083	ILS
EABQB119B	B	C1	2	1	8.948	0.269	32	0.0084	ILS
EABQB11AB	B	C1	2	1	8.833	0.269	32	0.0084	ILS
EABQB214B	B	C2	2	2	8.605	0.261	32	0.0081	ILS
EABQB215B	B	C2	2	2	8.534	0.261	32	0.0082	ILS
EABQB216B	B	C2	2	2	8.433	0.262	32	0.0082	1st: ILS/ 2nd: COMPRESSION
EABQC115B	C	C1	3	1	8.738	0.267	32	0.0084	ILS
EABQC116B	C	C1	3	1	8.098	0.264	32	0.0082	ILS
EABQC117B	C	C1	3	1	8.304	0.267	32	0.0083	ILS
EABQC118B	C	C1	3	1	8.857	0.266	32	0.0083	ILS
EABQC215B	C	C2	3	2	8.167	0.258	32	0.0081	ILS
EABQC216B	C	C2	3	2	8.389	0.259	32	0.0081	ILS
EABQC217B	C	C2	3	2	8.524	0.261	32	0.0082	ILS

\*4T span was used.

Span of 4.5T was used for all specimens except EABQA116B.

Average	8.466	Average	0.0083
Standard Dev.	0.281	Standard Dev.	
Coeff. of Var. [%]	3.323	Coeff. of Var. [%]	
Min.	7.885	Min.	0.0081
Max.	8.948	Max.	0.0085
Number of Spec.	22	Number of Spec.	22



**Short-Beam Strength Properties (SBS)--RTD  
Strength**

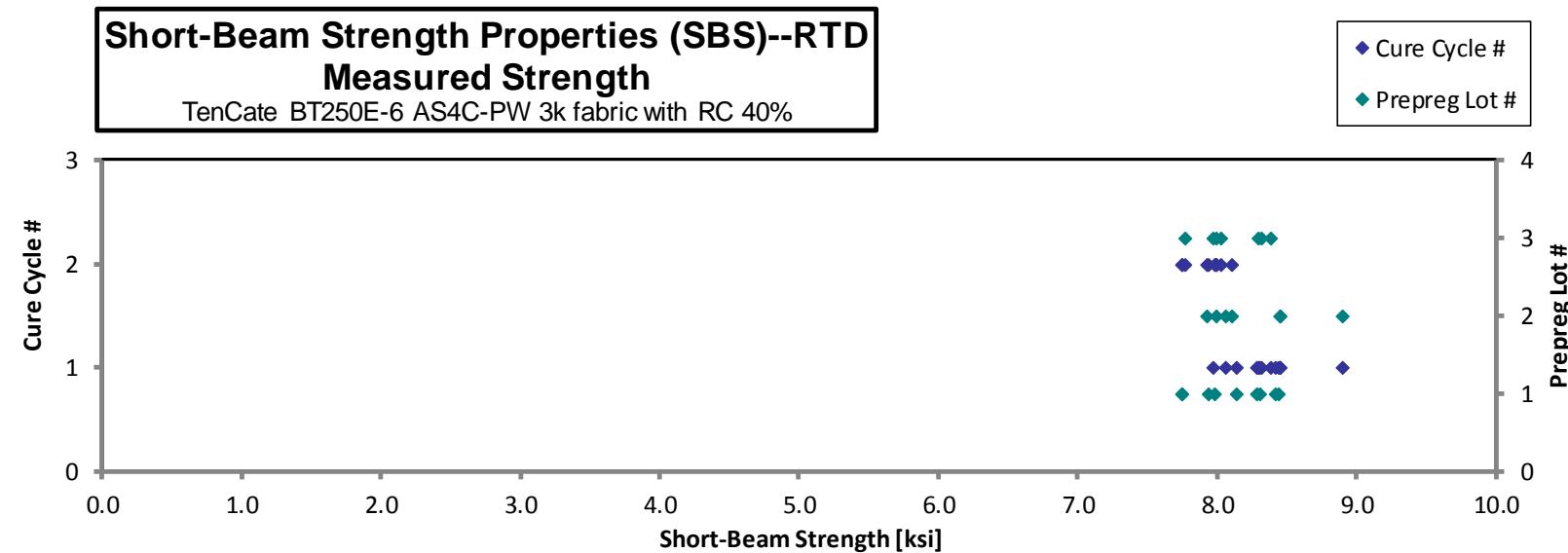
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. $t_{\text{ply}}$ [in]	Failure Mode
EABQA11A*	A	C1	1	1	8.130	0.264	32	0.0082	COMPRESSION
EABQA112A*	A	C1	1	1	8.309	0.266	32	0.0083	COMPRESSION
EABQA113A*	A	C1	1	1	8.417	0.267	32	0.0084	COMPRESSION
EABQA114A	A	C1	1	1	8.437	0.269	32	0.0084	ILS
EABQA115A	A	C1	1	1	8.275	0.269	32	0.0084	1st: COMPRESSION/ 2nd: ILS
EABQA211A	A	C2	1	2	7.746	0.262	32	0.0082	1st: ILS / 2nd: COMPRESSION
EABQA212A	A	C2	1	2	7.933	0.266	32	0.0083	ILS
EABQA213A	A	C2	1	2	7.978	0.268	32	0.0084	1st: ILS / 2nd: COMPRESSION
EABQA214A	A	C2	1	2	7.745	0.270	32	0.0084	1st: COMPRESSION/ 2nd: ILS
EABQB111A	B	C1	2	1	8.063	0.263	32	0.0082	COMPRESSION
EABQB112A	B	C1	2	1	8.454	0.264	32	0.0082	1st: ILS / 2nd: COMPRESSION
EABQB113A	B	C1	2	1	8.447	0.266	32	0.0083	1st: ILS / 2nd: COMPRESSION
EABQB114A	B	C1	2	1	8.896	0.267	32	0.0083	COMPRESSION
EABQB211A	B	C2	2	2	7.922	0.258	32	0.0081	1st:COMPRESSION / 2nd: ILS
EABQB212A	B	C2	2	2	7.990	0.260	32	0.0081	ILS
EABQB213A	B	C2	2	2	8.108	0.261	32	0.0082	1st: ILS / 2nd: COMPRESSION
EABQC111A	C	C1	3	1	7.966	0.262	32	0.0082	ILS
EABQC112A	C	C1	3	1	8.386	0.264	32	0.0082	COMPRESSION
EABQC113A	C	C1	3	1	8.296	0.265	32	0.0083	ILS
EABQC114A	C	C1	3	1	8.320	0.266	32	0.0083	COMPRESSION
EABQC211A	C	C2	3	2	7.766	0.257	32	0.0080	1st: ILS/ 2nd: COMPRESSION
EABQC212A	C	C2	3	2	7.994	0.259	32	0.0081	1st: ILS/ 2nd: COMPRESSION
EABQC213A	C	C2	3	2	8.025	0.260	32	0.0081	ILS

\*4T span was used.

Span of 4.5T was used for all specimens except EABQA111A, EABQA112A, EABQA113A.

Average	8.157	Average	0.0082
Standard Dev.	0.282	Standard Dev.	
Coeff. of Var. [%]	3.458	Coeff. of Var. [%]	
Min.	7.745	Min.	0.0080
Max.	8.896	Max.	0.0084
Number of Spec.	23	Number of Spec.	23



**Short-Beam Strength Properties (SBS)--ETD  
Strength**

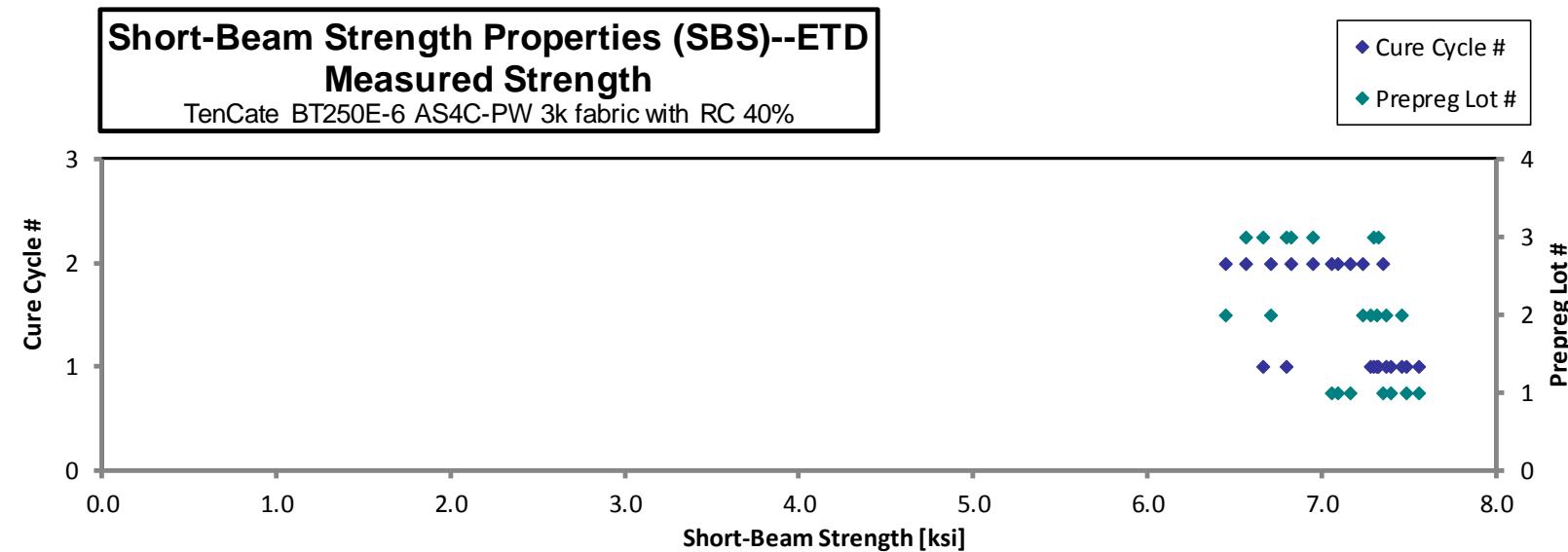
TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. $t_{\text{ply}}$ [in]	Failure Mode
EABQA11BC*	A	C1	1	1	7.554	0.269	32	0.0084	COMPRESSION/ ILS
EABQA11CC	A	C1	1	1	7.398	0.270	32	0.0085	ILS
EABQA11FC	A	C1	1	1	7.483	0.275	32	0.0086	ILS/ COMPRESSION
EABQA219C	A	C2	1	2	7.055	0.270	32	0.0084	COMPRESSION/ ILS
EABQA21AC	A	C2	1	2	7.087	0.273	32	0.0085	COMPRESSION/ ILS
EABQA21BC	A	C2	1	2	7.164	0.275	32	0.0086	ILS
EABQA21CC	A	C2	1	2	7.349	0.277	32	0.0087	ILS
EABQB11BC	B	C1	2	1	7.370	0.267	32	0.0083	COMPRESSION/ ILS
EABQB11CC	B	C1	2	1	7.456	0.269	32	0.0084	ILS
EABQB11DC	B	C1	2	1	7.315	0.268	32	0.0084	COMPRESSION/ ILS
EABQB11EC	B	C1	2	1	7.276	0.269	32	0.0084	ILS/ COMPRESSION
EABQB218C	B	C2	2	2	7.228	0.263	32	0.0082	COMPRESSION/ ILS
EABQB219C	B	C2	2	2	6.443	0.264	32	0.0082	COMPRESSION/ ILS
EABQB21AC	B	C2	2	2	6.709	0.265	32	0.0083	COMPRESSION/ ILS
EABQC11AC	C	C1	3	1	6.799	0.269	32	0.0084	ILS/ COMPRESSION
EABQC11BC	C	C1	3	1	6.659	0.262	32	0.0082	ILS/ COMPRESSION
EABQC11CC	C	C1	3	1	7.322	0.263	32	0.0082	ILS/ COMPRESSION
EABQC11DC	C	C1	3	1	7.291	0.265	32	0.0083	ILS/ COMPRESSION
EABQC21AC	C	C2	3	2	6.558	0.262	32	0.0082	ILS/ COMPRESSION
EABQC21BC	C	C2	3	2	6.818	0.261	32	0.0081	ILS/ COMPRESSION
EABQC21CC	C	C2	3	2	6.945	0.262	32	0.0082	ILS/ COMPRESSION

\*4T span was used.

Span of 4.5T was used for all specimens except EABQA11BC.

Average	7.109	Average	0.0084
Standard Dev.	0.328	Standard Dev.	
Coeff. of Var. [%]	4.617	Coeff. of Var. [%]	
Min.	6.443	Min.	0.0081
Max.	7.554	Max.	0.0087
Number of Spec.	21	Number of Spec.	21



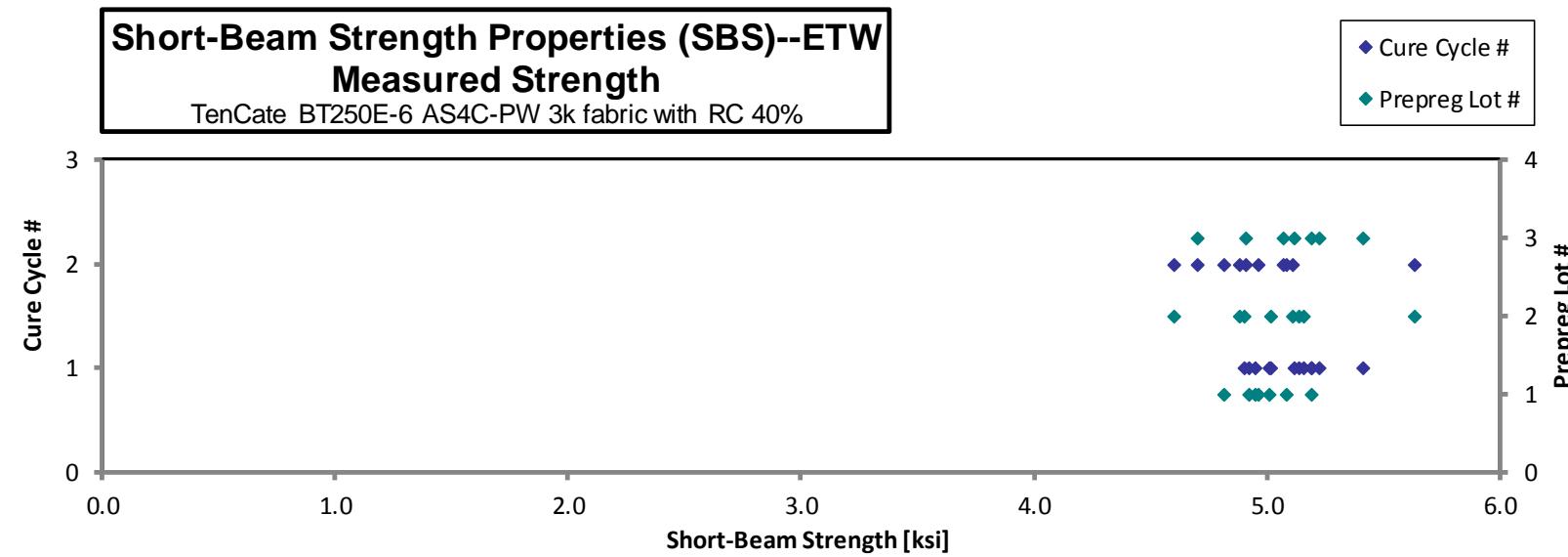
**Short-Beam Strength Properties (SBS)--ETW  
Strength**

TenCate BT250E-6 AS4C-PW 3k fabric with RC 40%

Specimen Number	EAC Batch #	EAC Cure Cycle	Prepreg Lot #	Cure Cycle #	Strength [ksi]	Avg. Specimen Thickness [in]	# Plies in Laminate	Avg. $t_{\text{ply}}$ [in]	Failure Mode
EABQA11GD	A	C1	1	1	5.193	0.268	32	0.0084	ILS
EABQA11HD	A	C1	1	1	4.949	0.270	32	0.0084	ILS
EABQA11ID	A	C1	1	1	4.920	0.272	32	0.0085	ILS
EABQA11JD	A	C1	1	1	5.011	0.273	32	0.0085	ILS
EABQA21DD	A	C2	1	2	4.812	0.270	32	0.0084	ILS
EABQA21ED	A	C2	1	2	4.963	0.273	32	0.0085	ILS
EABQA21FD	A	C2	1	2	5.081	0.275	32	0.0086	ILS
EABQB11GD	B	C1	2	1	4.901	0.264	32	0.0082	ILS
EABQB11HD	B	C1	2	1	5.017	0.266	32	0.0083	ILS
EABQB11ID	B	C1	2	1	5.138	0.267	32	0.0083	ILS
EABQB11JD	B	C1	2	1	5.154	0.268	32	0.0084	ILS
EABQB21DD	B	C2	2	2	5.630	0.263	32	0.0082	ILS
EABQB21ED	B	C2	2	2	4.597	0.266	32	0.0083	ILS
EABQB21FD	B	C2	2	2	4.884	0.263	32	0.0082	ILS
EABQB21GD	B	C2	2	2	5.109	0.263	32	0.0082	ILS
EABQC11FD	C	C1	3	1	5.409	0.267	32	0.0083	ILS
EABQC11GD	C	C1	3	1	5.113	0.262	32	0.0082	ILS
EABQC11HD	C	C1	3	1	5.193	0.263	32	0.0082	ILS
EABQC11ID	C	C1	3	1	5.226	0.264	32	0.0083	ILS
EABQC21ED	C	C2	3	2	4.702	0.261	32	0.0082	ILS
EABQC21FD	C	C2	3	2	4.905	0.260	32	0.0081	ILS
EABQC21GD	C	C2	3	2	5.070	0.261	32	0.0082	ILS

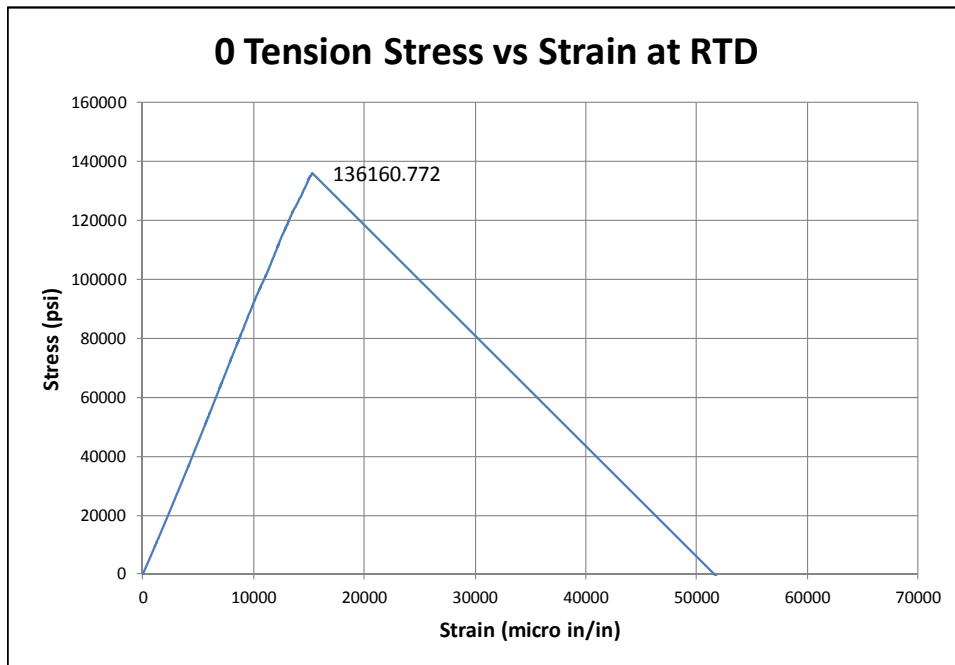
Span of 4T was used for all specimens.

Average	5.044	Average	0.0083
Standard Dev.	0.224	Standard Dev.	
Coeff. of Var. [%]	4.444	Coeff. of Var. [%]	
Min.	4.597	Min.	0.0081
Max.	5.630	Max.	0.0086
Number of Spec.	22	Number of Spec.	22

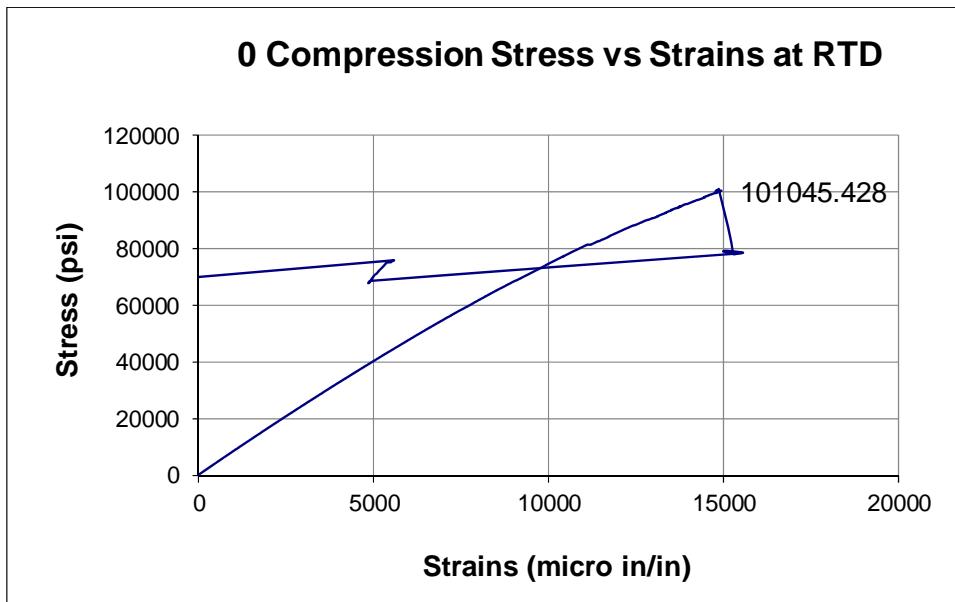


## 5. Full Stress vs. Strain Curve

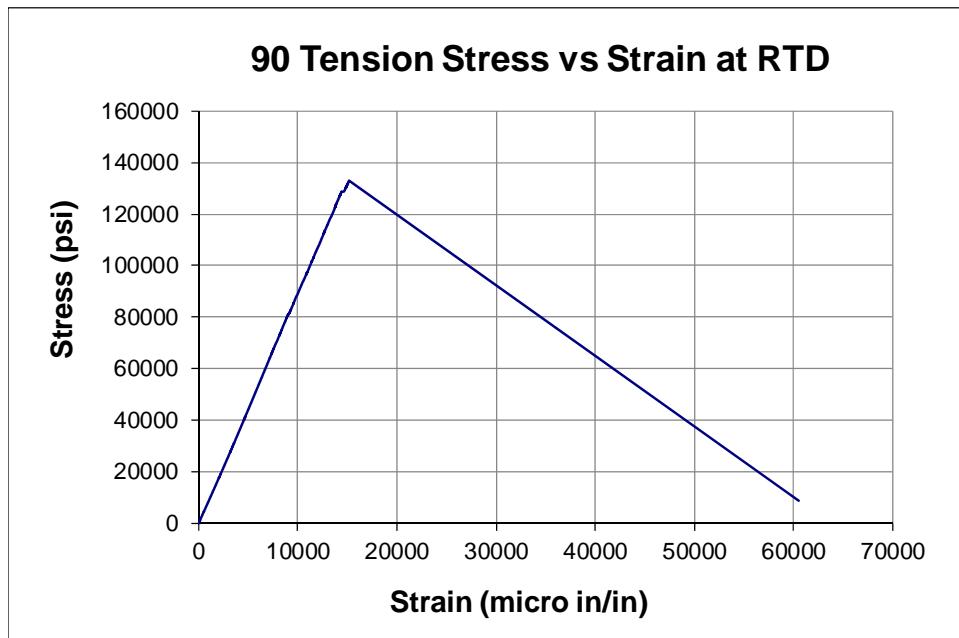
### 5.1 Warp Tension



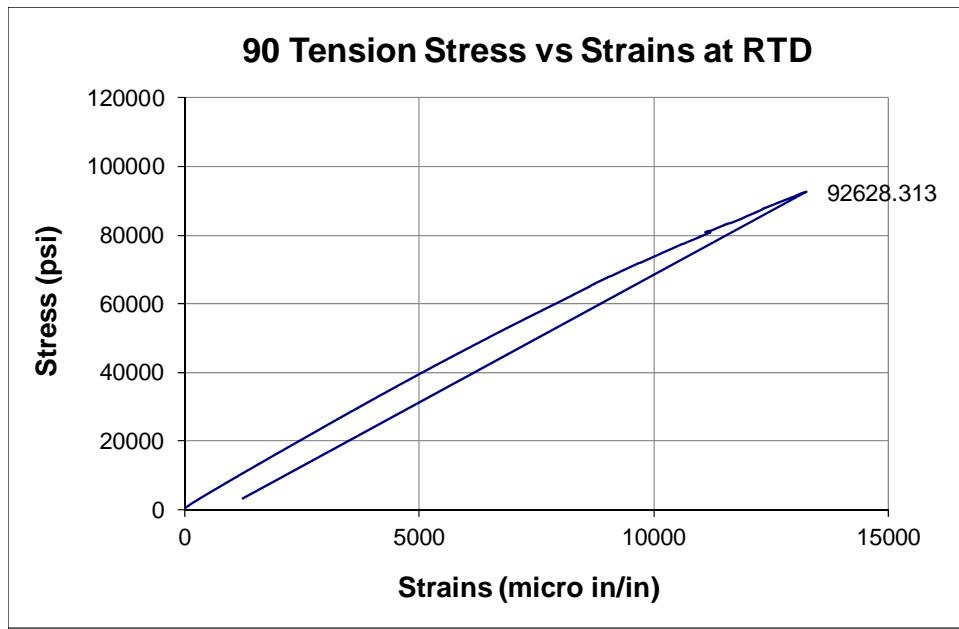
### 5.2 Warp Compression



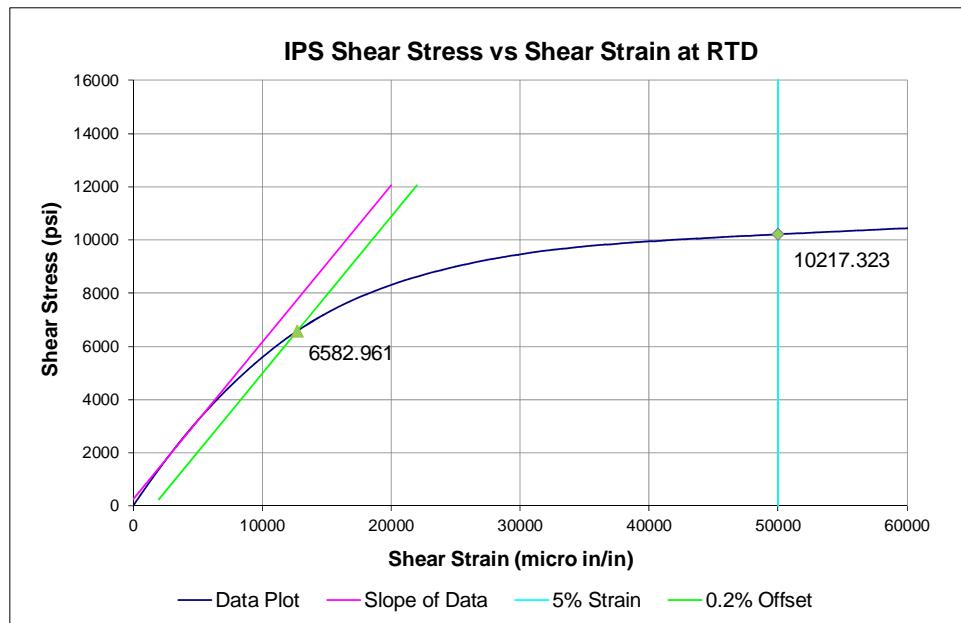
### 5.3 Fill Tension



### 5.4 Fill Compression

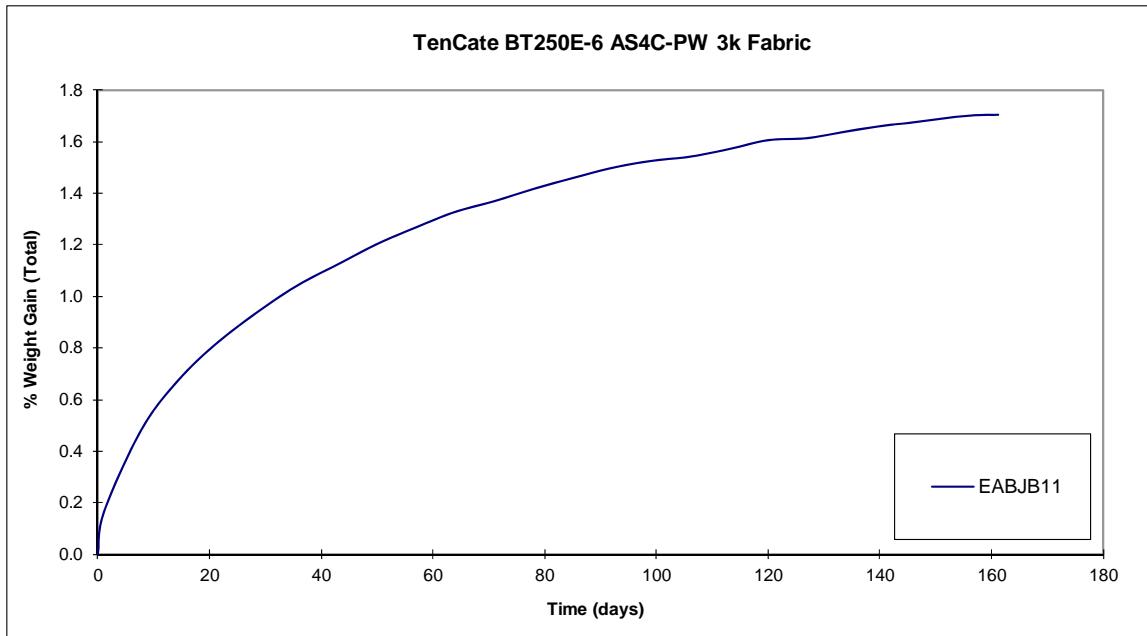


## 5.5 In-Plane Shear

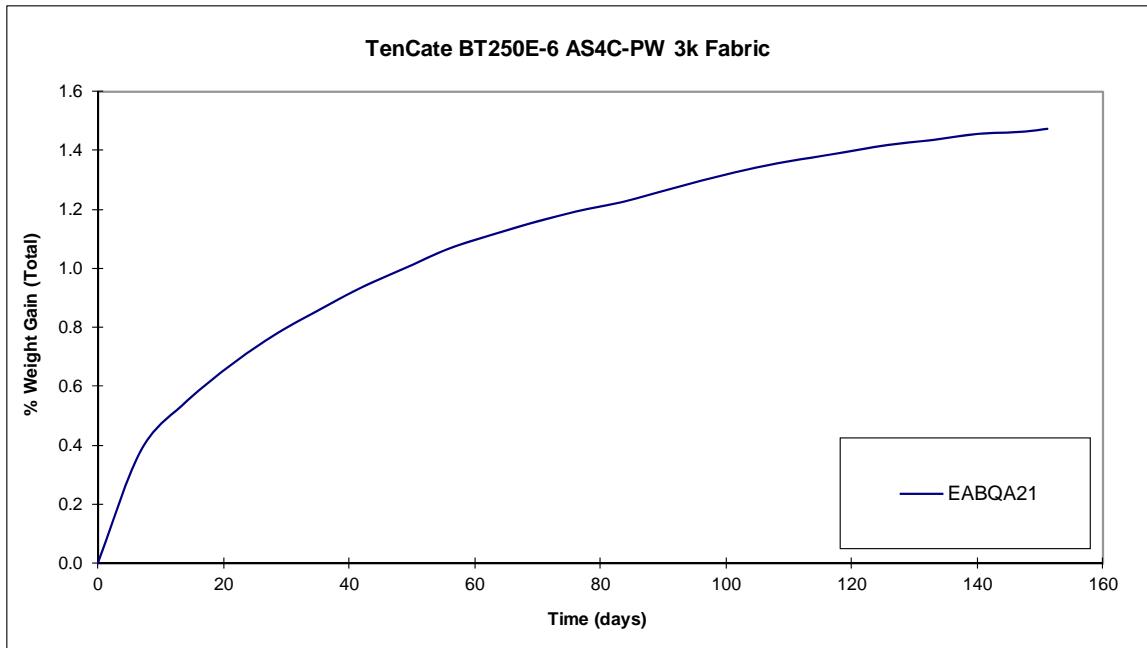


## 6. Moisture Conditioning Charts

### 6.1 Warp Tension – Thinnest Panel



### 6.2 Short-Beam Strength – Thickest Panel



For “wet” mechanical test specimens, the drying procedures may not have completely dried the specimens prior to moisture conditioning, so the total amount of moisture absorbed by the specimens may be higher than those recorded in the moisture gain charts.

## 7. DMA Results

DMA Results Summary				
Sample #	Onset Storage Modulus		Peak of Tangent Delta	
	Average		Average	
	Tg [°C]	Tg [°F]	Tg [°C]	Tg [°F]
EABJA11 - 1 (EA-E02-WT-A-C1)	135.18	275.32	178.51	353.32
EABJA11 - 2 (EA-E02-WT-A-C1)	135.79	276.42	177.73	351.91
EABJA21 - 1 (EA-E02-WT-A-C2)	137.81	280.06	178.25	352.85
EABJA21 - 2 (EA-E02-WT-A-C2)	138.03	280.45	177.79	352.02
EABJB11 - 1 (EA-E02-WT-B-C1)	134.57	274.23	174.81	346.66
EABJB11 - 2 (EA-E02-WT-B-C1)	133.84	272.91	174.30	345.74
EABJB21 - 1 (EA-E02-WT-B-C2)	136.15	277.07	180.06	356.11
EABJB21 - 2 (EA-E02-WT-B-C2)	136.16	277.09	178.93	354.07
EABJC11 - 1 (EA-E02-WT-C-C1)	130.53	266.95	176.41	349.54
EABJC11 - 2 (EA-E02-WT-C-C1)	130.15	266.27	178.42	353.16
EABJC21 - 1 (EA-E02-WT-C-C2)	132.73	270.91	180.43	356.77
EABJC21 - 2 (EA-E02-WT-C-C2)	132.27	270.09	180.17	356.31
EABL21 - 1 (EA-E02-WC-B-C3)	136.56	277.81	179.49	355.08
EABL21 - 2 (EA-E02-WC-B-C3)	137.28	279.10	180.20	356.36
EABUA11 - 1 (EA-E02-FT-A-C3)	136.84	278.31	178.49	353.28
EABUA11 - 2 (EA-E02-FT-A-C3)	137.19	278.94	178.57	353.43
EABUA21 - 1 (EA-E02-FT-A-C4)	137.43	279.37	178.63	353.53
EABUA21 - 2 (EA-E02-FT-A-C4)	136.52	277.74	179.17	354.51
EABUB11 - 1 (EA-E02-FT-B-C4)	135.75	276.35	178.78	353.80
EABUB11 - 2 (EA-E02-FT-B-C4)	135.81	276.46	178.96	354.13
EABUB21 - 1 (EA-E02-FT-B-C5)	137.23	279.01	182.06	359.71
EABUB21 - 2 (EA-E02-FT-B-C5)	136.22	277.20	182.05	359.69
EABUC11 - 1 (EA-E02-FT-C-C4)	134.35	273.83	182.32	360.18
EABUC11 - 2 (EA-E02-FT-C-C4)	134.84	274.71	181.97	359.55
EABUC21 - 1 (EA-E02-FT-C-C5)	136.09	276.96	182.35	360.23
EABUC21 - 2 (EA-E02-FT-C-C5)	137.01	278.62	182.50	360.50
EABNB21 - 1 (EA-E02-IPS-B-C11)	132.93	271.27	157.71	315.88
EABNB21 - 2 (EA-E02-IPS-B-C11)	131.97	269.55	158.08	316.54
Average	135.26	275.46	177.76	351.96
Standard Deviation	2.17	3.90	5.98	10.77

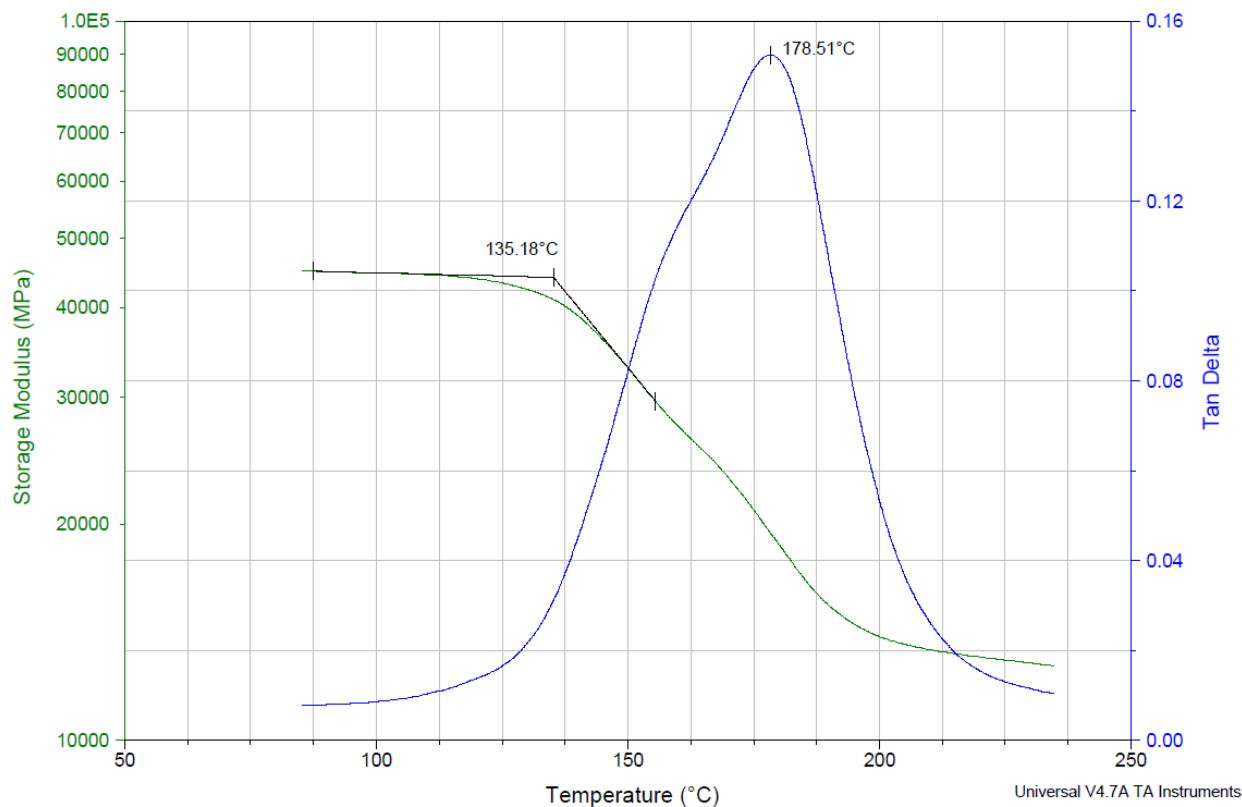
<b>DMA Results Summary</b>				
<b>TenCate BT250E-6 AS4C-PW 3k Fabric WET</b>				
Sample #	Onset Storage Modulus		Peak of Tangent Delta	
	Average		Average	
	Tg [°C]	Tg [°F]	Tg [°C]	Tg [°F]
EABJA11 - 1 (EA-E02-WT-A-C1)	122.33	252.19	142.99	289.38
EABJA11 - 2 (EA-E02-WT-A-C1)	121.36	250.45	143.35	290.03
EABJA21 - 1 (EA-E02-WT-A-C2)	120.03	248.05	140.26	284.47
EABJA21 - 2 (EA-E02-WT-A-C2)	121.23	250.21	142.62	288.72
EABJB11 - 1 (EA-E02-WT-B-C1)	125.51	257.92	147.24	297.03
EABJB11 - 2 (EA-E02-WT-B-C1)	125.52	257.94	147.22	297.00
EABJB21 - 1 (EA-E02-WT-B-C2)	121.48	250.66	142.08	287.74
EABJB21 - 2 (EA-E02-WT-B-C2)	120.21	248.38	143.33	289.99
EABJC11 - 1 (EA-E02-WT-C-C1)	122.48	252.46	146.24	295.23
EABJC11 - 2 (EA-E02-WT-C-C1)	121.23	250.21	145.59	294.06
EABJC21 - 1 (EA-E02-WT-C-C2)	120.46	248.83	143.27	289.89
EABJC21 - 2 (EA-E02-WT-C-C2)	122.41	252.34	143.22	289.80
EABLB21 - 1 (EA-E02-WC-B-C3)	121.42	250.56	141.32	286.38
EABLB21 - 2 (EA-E02-WC-B-C3)	122.05	251.69	142.16	287.89
EABUA11 - 1 (EA-E02-FT-A-C3)	117.46	243.43	139.11	282.40
EABUA11 - 2 (EA-E02-FT-A-C3)	118.03	244.45	138.75	281.75
EABUA21 - 1 (EA-E02-FT-A-C4)	120.66	249.19	141.65	286.97
EABUA21 - 2 (EA-E02-FT-A-C4)	119.41	246.94	140.86	285.55
EABUB11 - 1 (EA-E02-FT-B-C4)	121.74	251.13	144.23	291.61
EABUB11 - 2 (EA-E02-FT-B-C4)	123.53	254.35	143.69	290.64
EABUB21 - 1 (EA-E02-FT-B-C5)	120.52	248.94	141.13	286.03
EABUB21 - 2 (EA-E02-FT-B-C5)	118.51	245.32	140.42	284.76
EABUC11 - 1 (EA-E02-FT-C-C4)	121.17	250.11	143.59	290.46
EABUC11 - 2 (EA-E02-FT-C-C4)	121.70	251.06	142.71	288.88
EABUC21 - 1 (EA-E02-FT-C-C5)	122.09	251.76	142.20	287.96
EABUC21 - 2 (EA-E02-FT-C-C5)	121.16	250.09	142.41	288.34
EABNB21 - 1 (EA-E02-IPS-B-C11)	114.41	237.94	145.63	294.13
EABNB21 - 2 (EA-E02-IPS-B-C11)	116.06	240.91	146.15	295.07
Average	120.86	249.55	142.98	289.36
Standard Deviation	2.41	4.34	2.24	4.03

## 7.1 DMA Dry Batch A

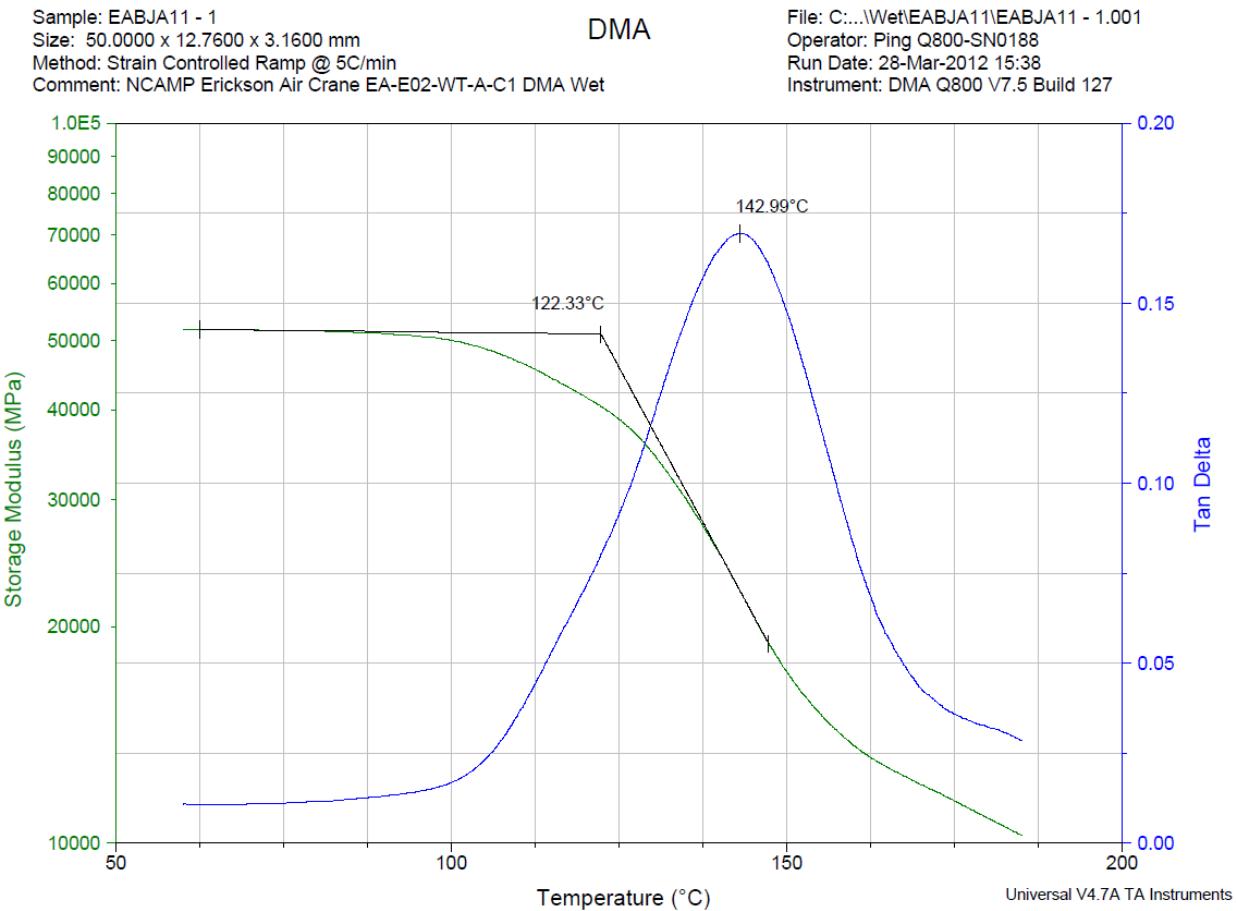
Sample: EABJA11 - 1  
Size: 50.0000 x 12.7400 x 3.2000 mm  
Method: Strain Controlled Ramp @ 5C/min  
Comment: NCAMP Erickson Air Crane EA-E02-WT-A-C1 DMA Dry

DMA

File: C:\Dry\EABJA11\EABJA11 - 1.001  
Operator: Ping Q800-SN0188  
Run Date: 27-Oct-2011 10:42  
Instrument: DMA Q800 V7.5 Build 127



## 7.2 DMA Wet Batch A



## 8. Deviations

- D2344 SBS for CTD, RTD and ETD conditions were tested with a mixture of 4T and 4.5T span length. The SBS for ETW condition was tested with 4T span length.